

Influence of Downstream Waves on Impinging SBLI

Ramachandra Kannan

Timothy Missing, Holger Babinsky



Funded by
Harding Distinguished Postgraduate
Scholars Programme

Shock-Boundary Layer Interactions

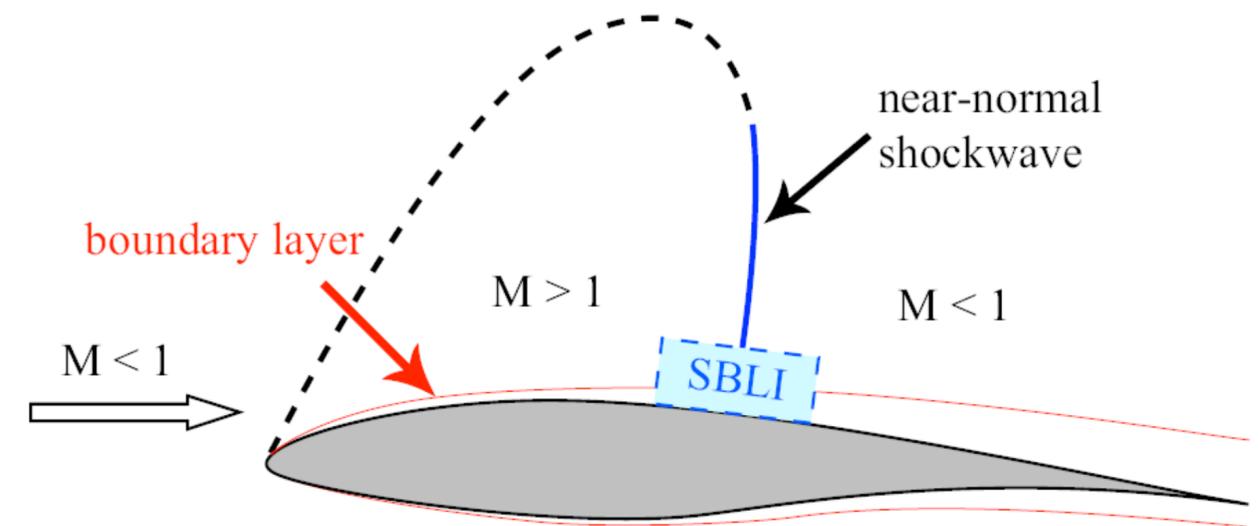
Widely observed in high-speed wall-bounded flow fields

Occur in many systems - supersonic intakes and transonic airfoils

Associated with flow separation and losses



<https://www.heritageconcorde.com/air-in-take-system>



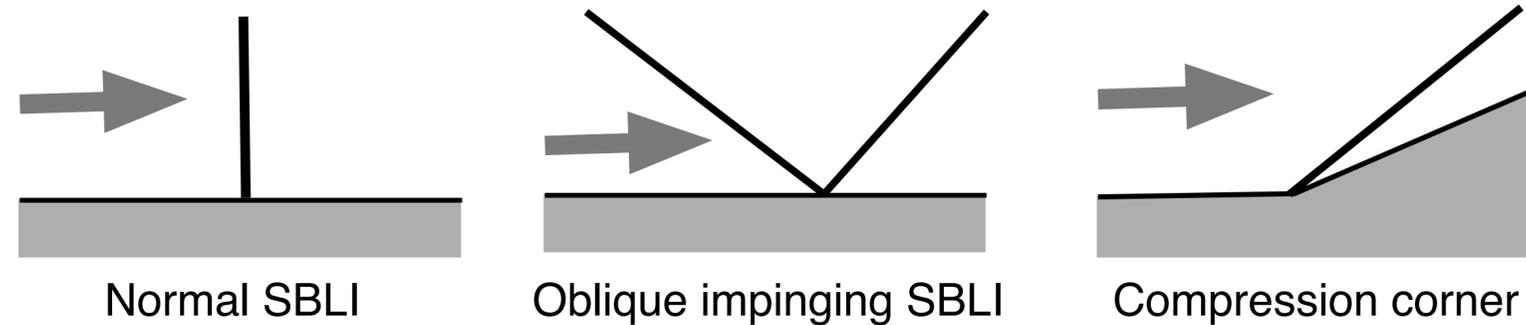
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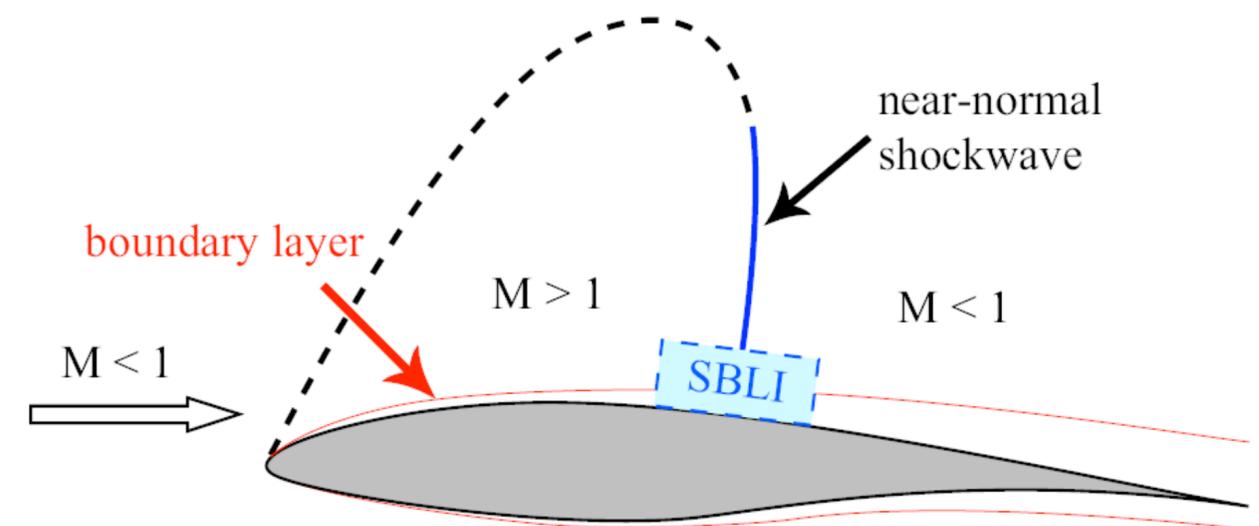
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Canonical configurations:



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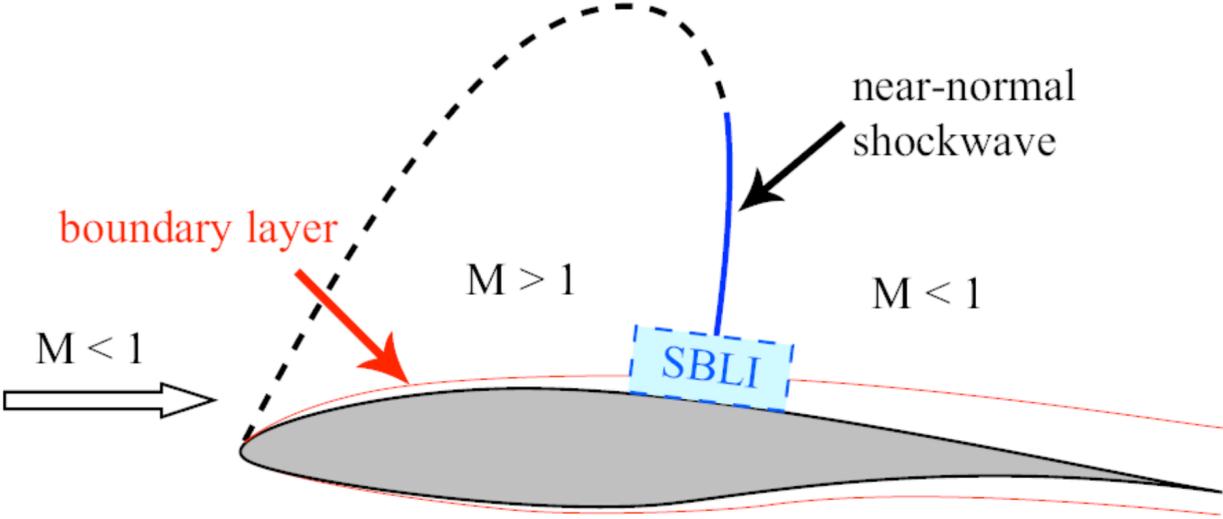
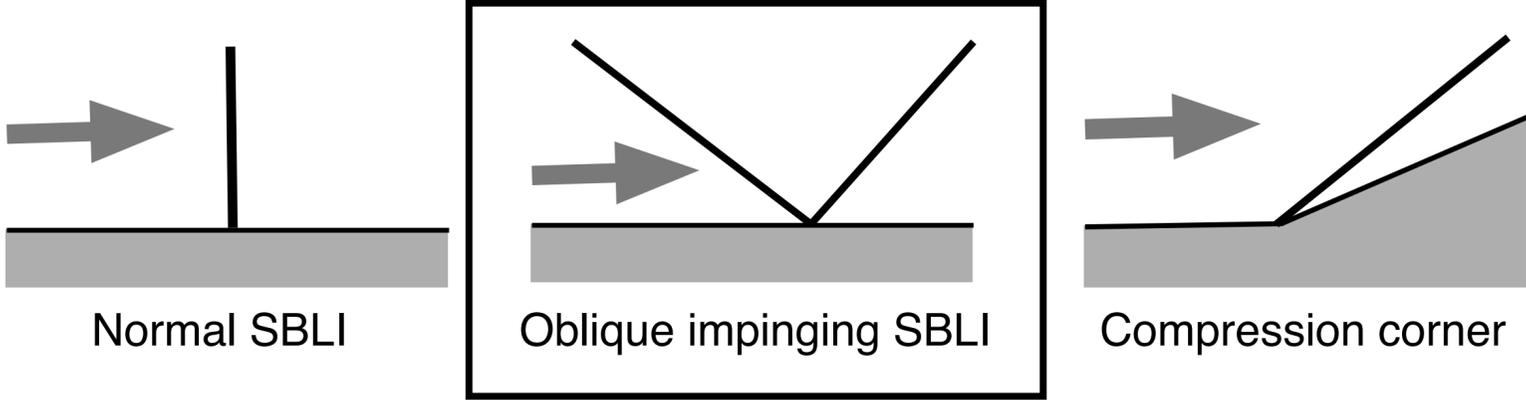
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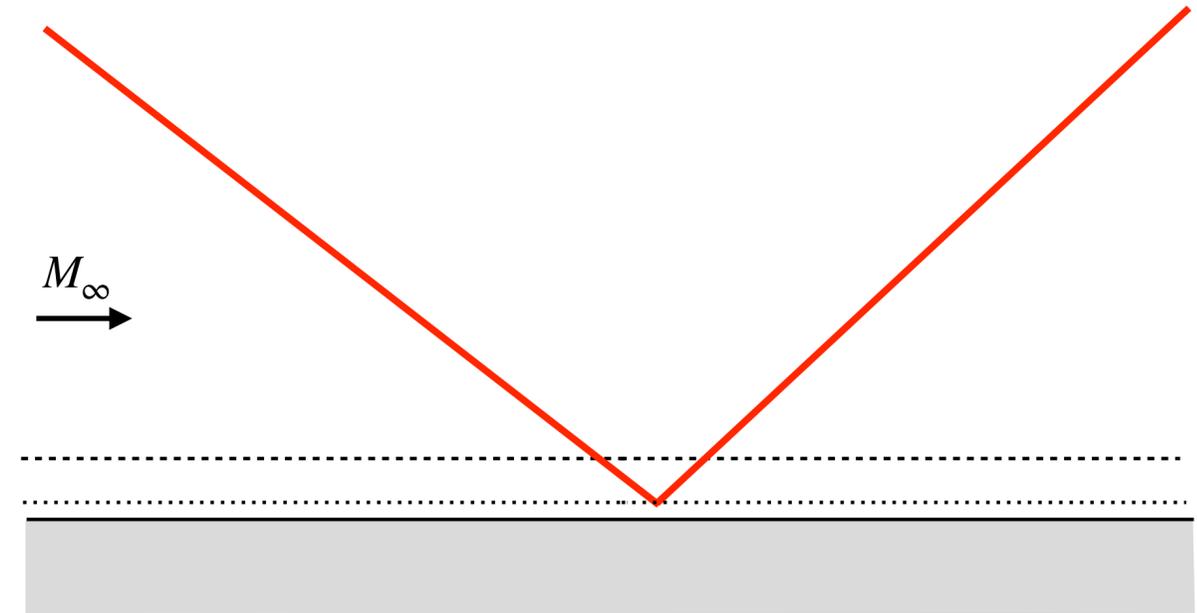


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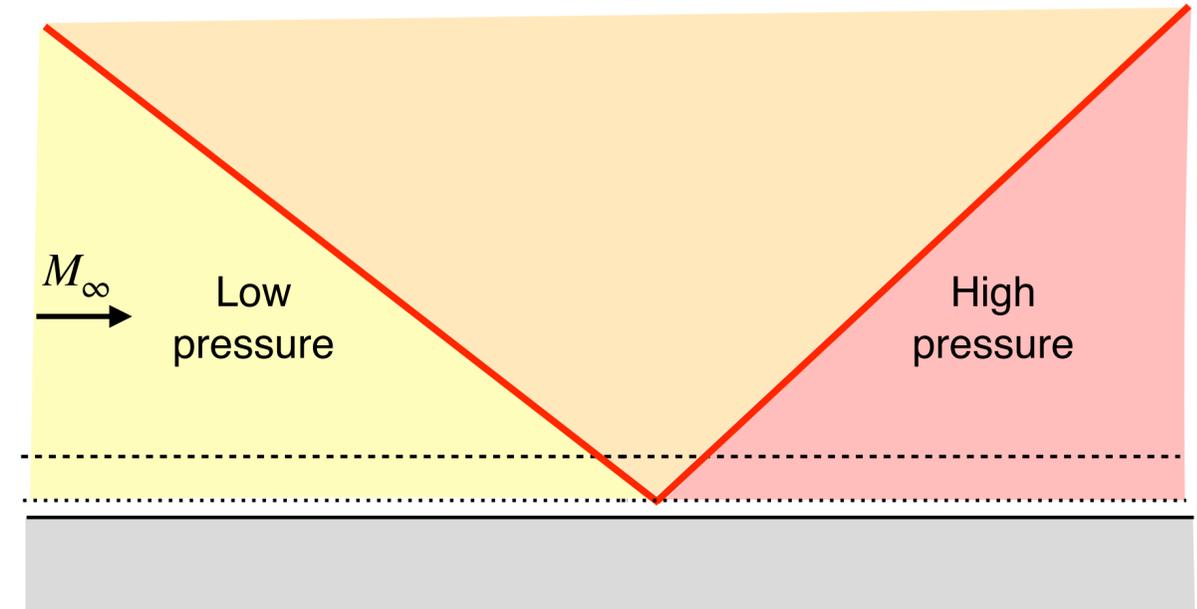


Oblique SBLI



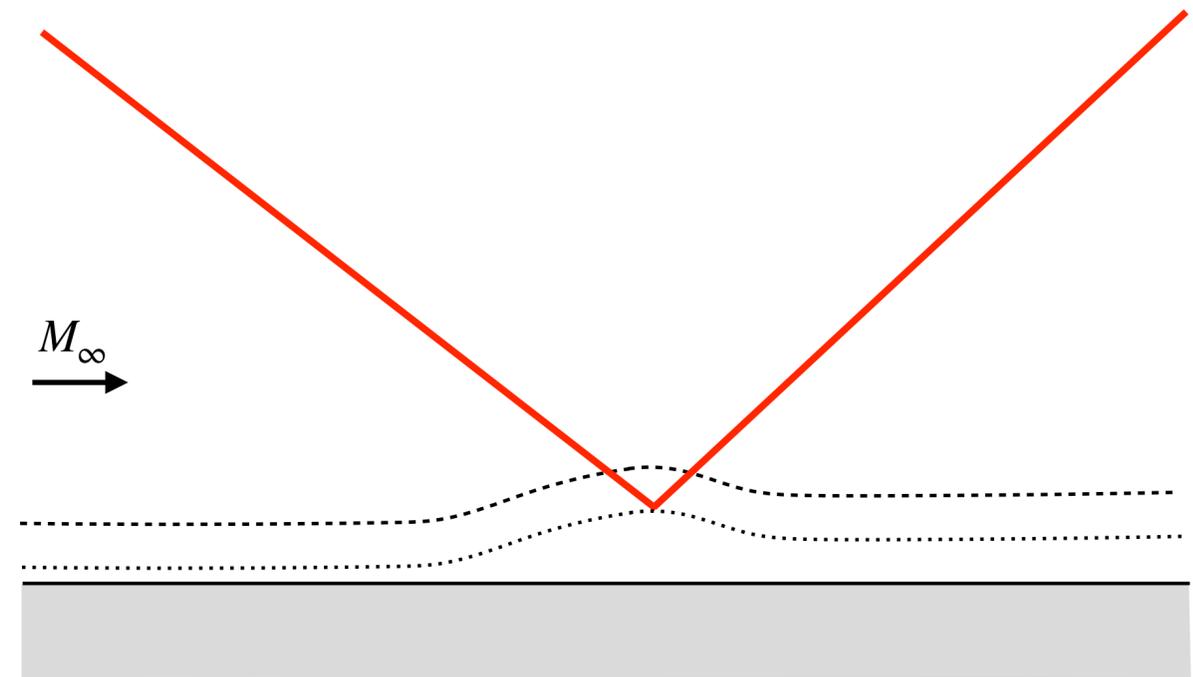
Oblique SBLI

Adverse pressure gradient imposed by the shock



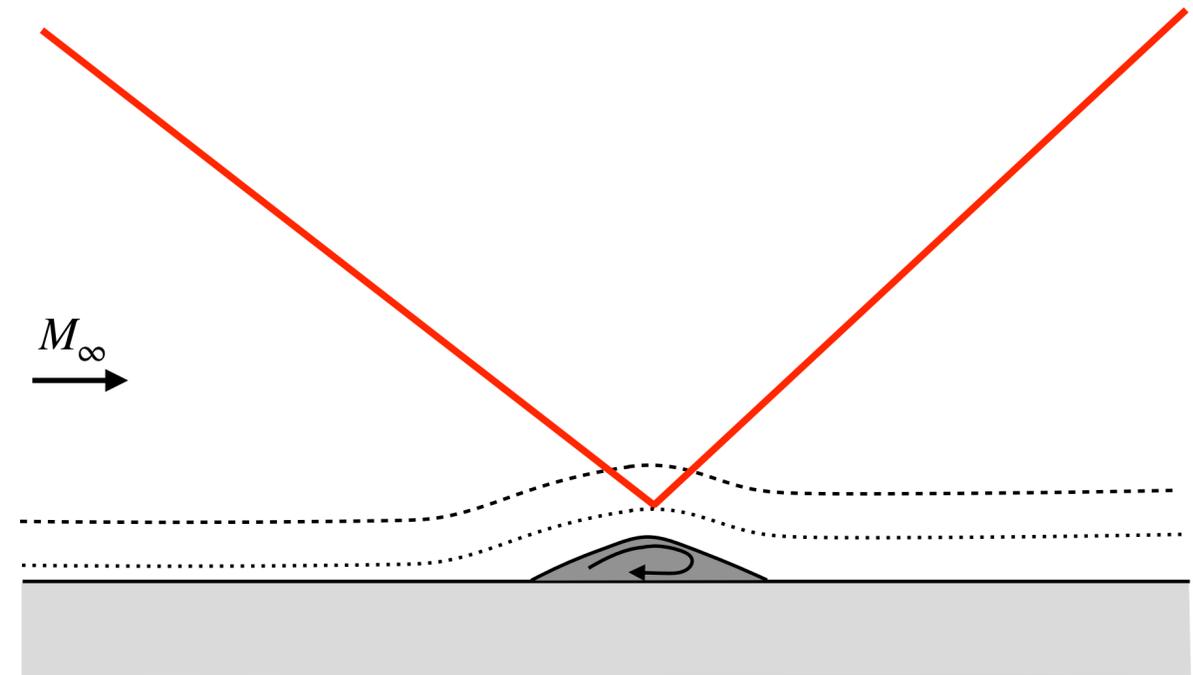
Oblique SBLI

Adverse pressure gradient imposed by the shock
Boundary layer thickens ahead of shock



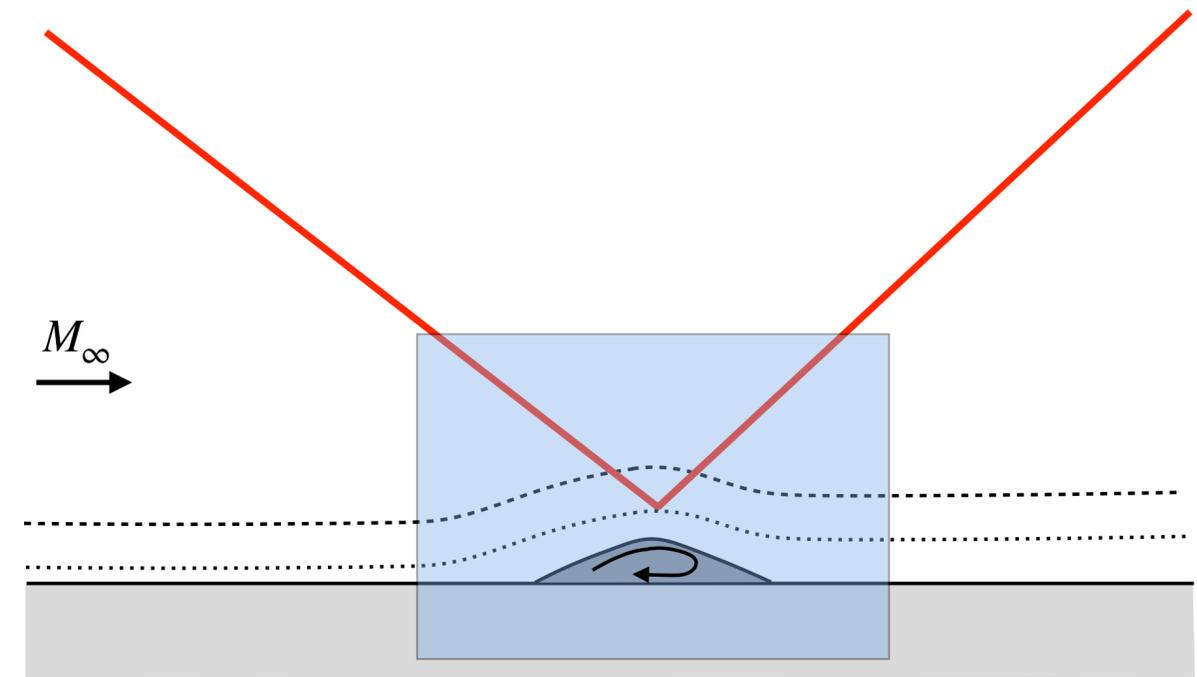
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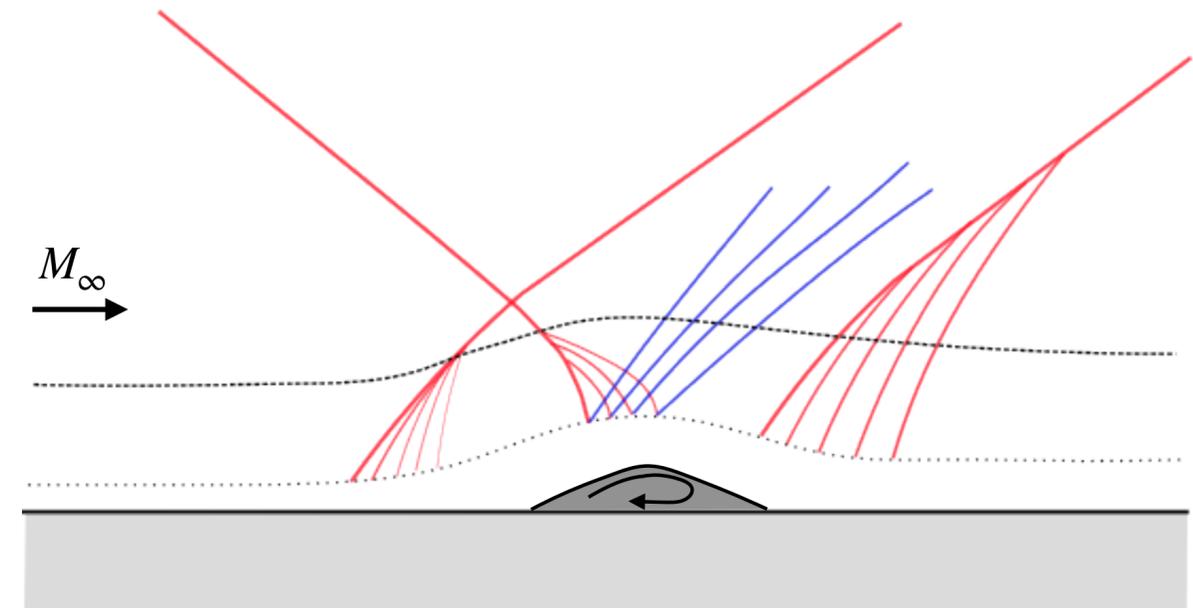


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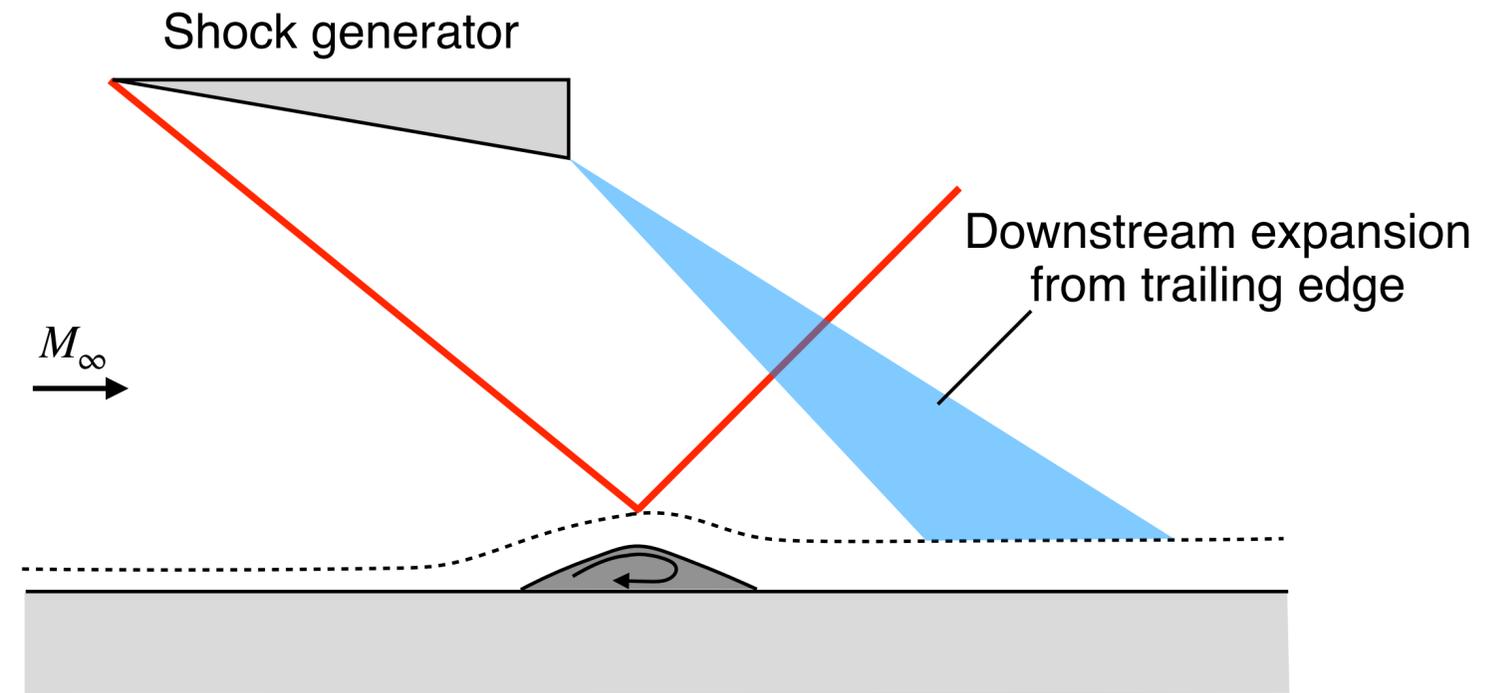
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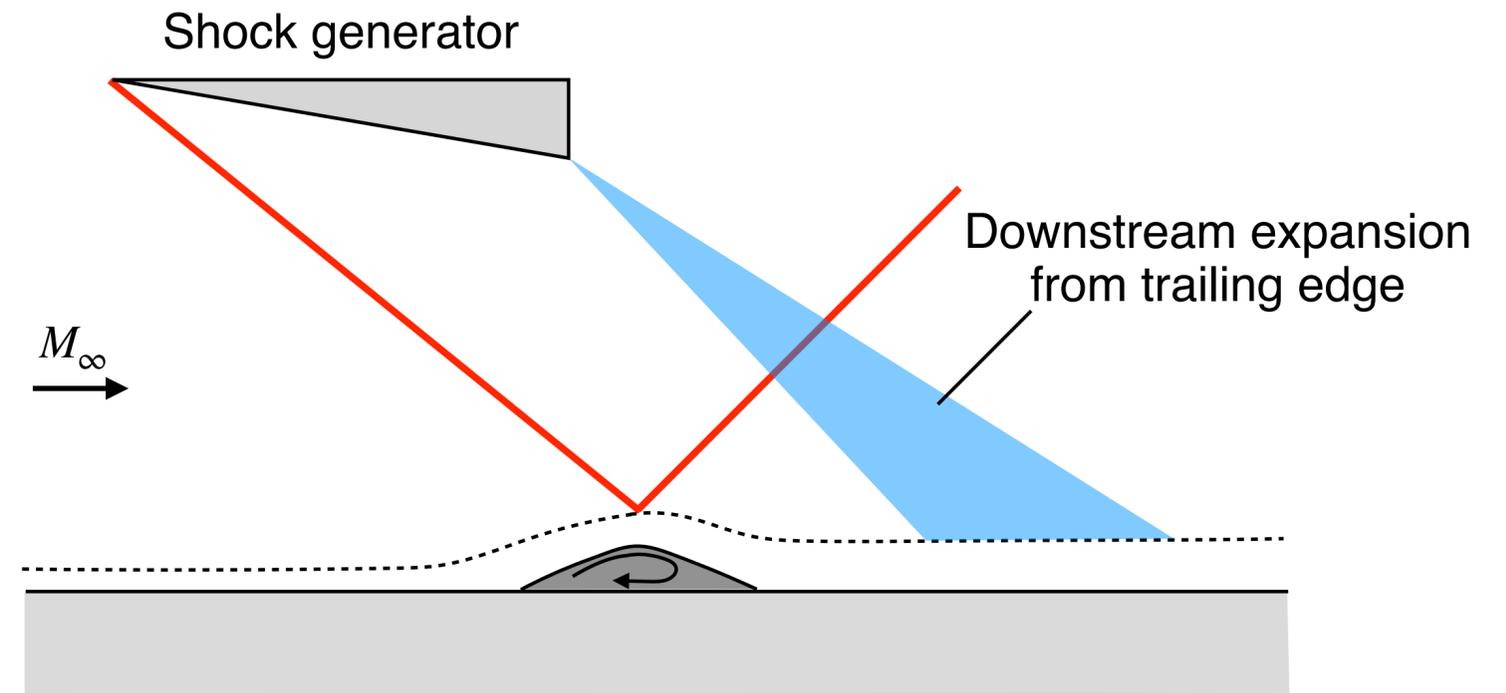
Oblique SBLI

- Adverse pressure gradient imposed by the shock
- Boundary layer thickens ahead of shock
- Separation possible
- Typical experiments introduce additional expansion



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Downstream waves are generally ignored - **Is this valid?**

Downstream waves in literature

Grossman and Bruce, 2018 [1]:

Separation size reduces with decreasing distance D

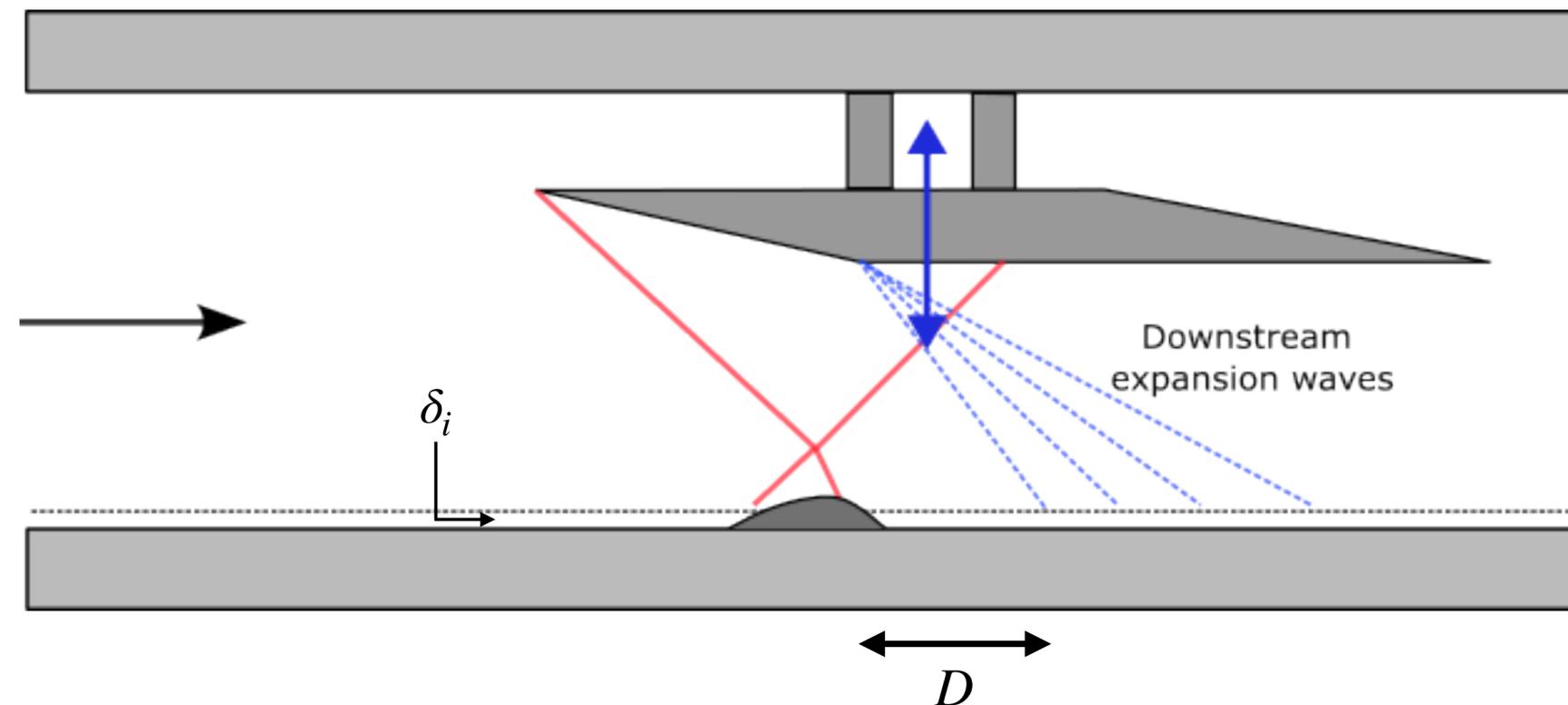
Effect observed for D up to $11\delta_i$

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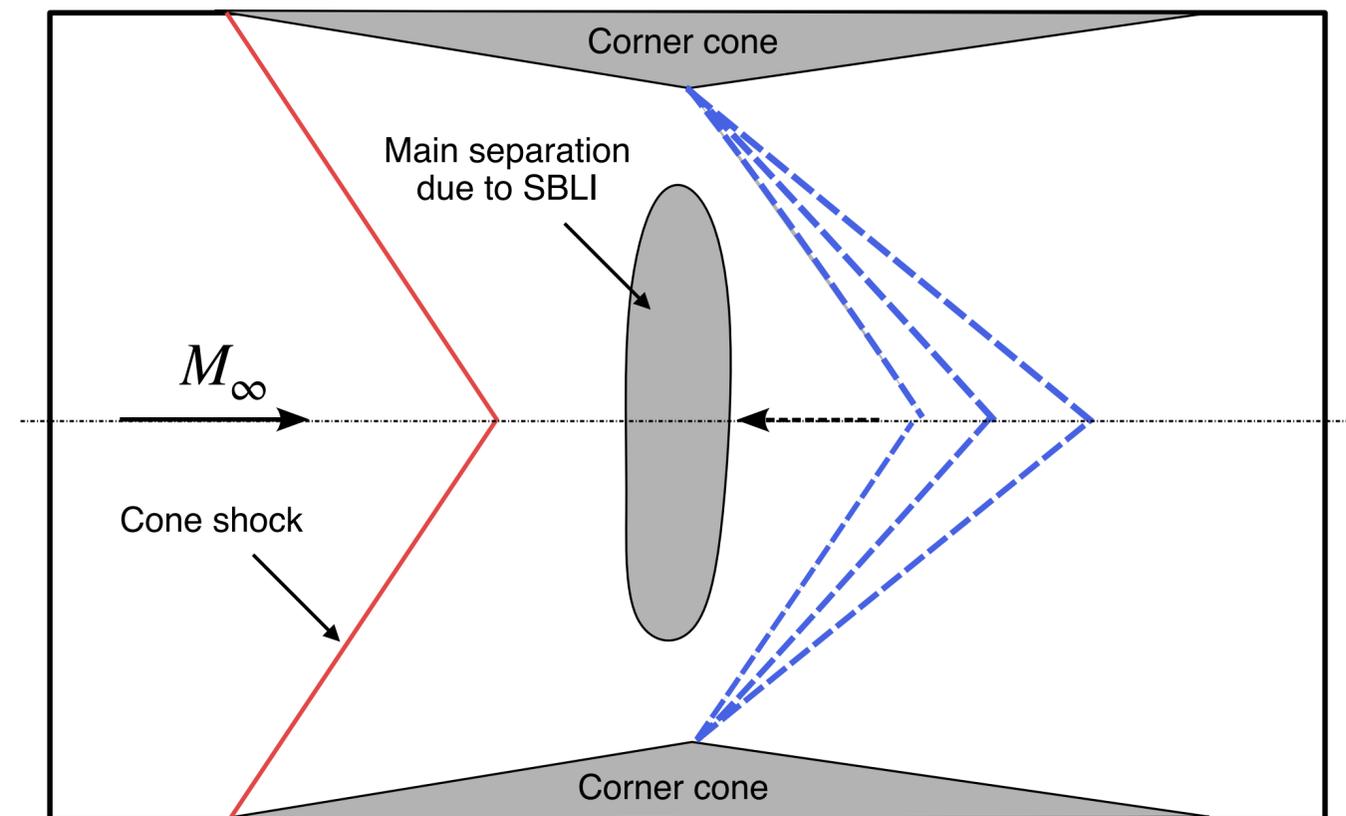


Downstream waves in literature

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Missing and Babinsky, 2023 [2]:

Use of corner cones to study the effect of corner separation

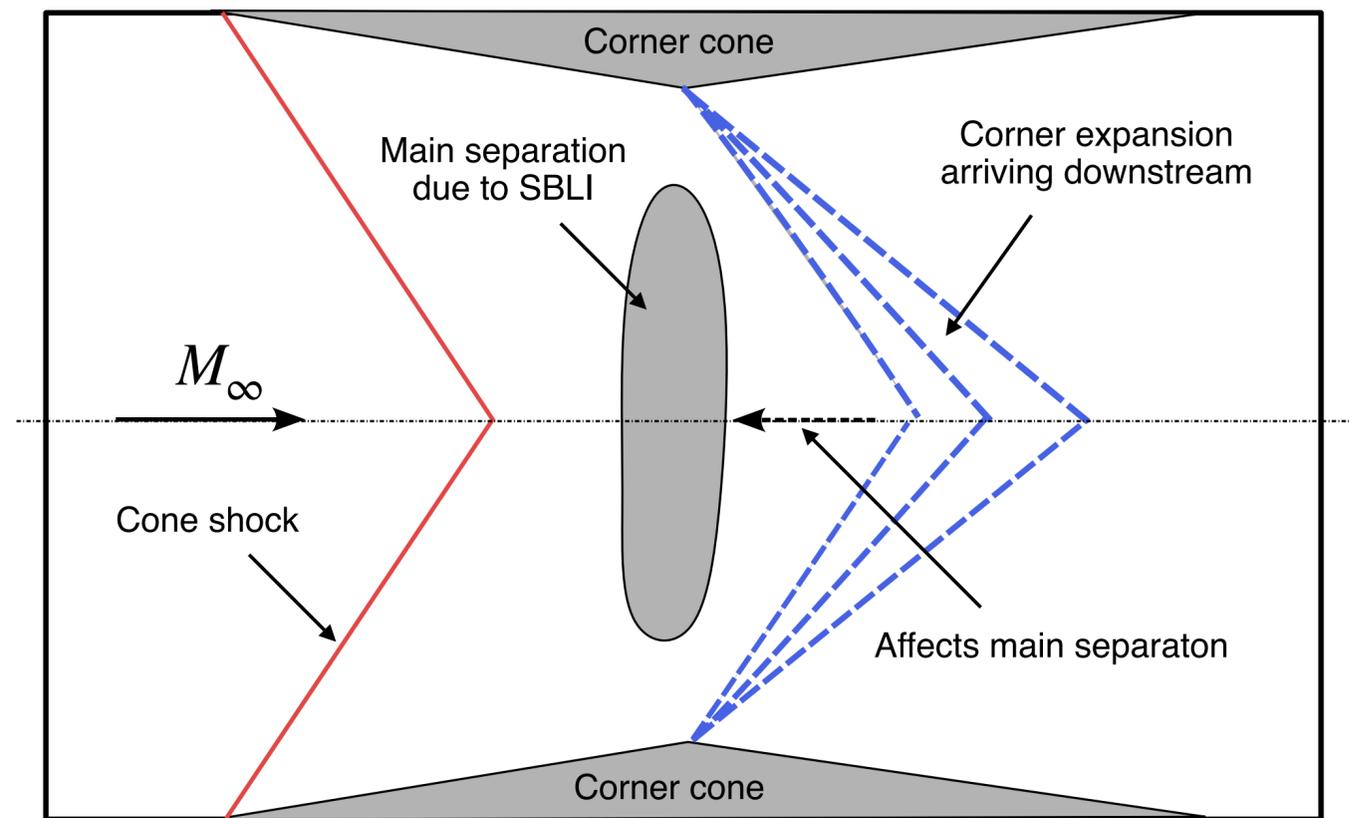


Downstream waves in literature

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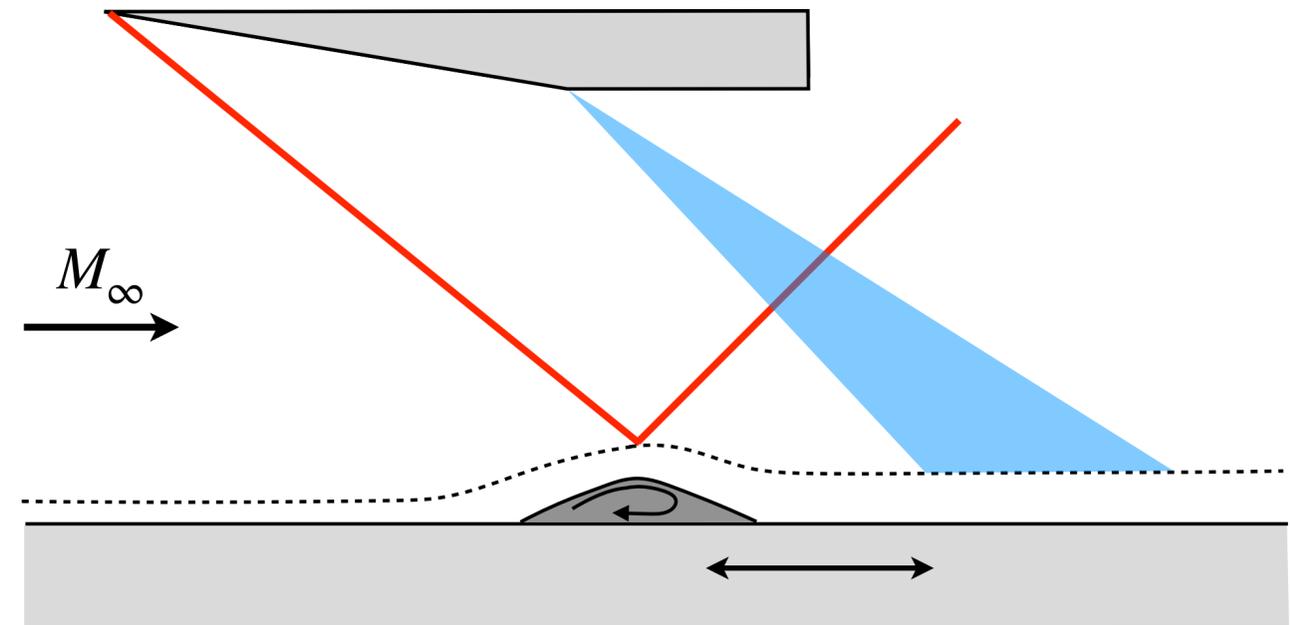
Downstream expansion influenced the primary SBLI



Key Questions

Are downstream waves important?

Do these factors play significant roles?

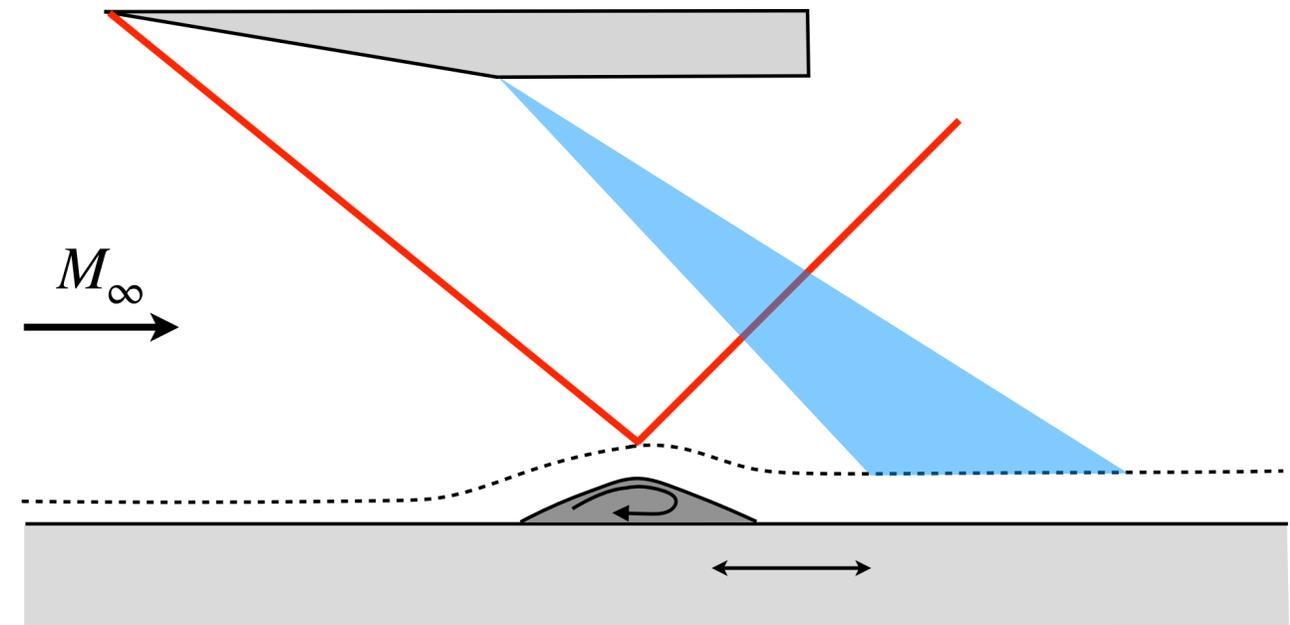


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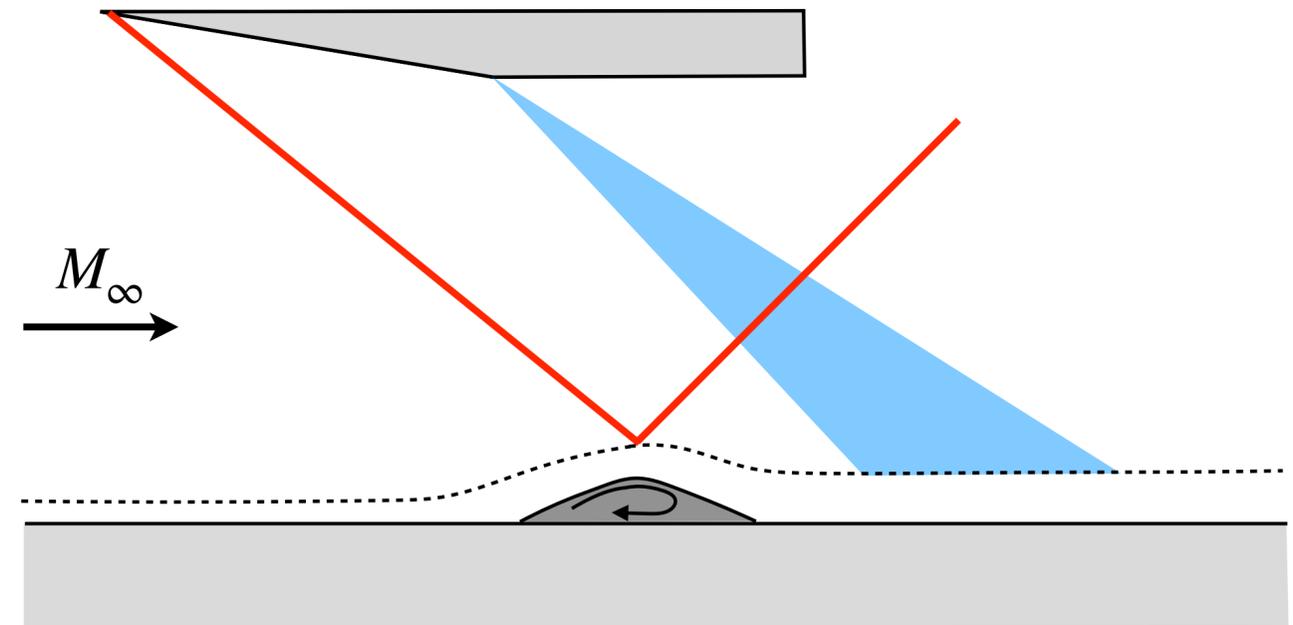


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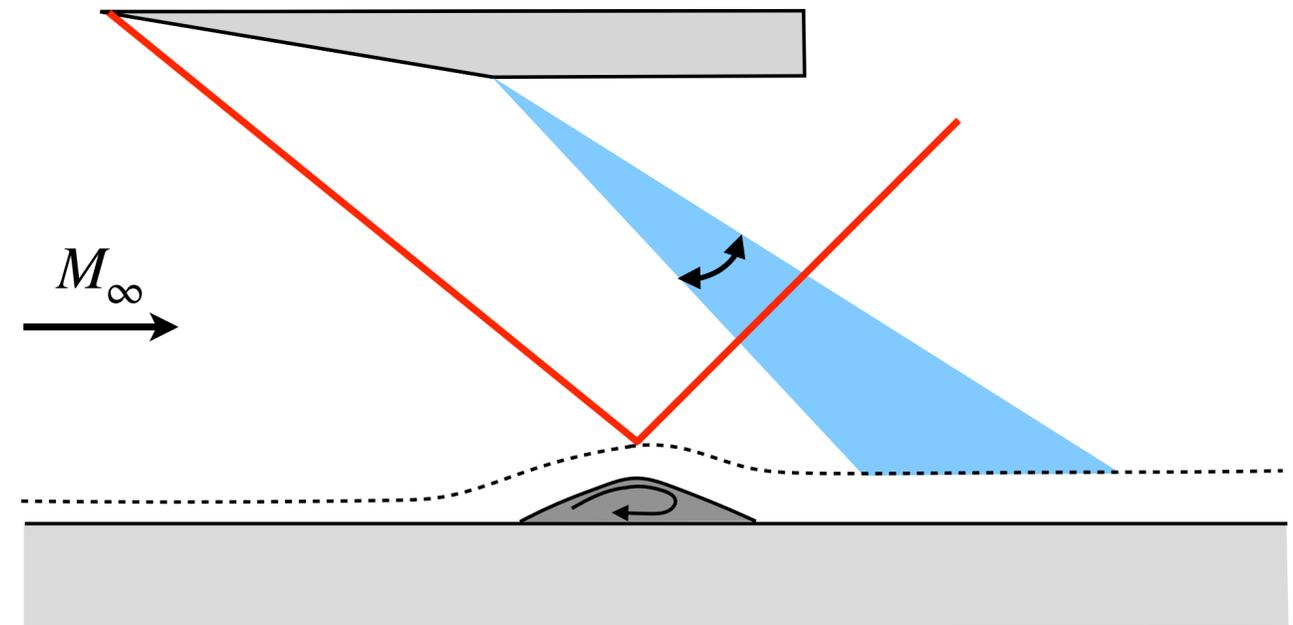


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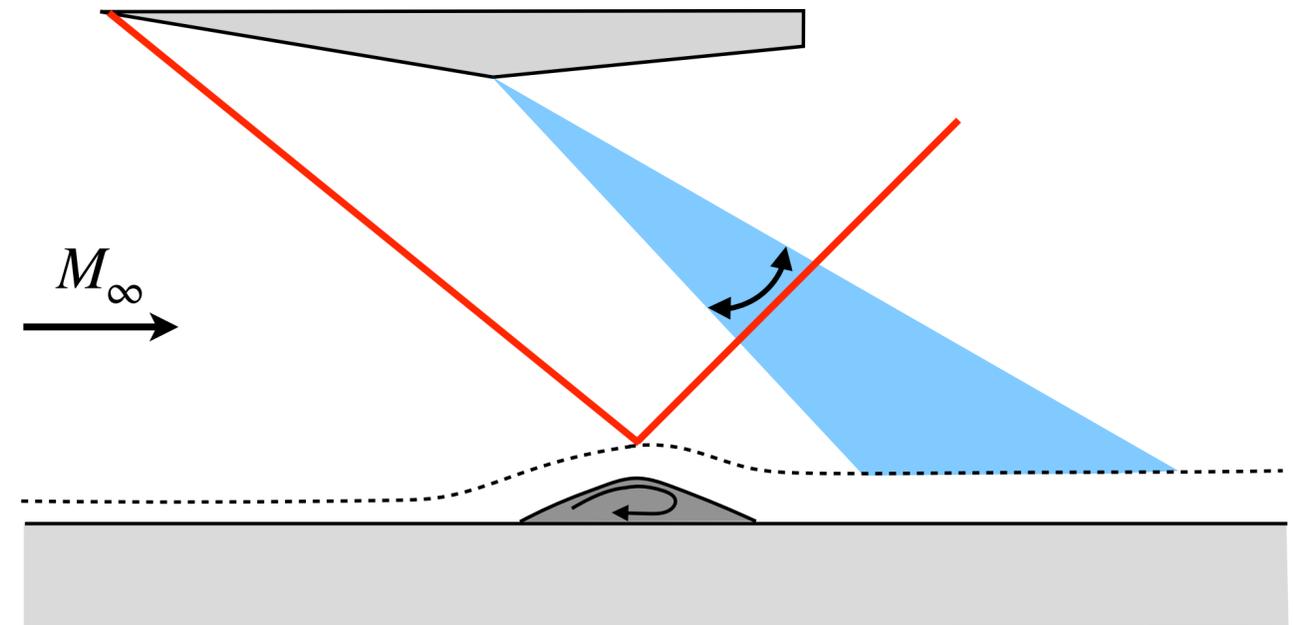


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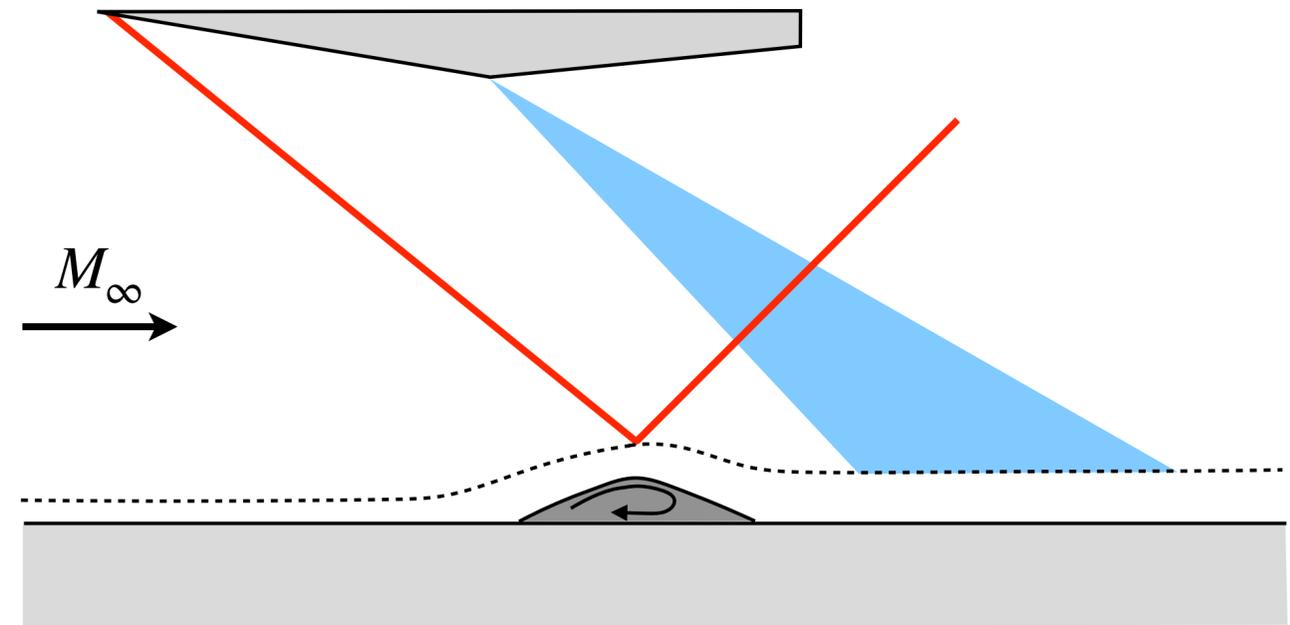


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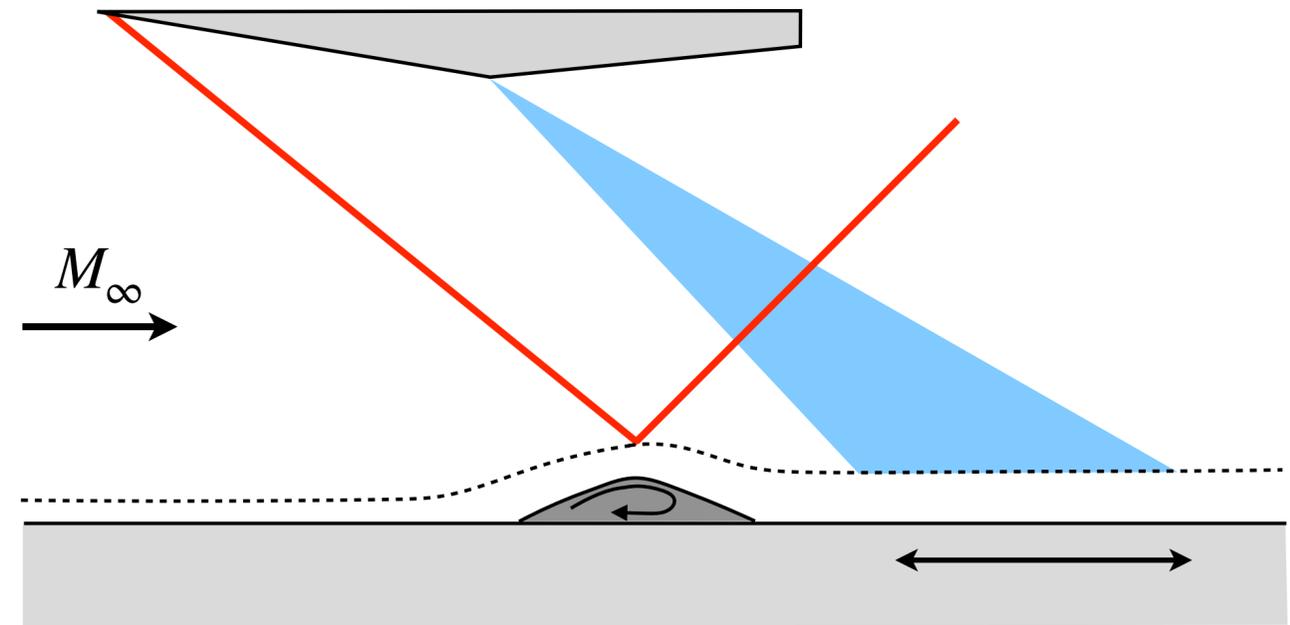


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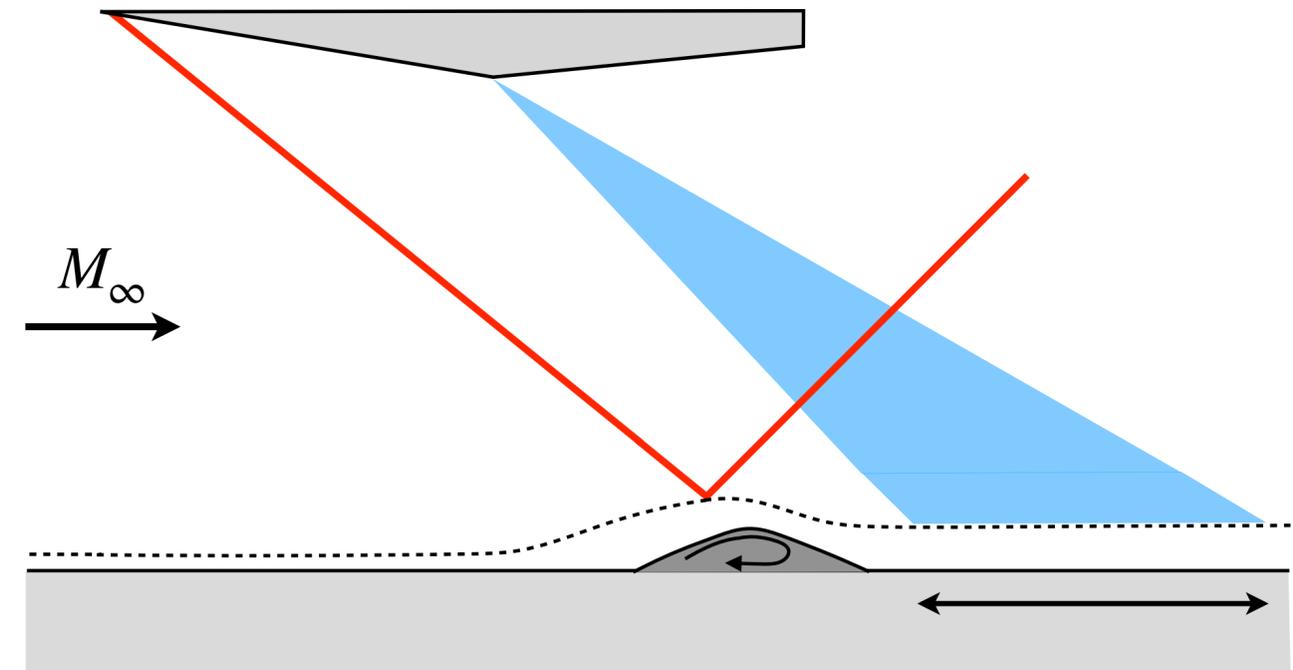


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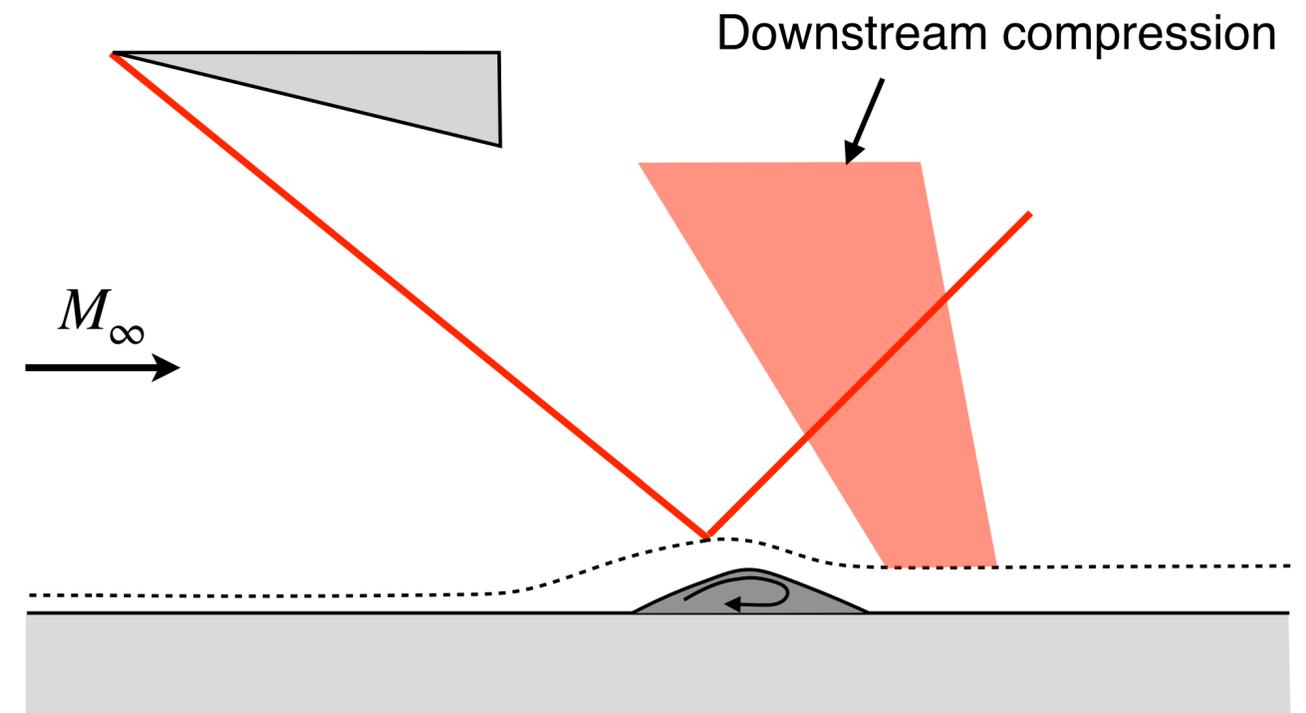


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- Type of wave



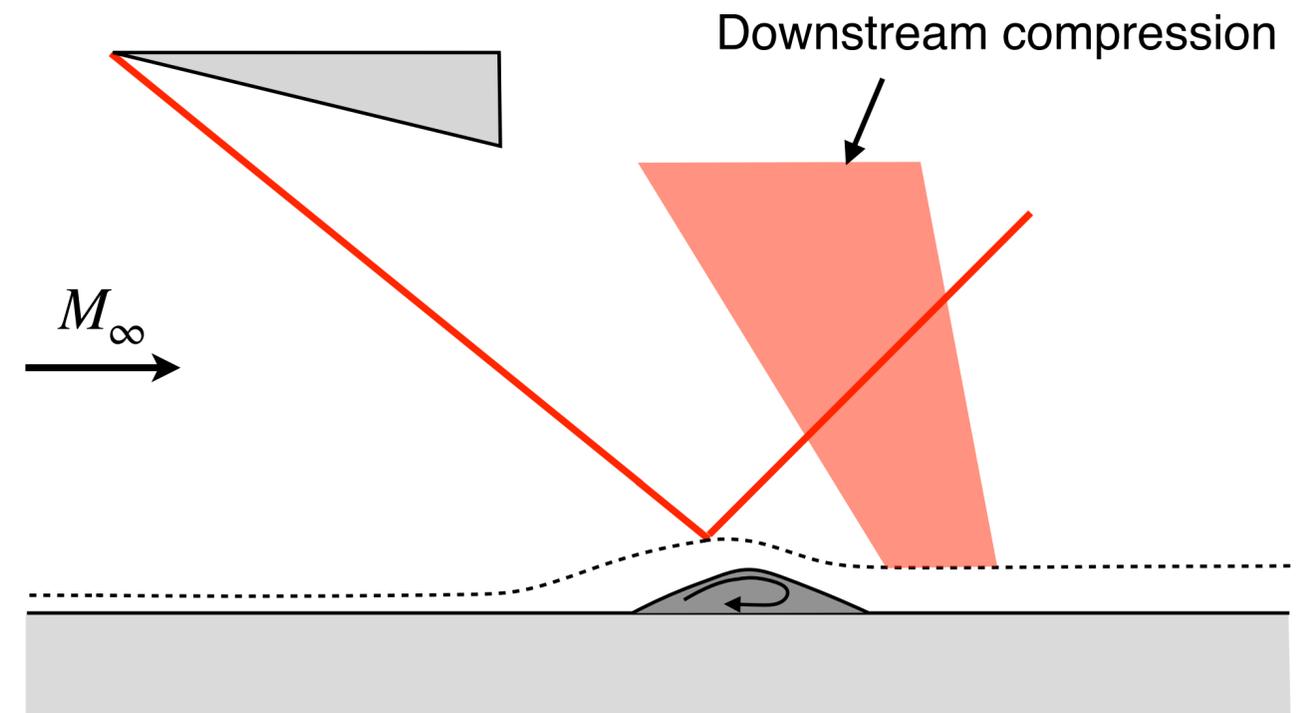
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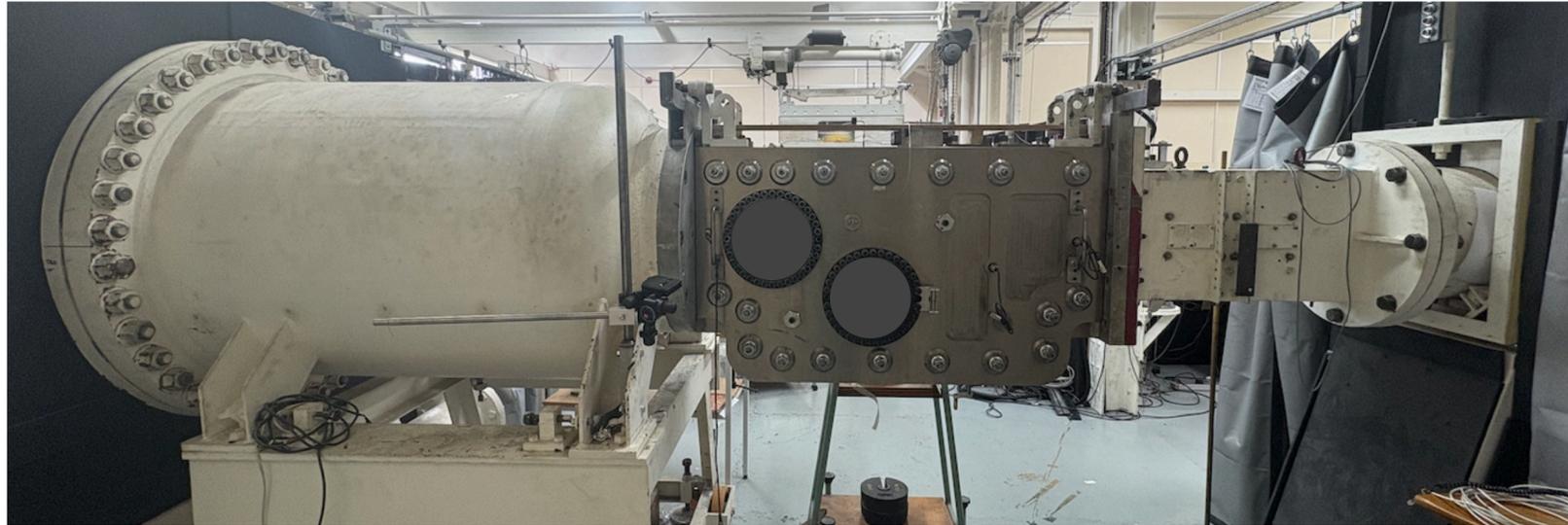
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What is the underlying mechanism?



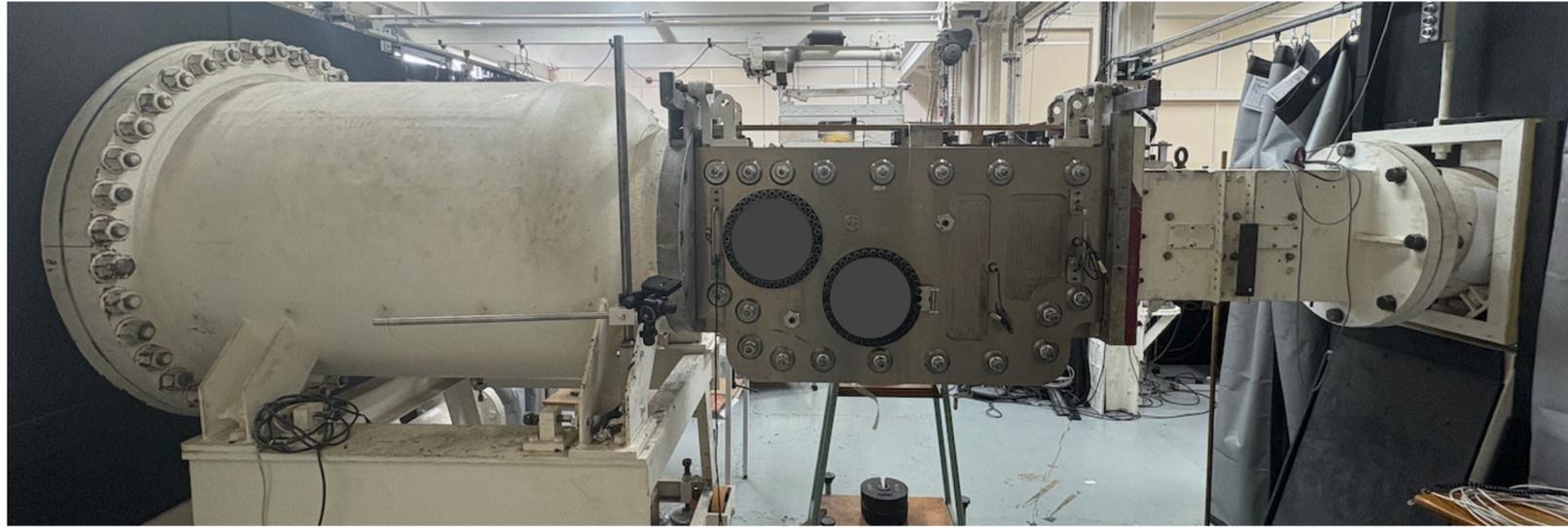
Test Facilities at Cambridge

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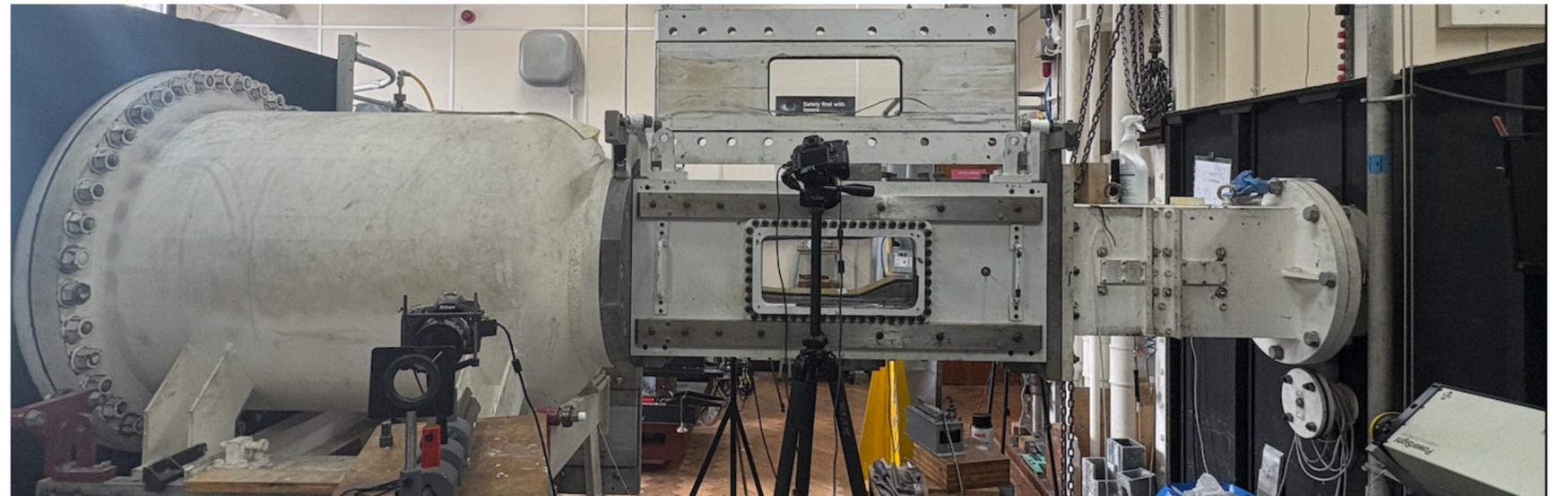


Intake studies at transonic Mach numbers

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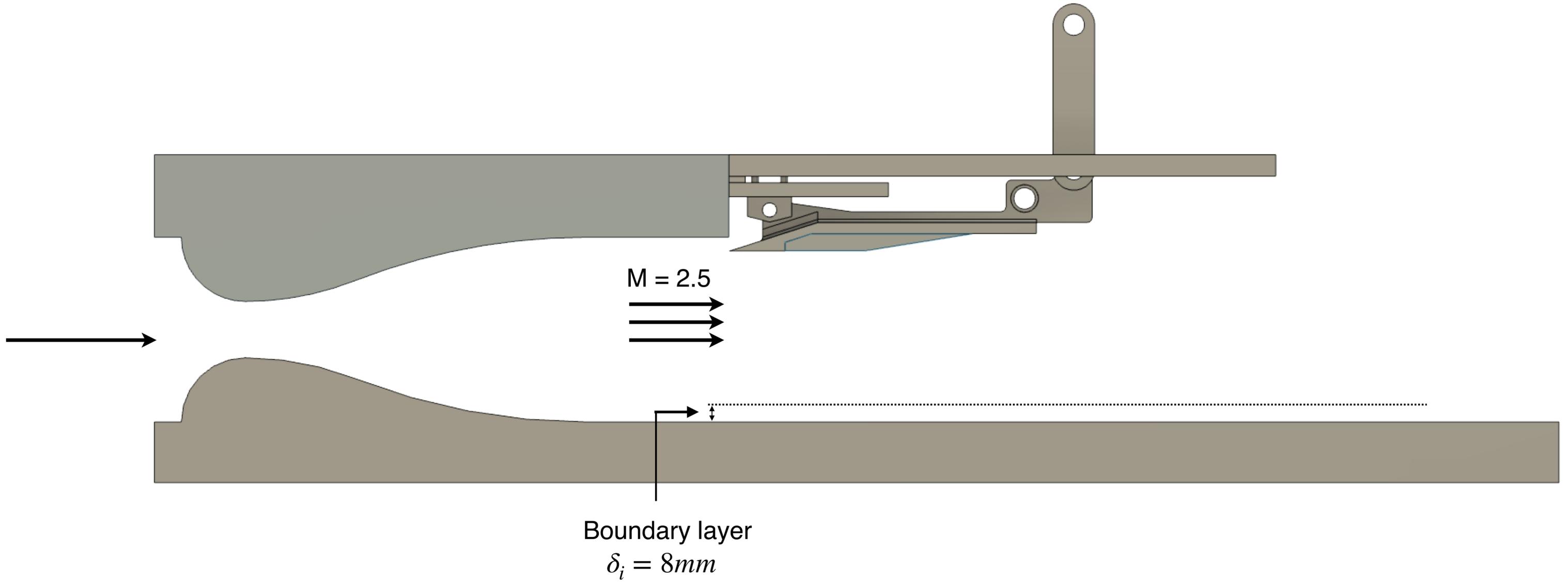
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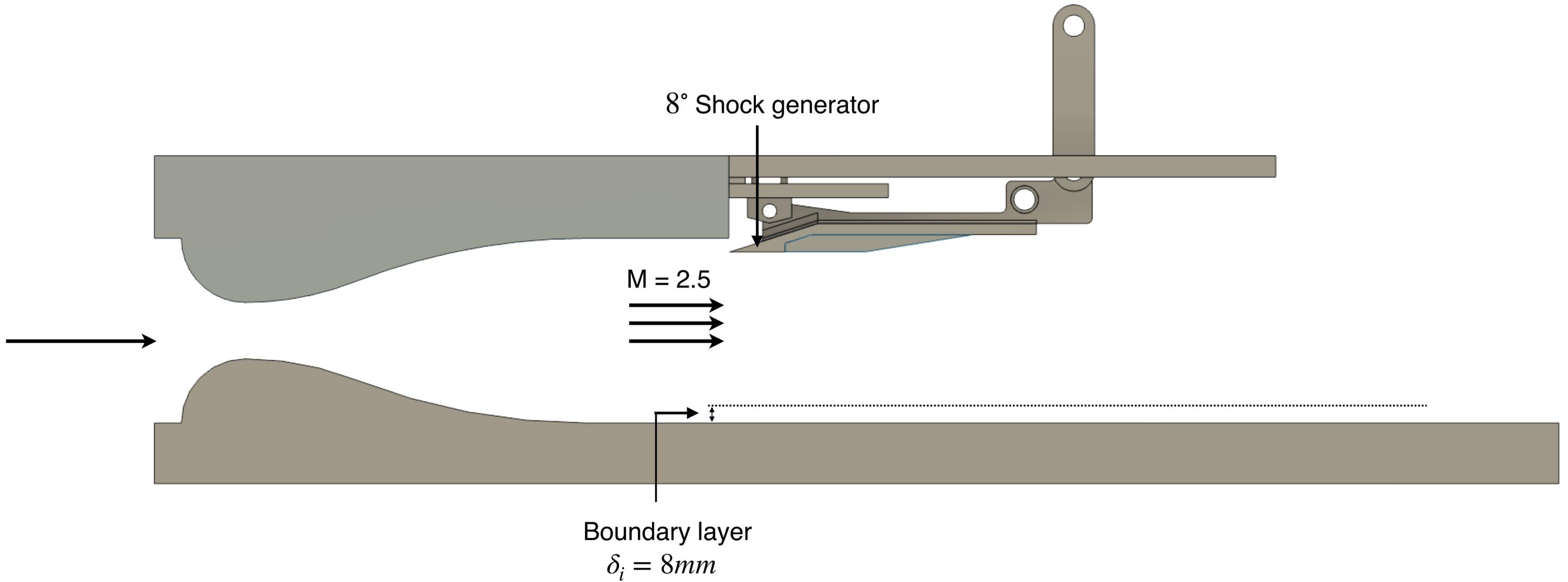
Fundamental SBLI and flow control studies upto $M = 3.5$

Experimental Setup

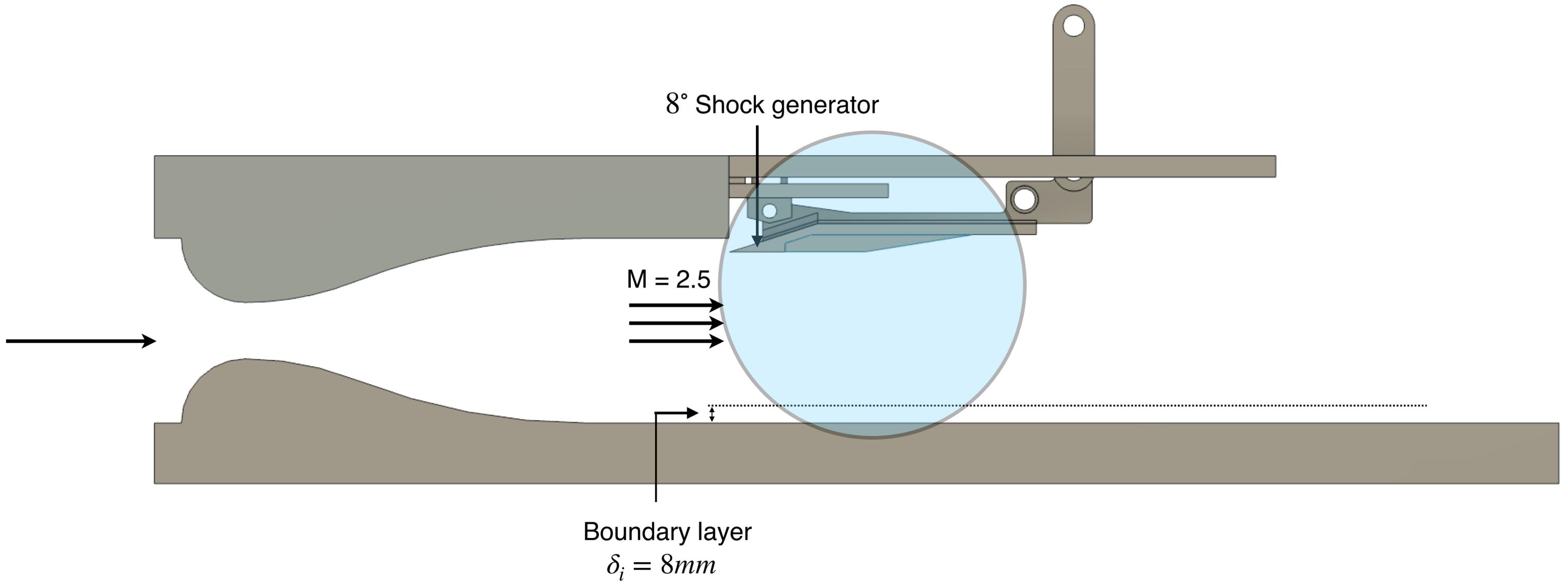
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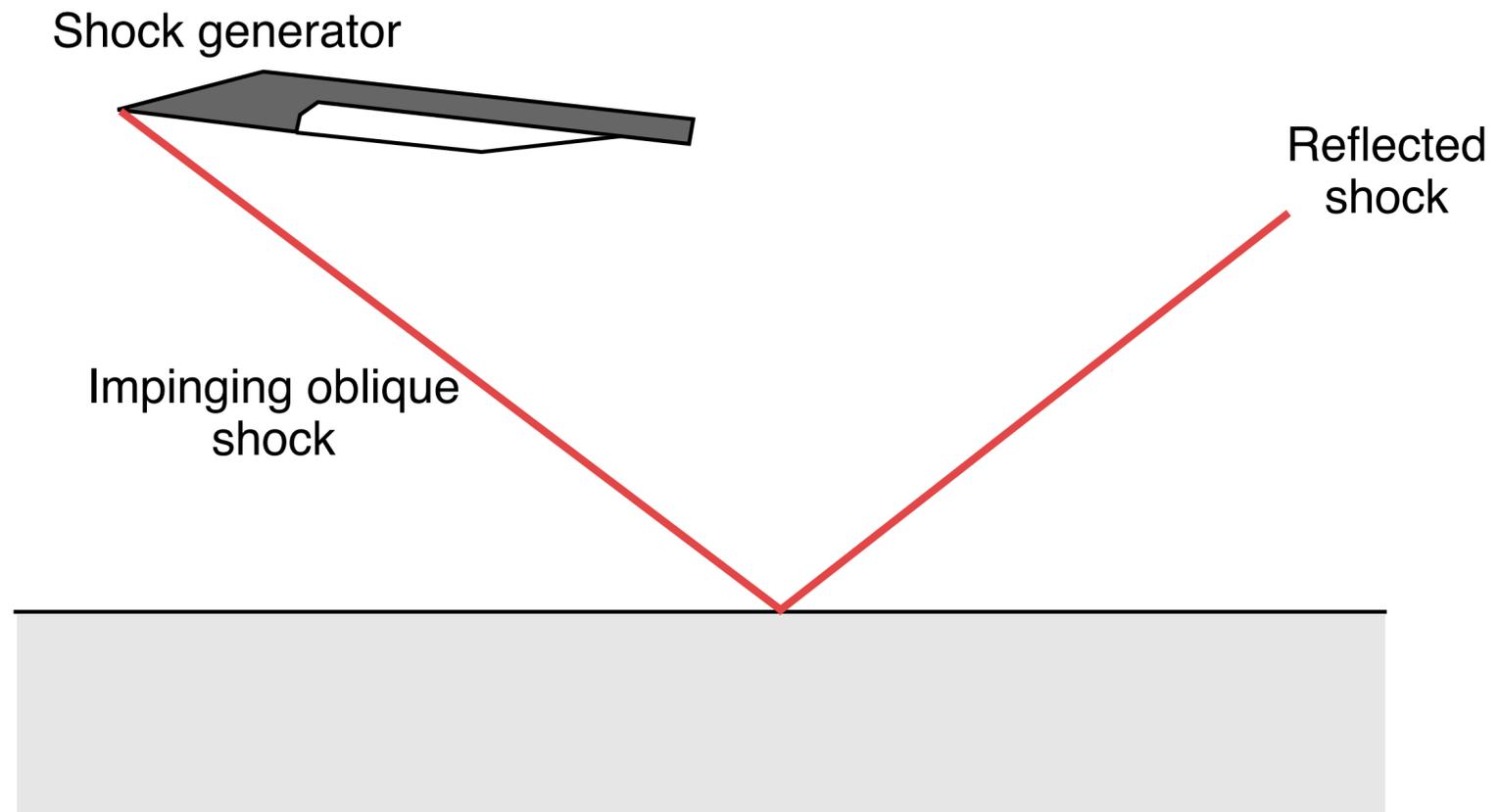


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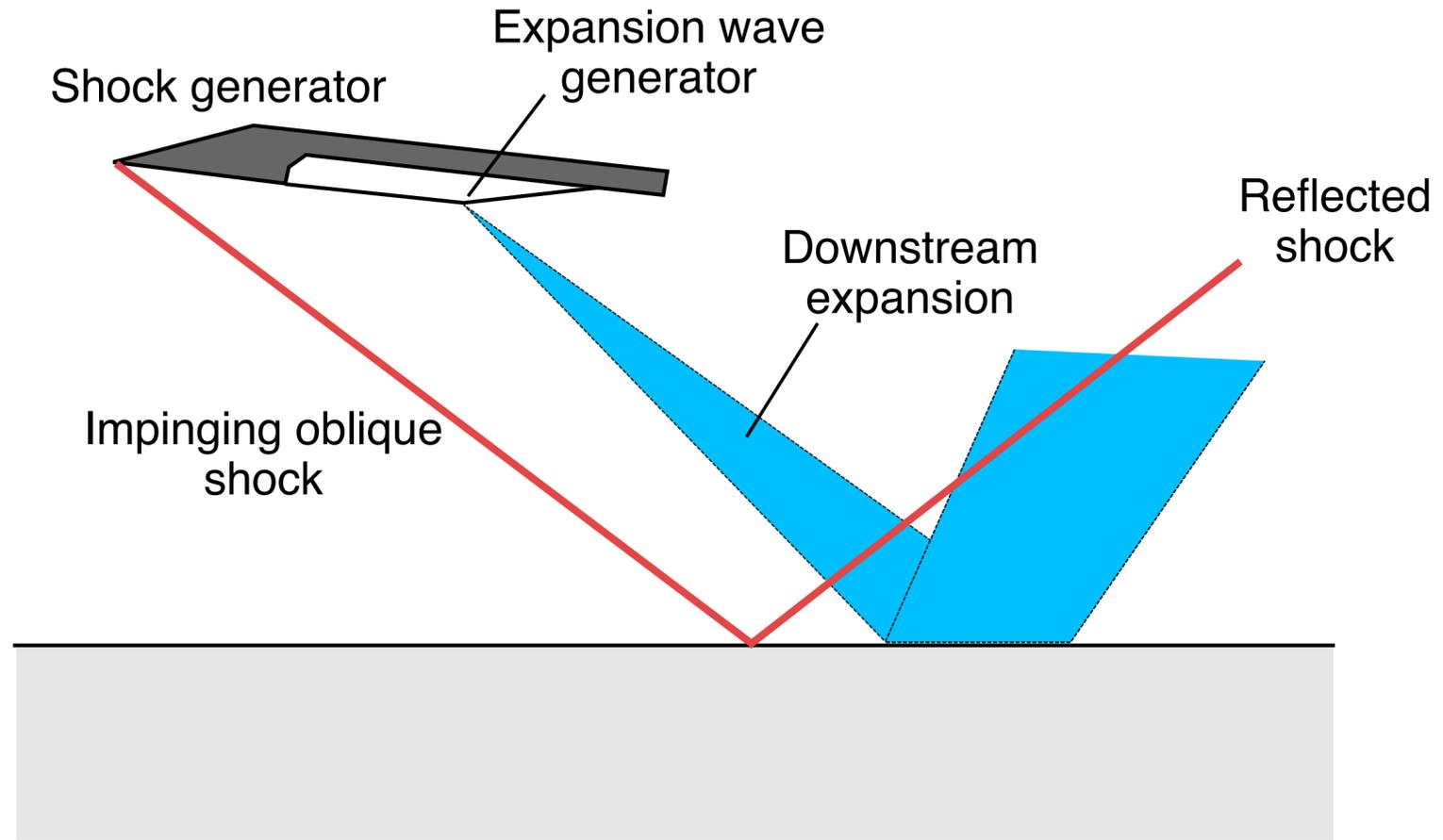


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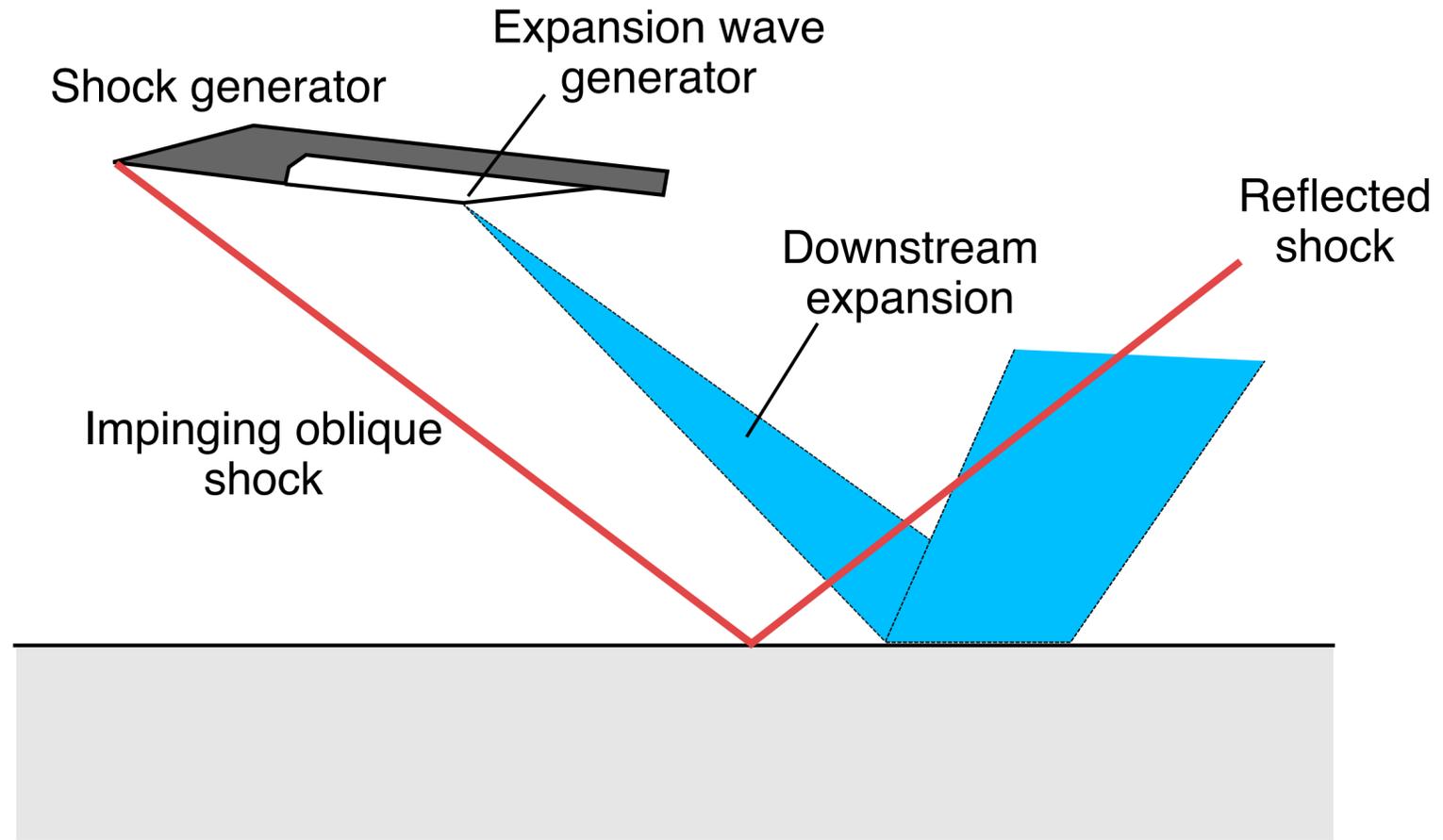
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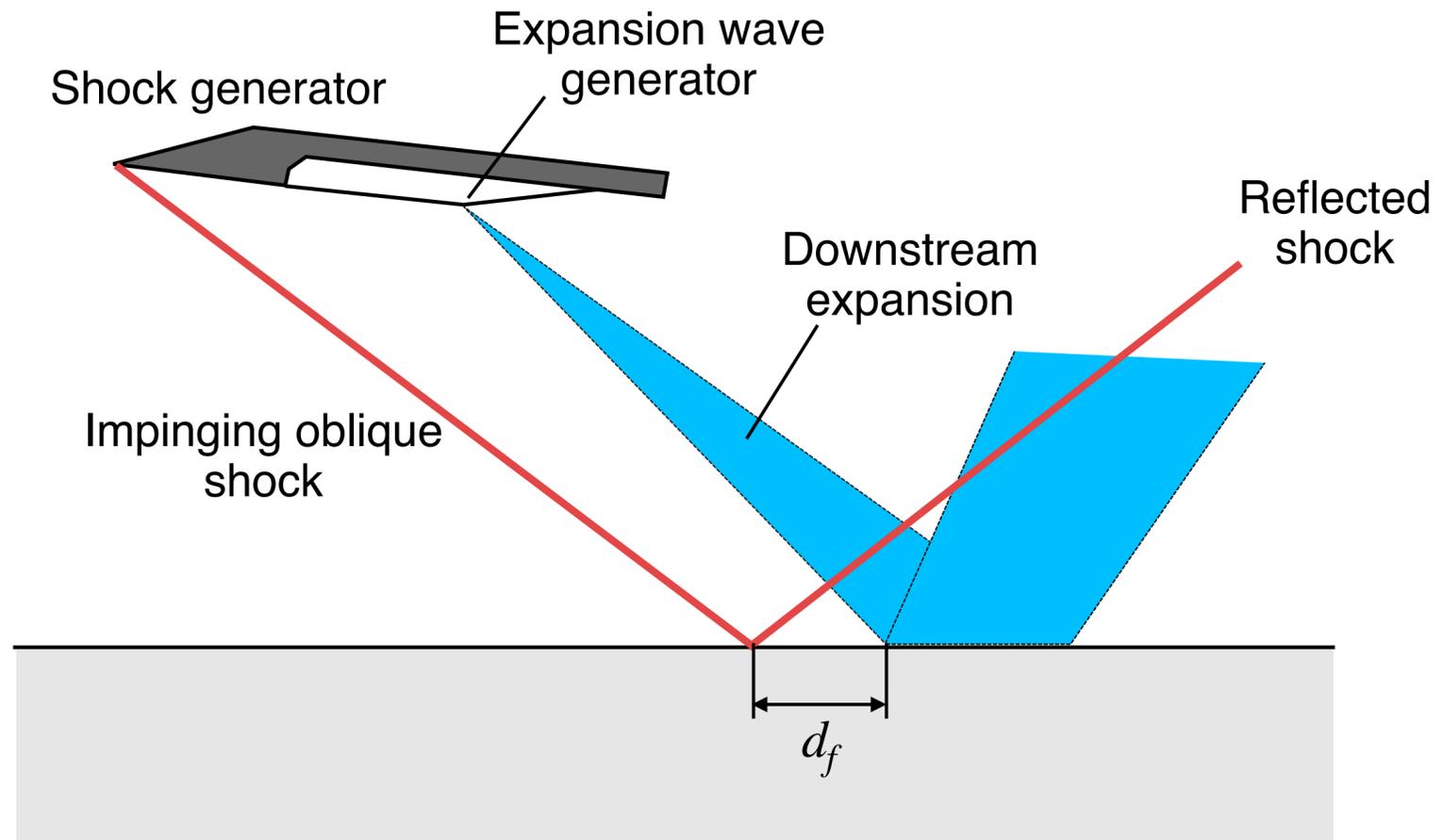


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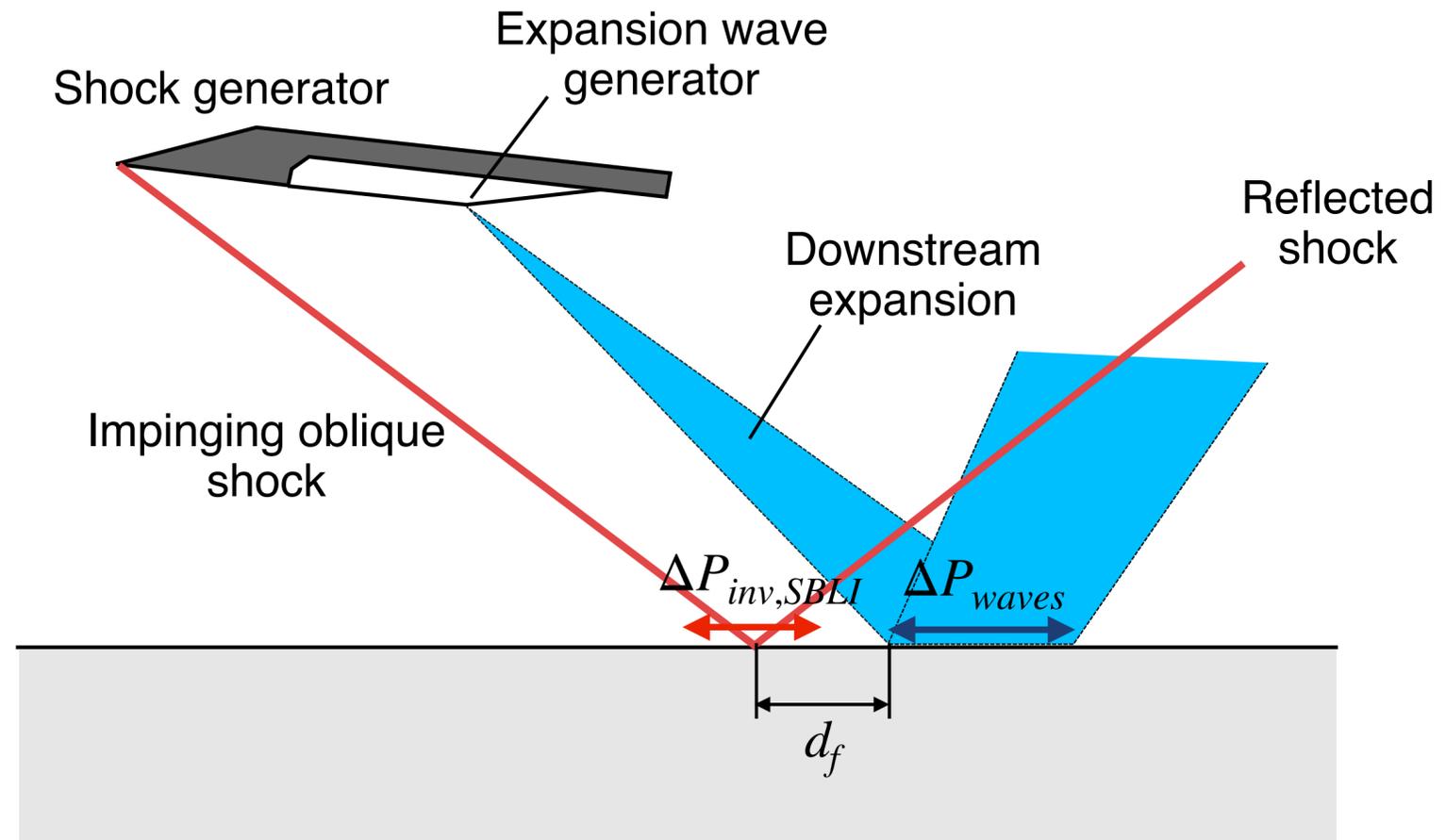
Variables:

Setup



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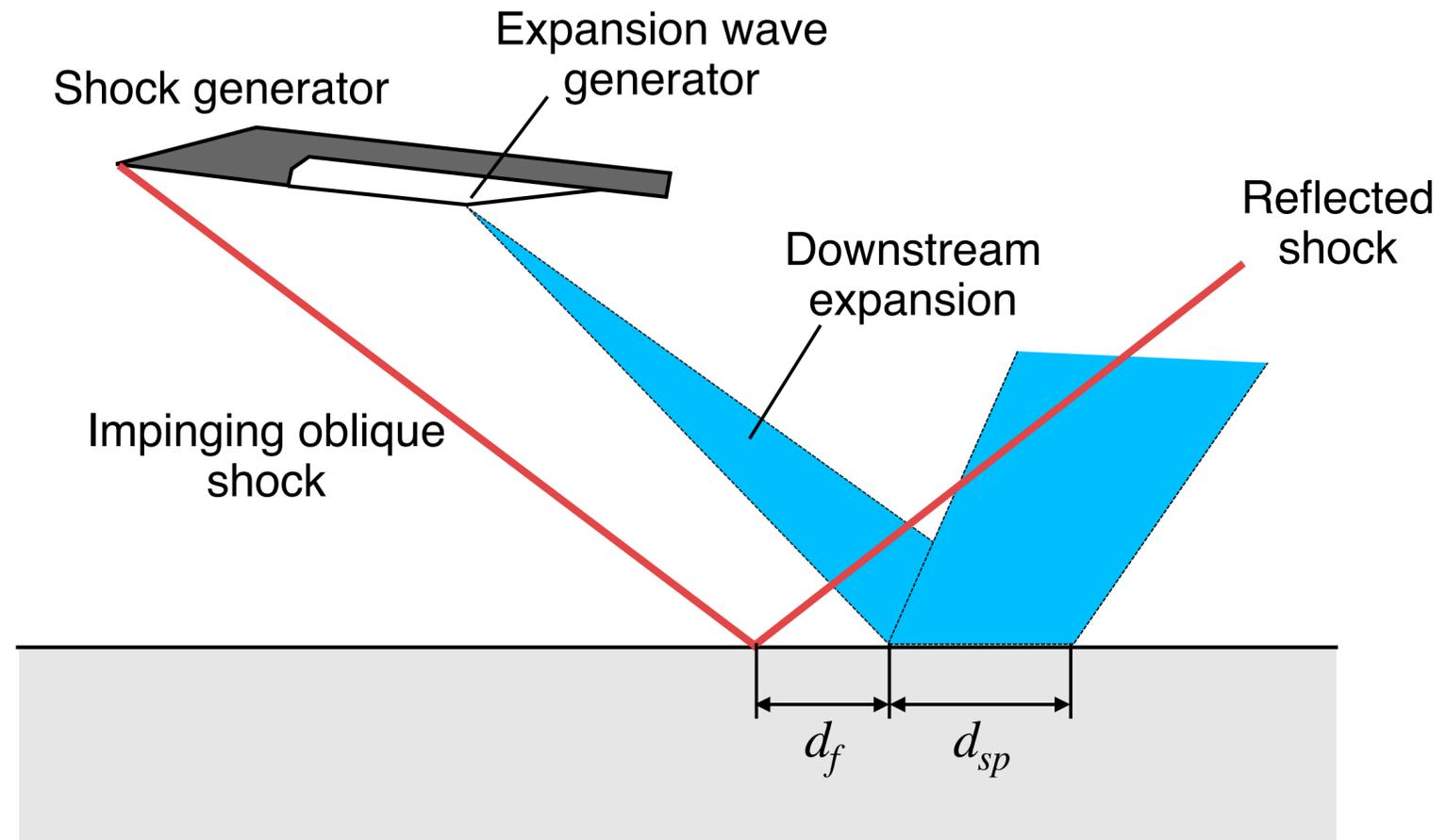
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$$\bar{\Delta P} = \Delta P_{waves} / \Delta P_{inv,SBLI}$$

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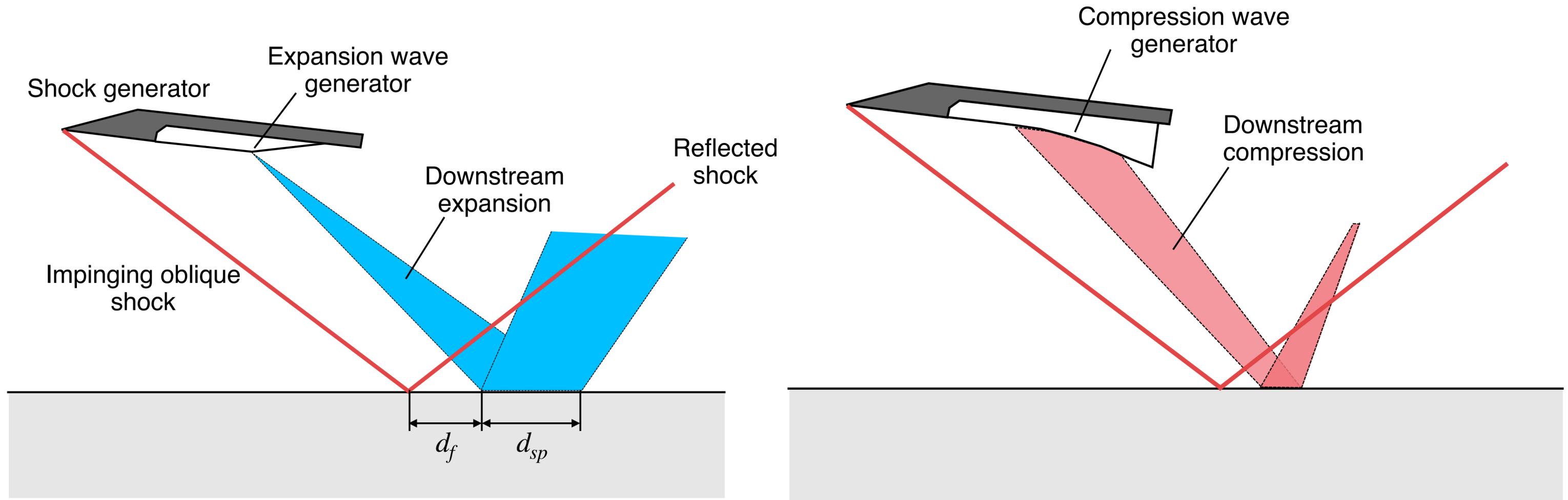


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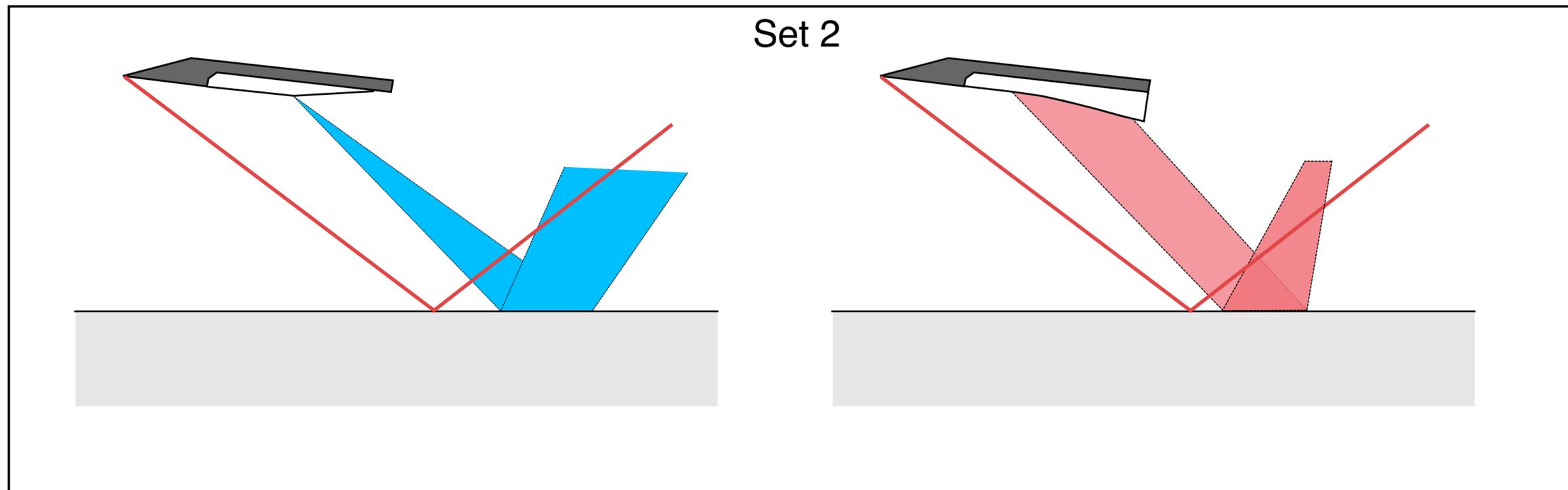
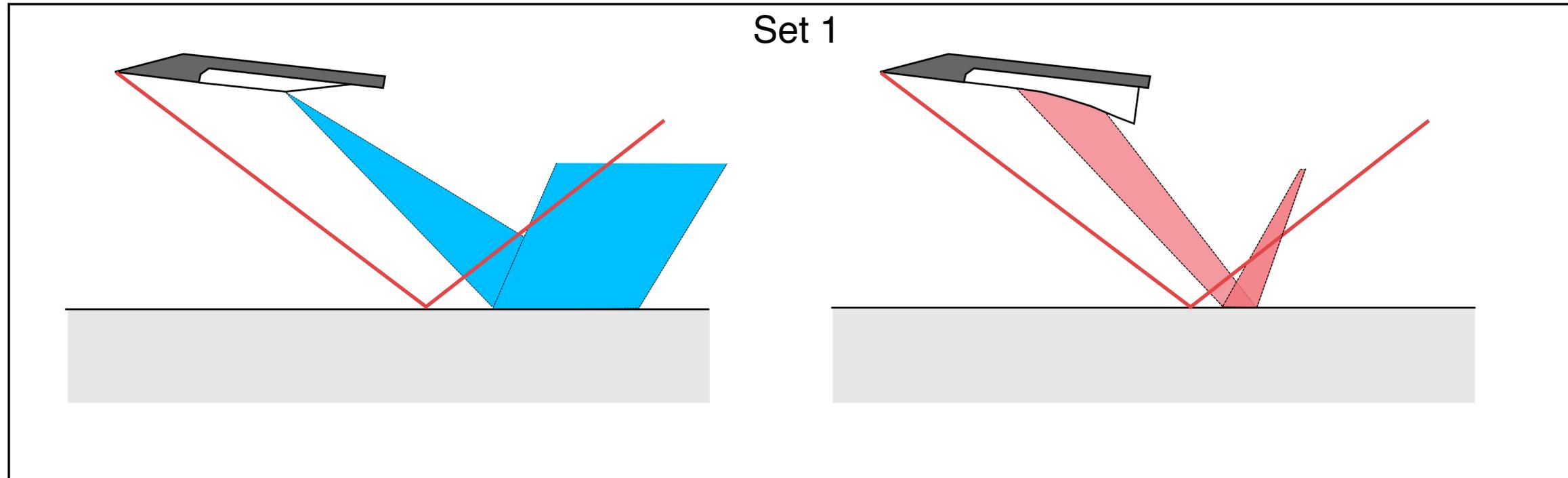
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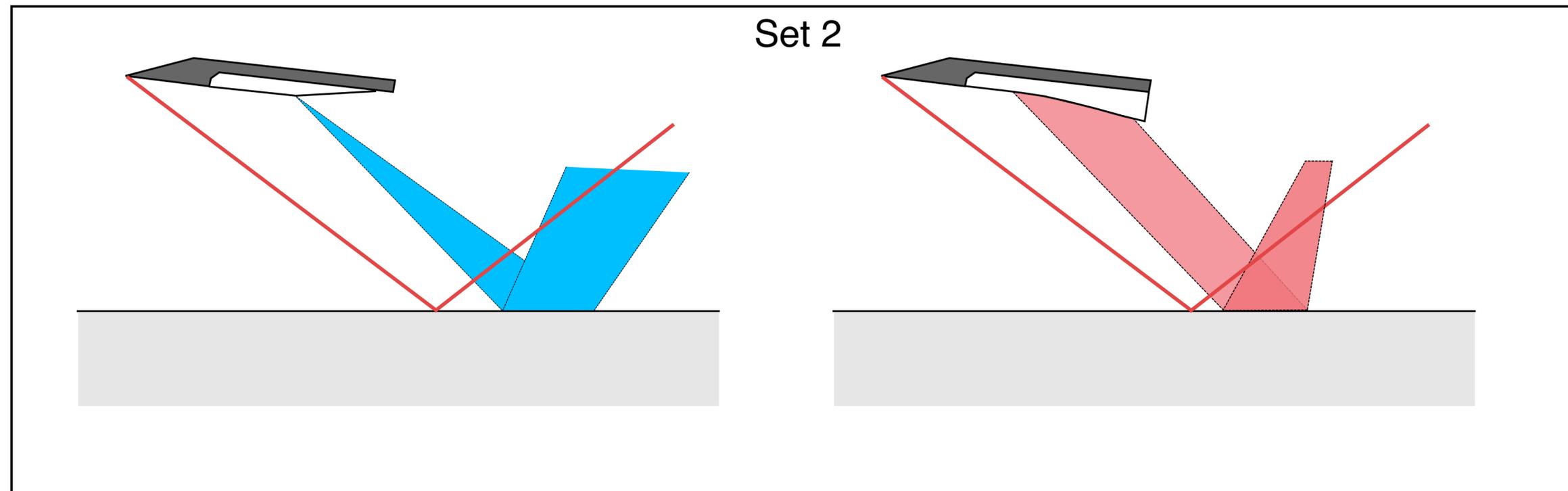
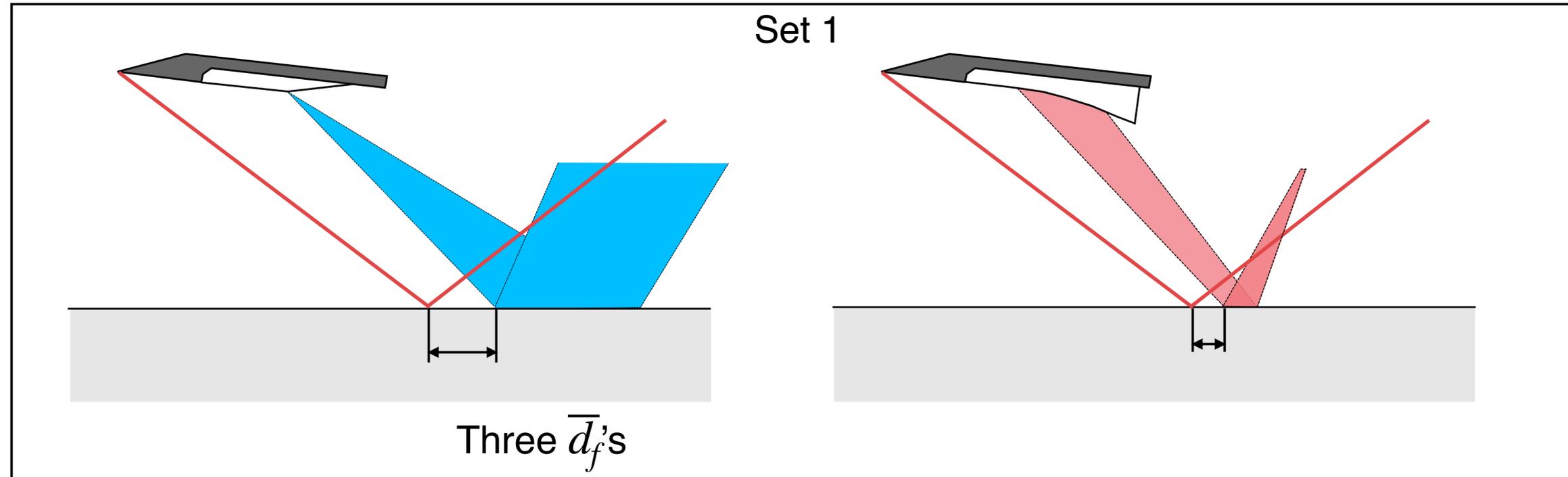
Set 1

Set 2

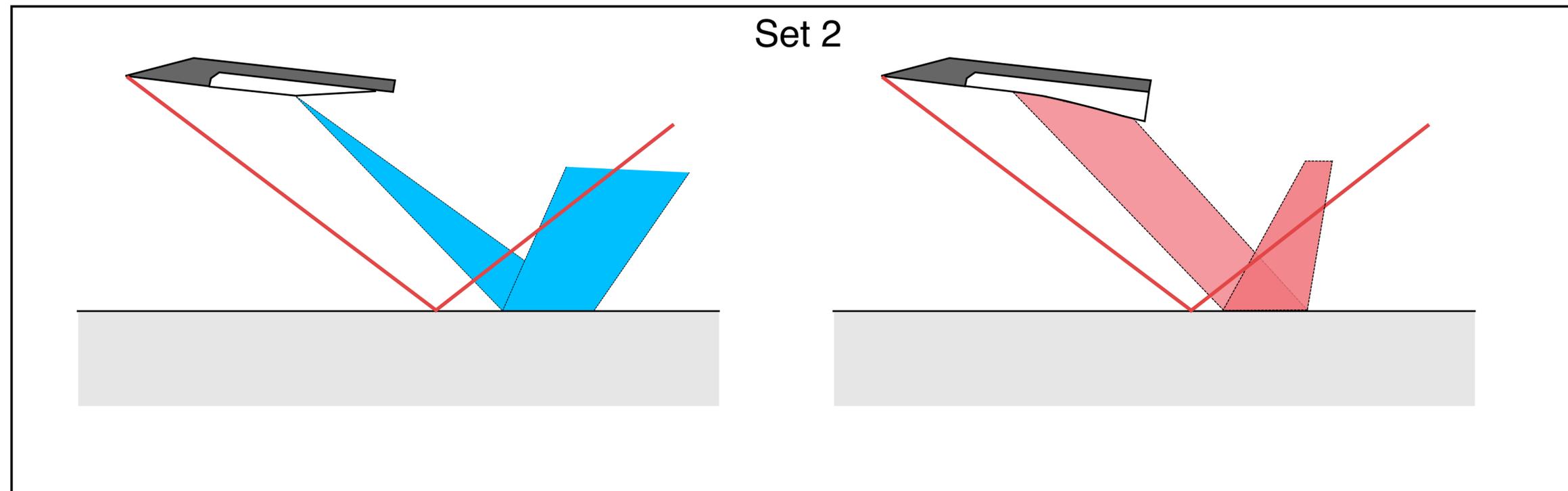
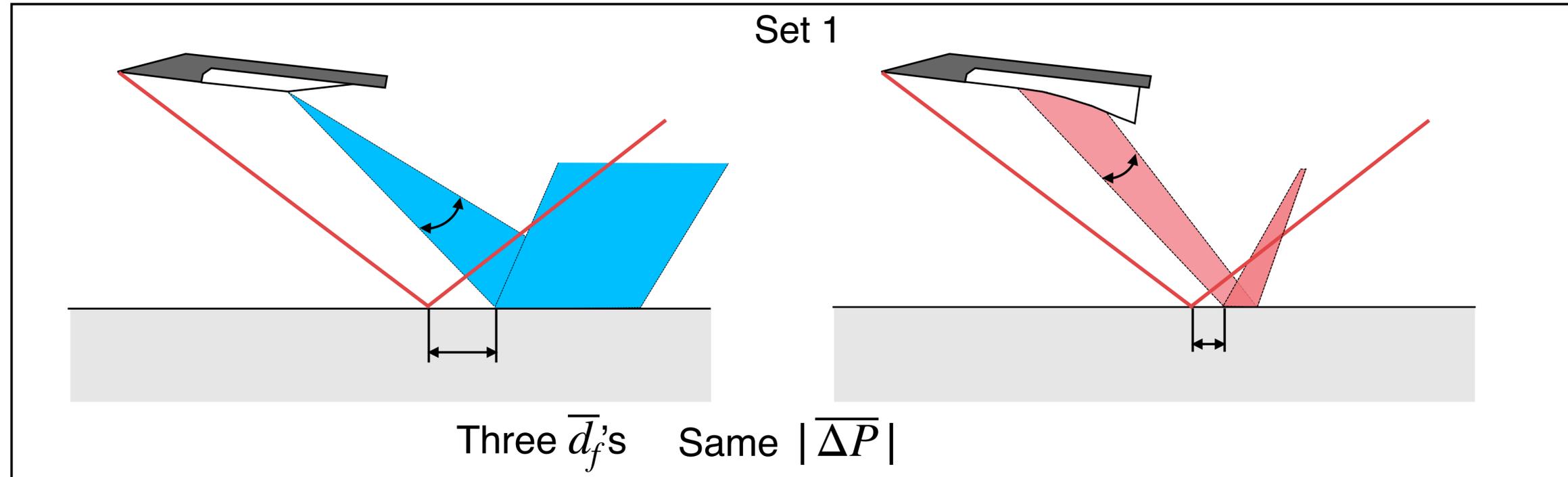
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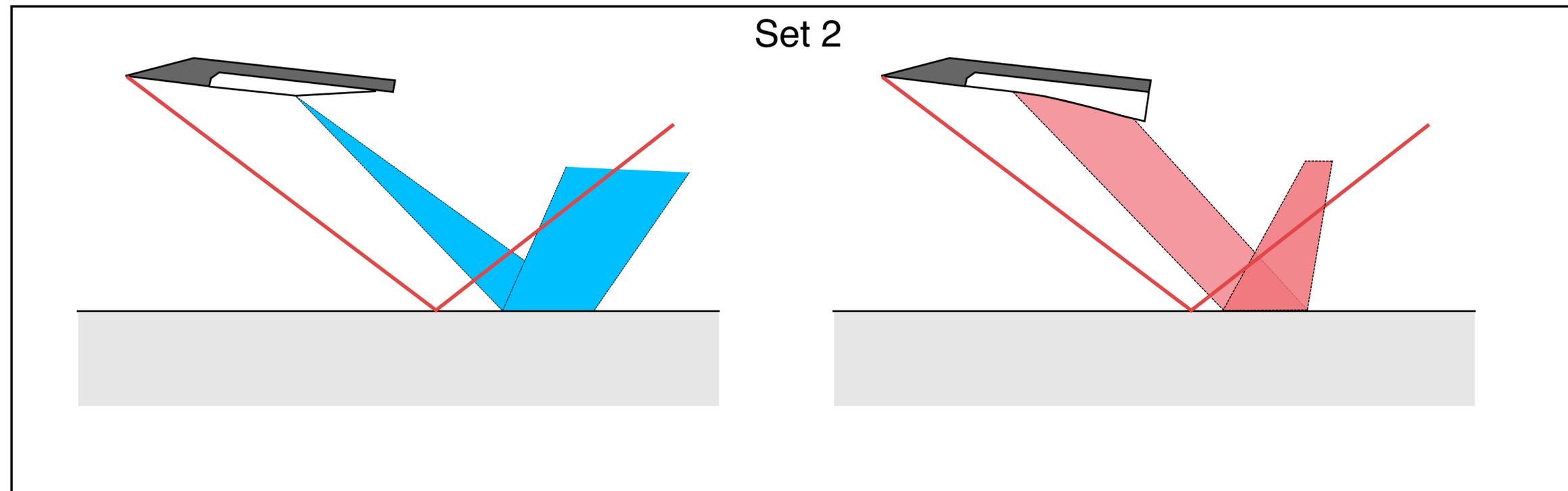
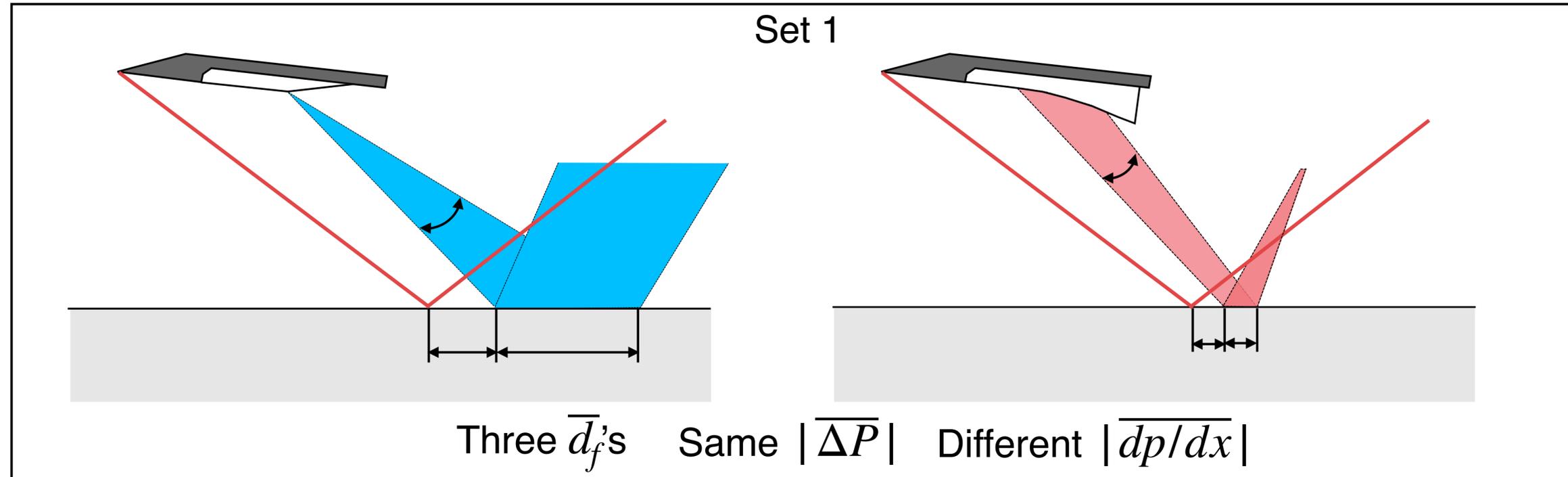
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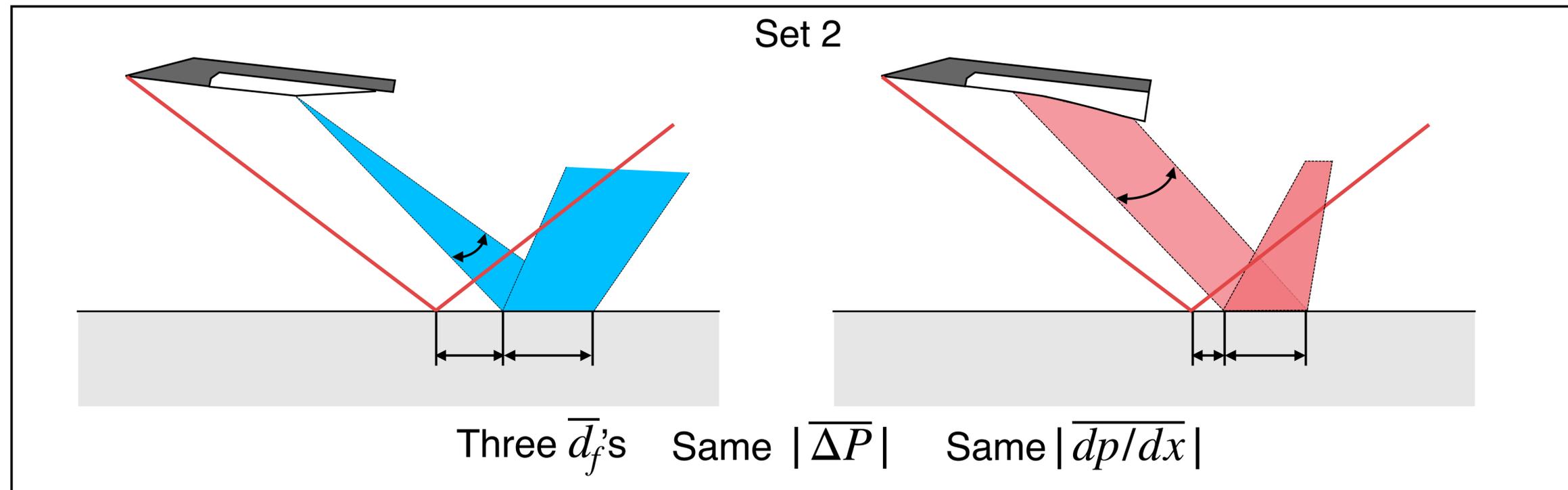
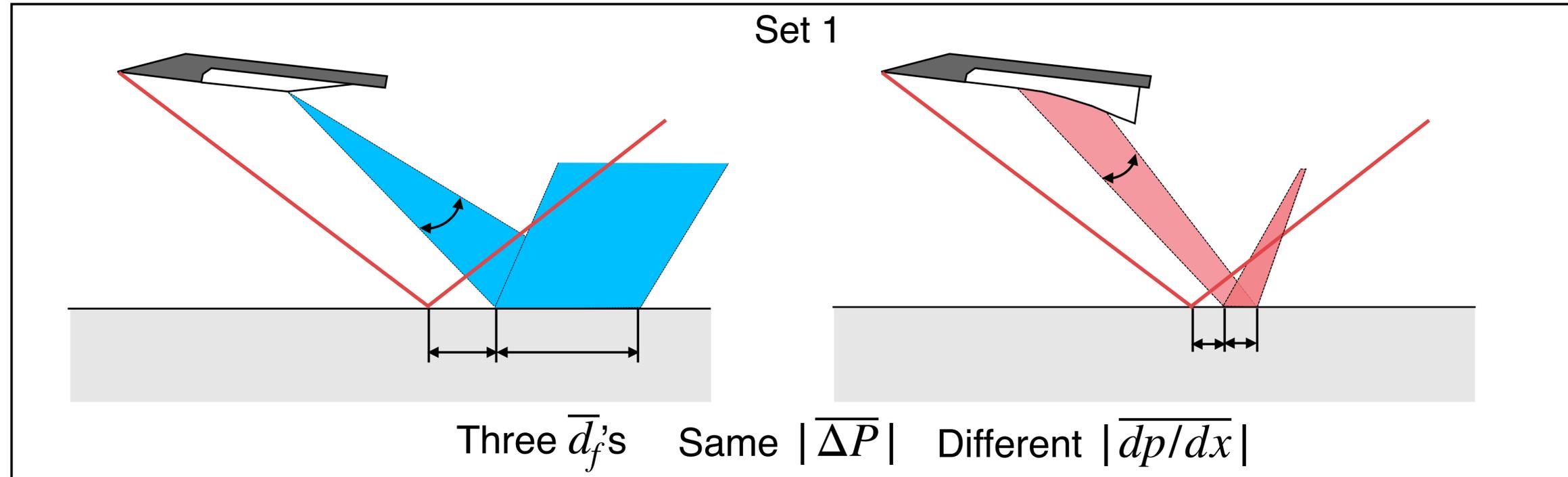
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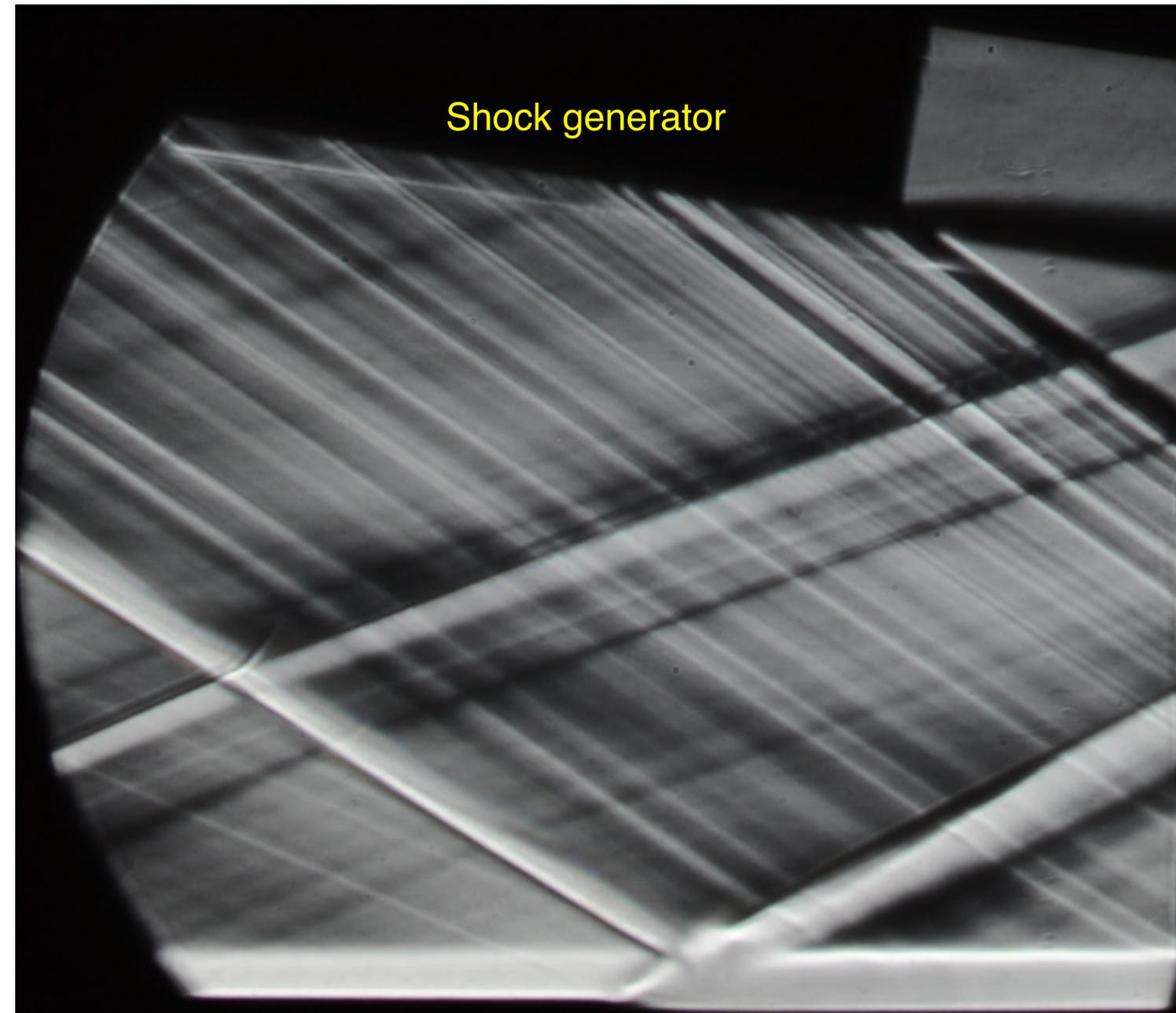
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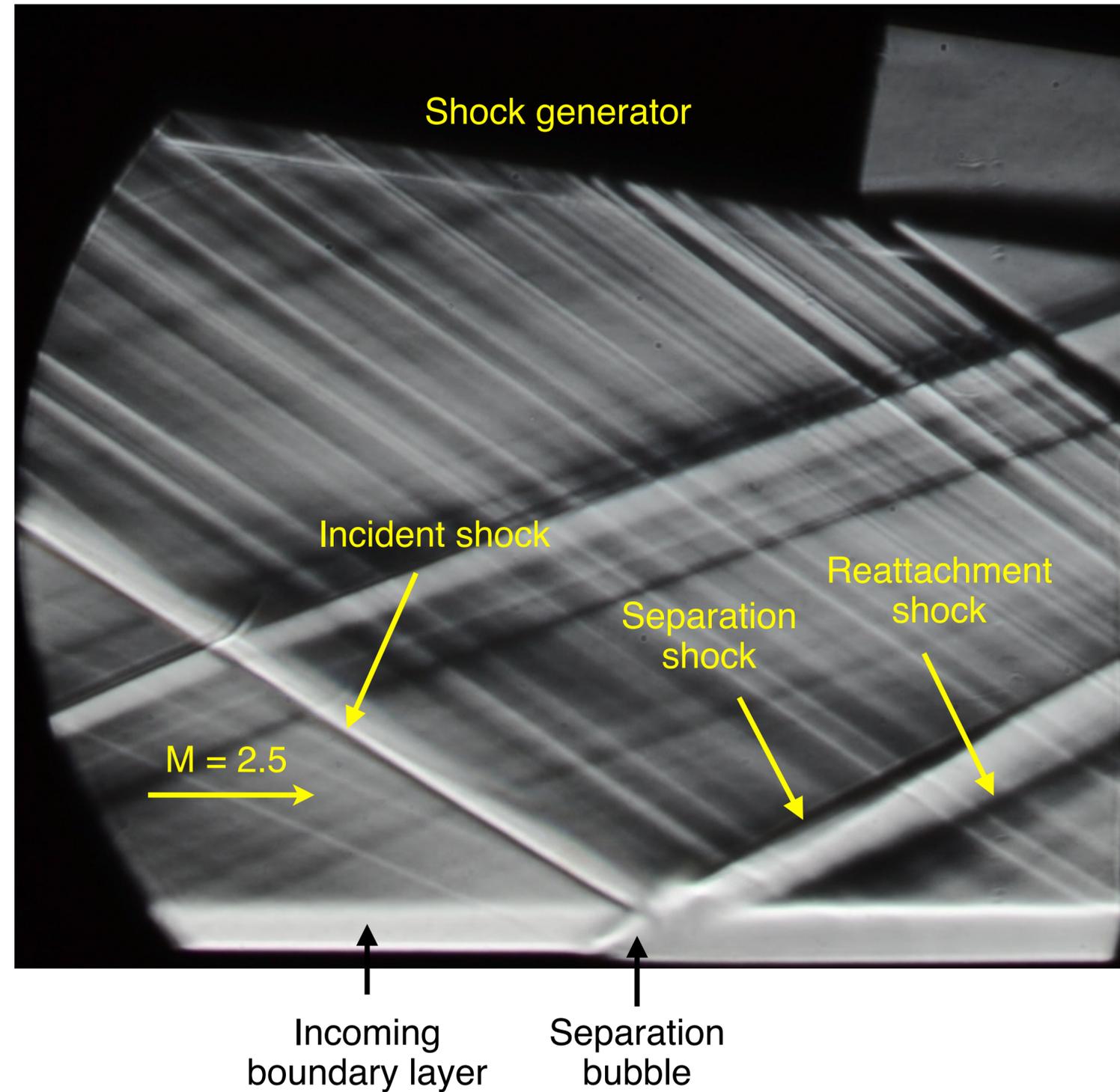
Setup



Baseline flowfield

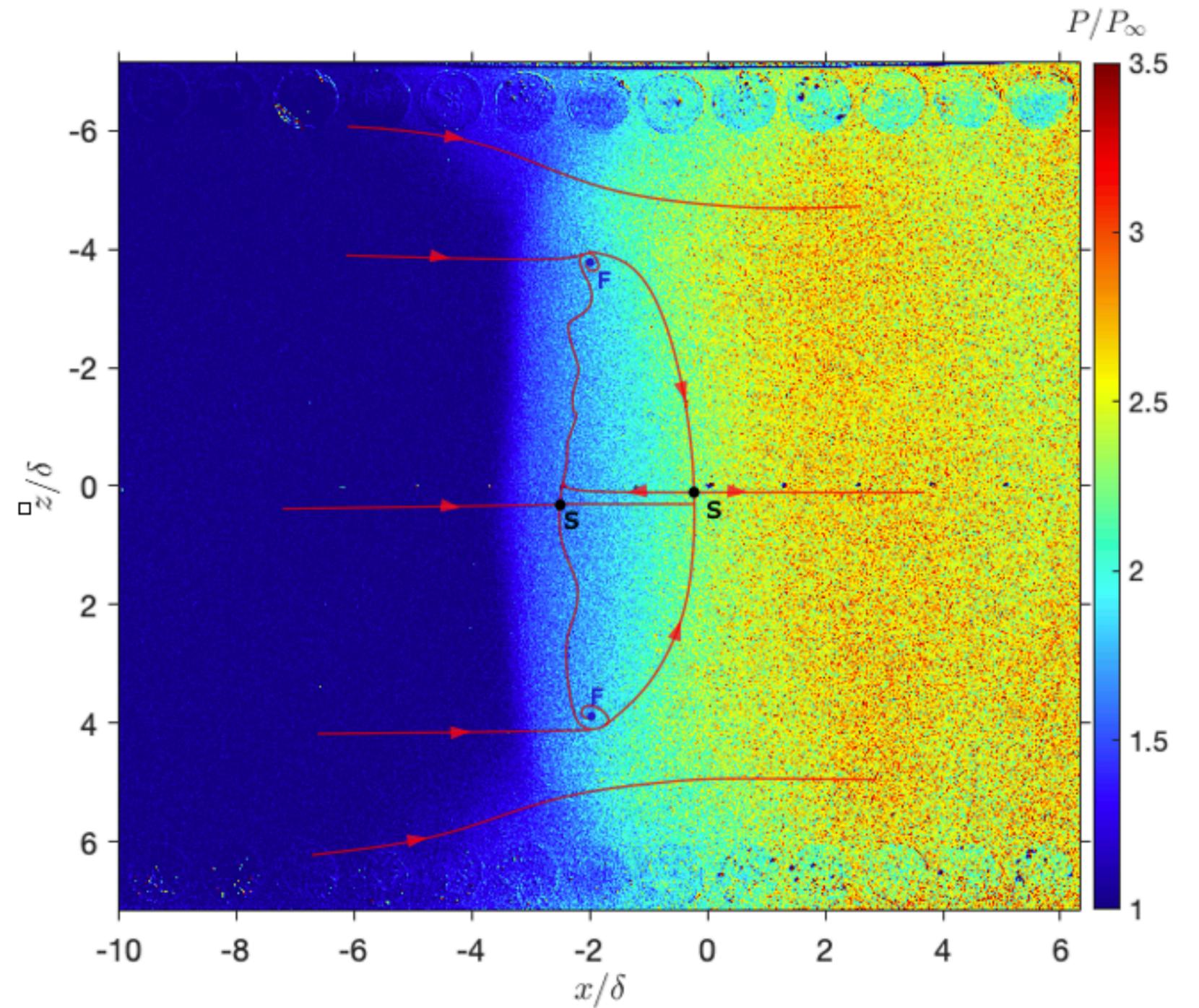
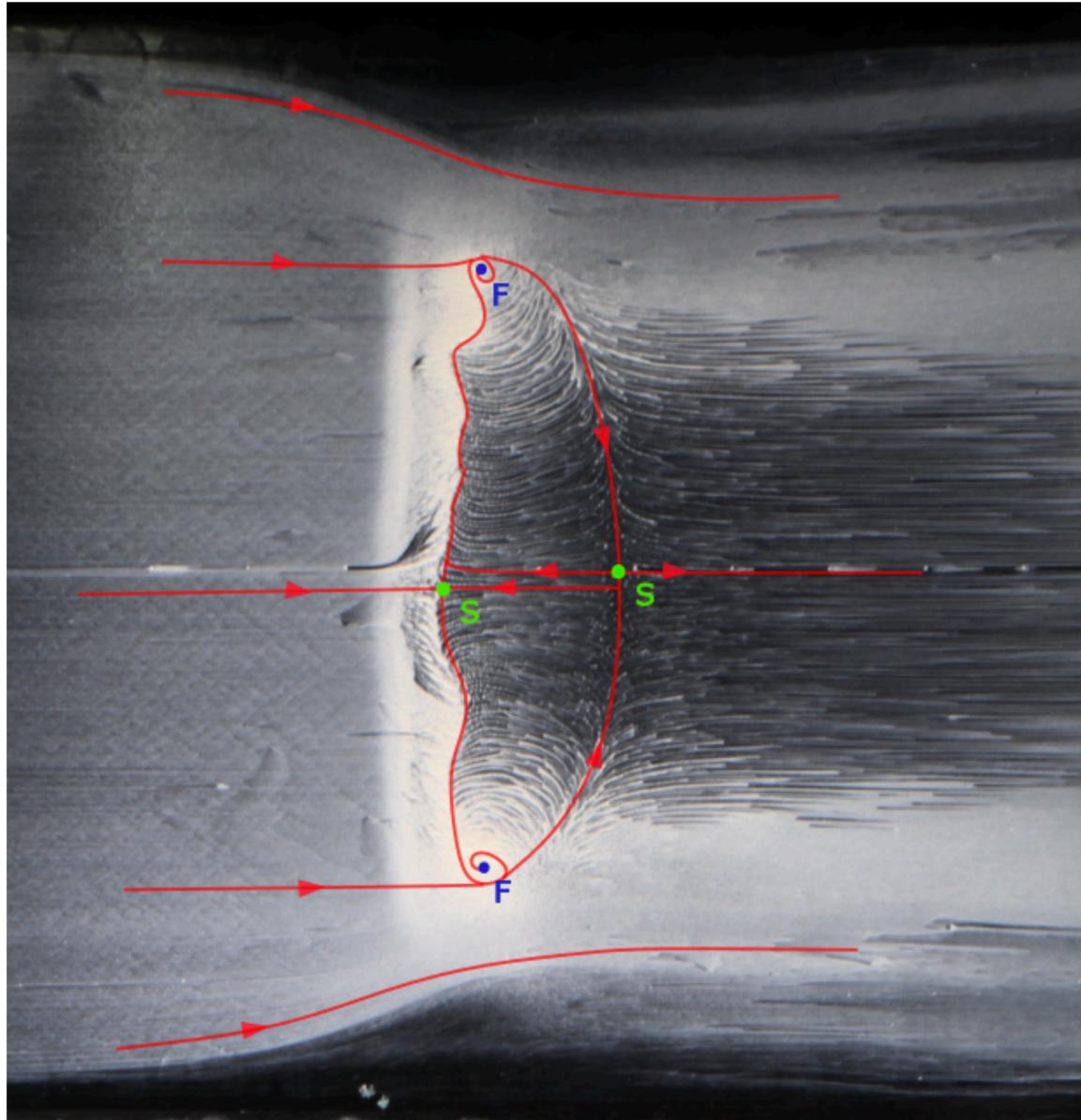


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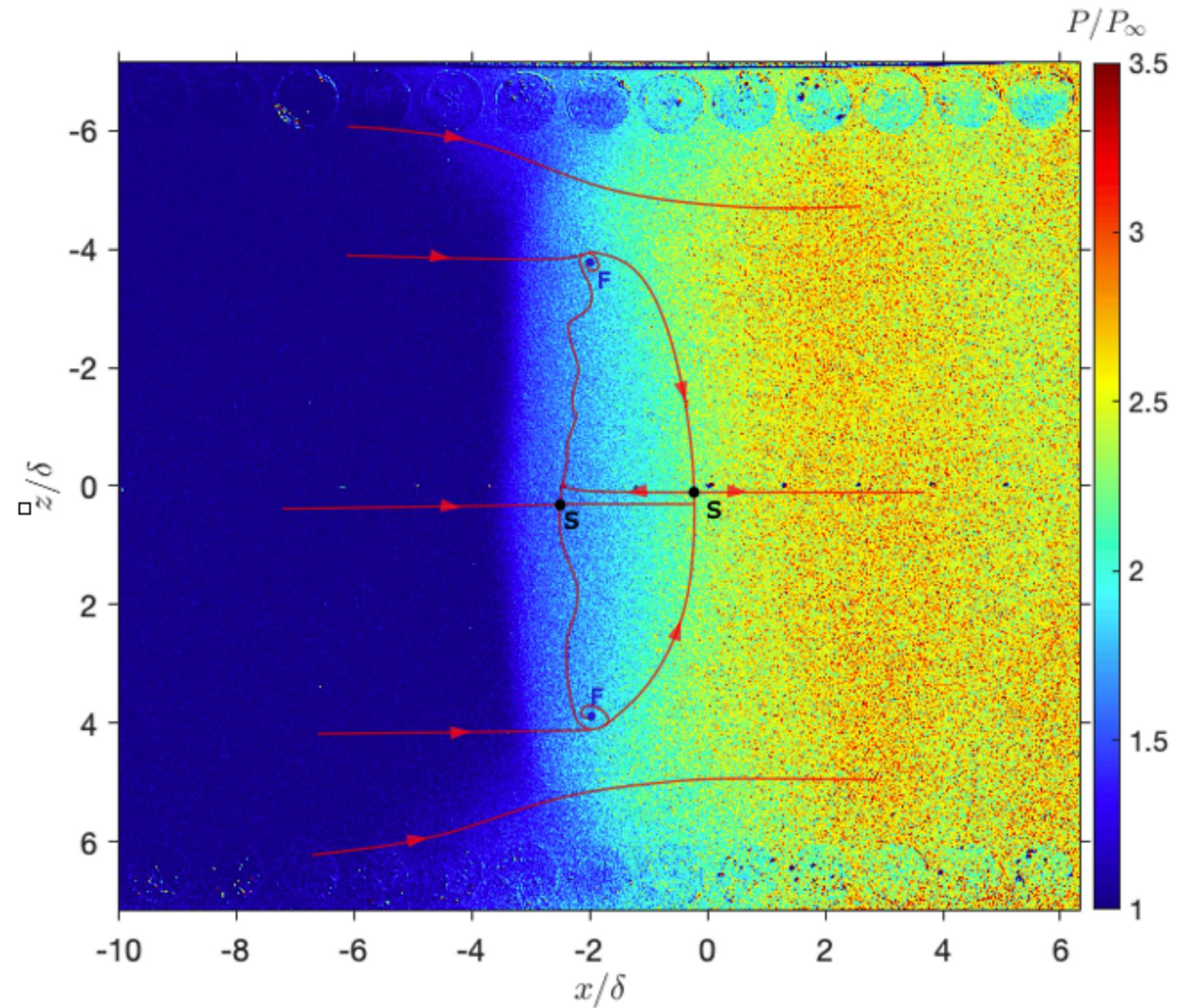
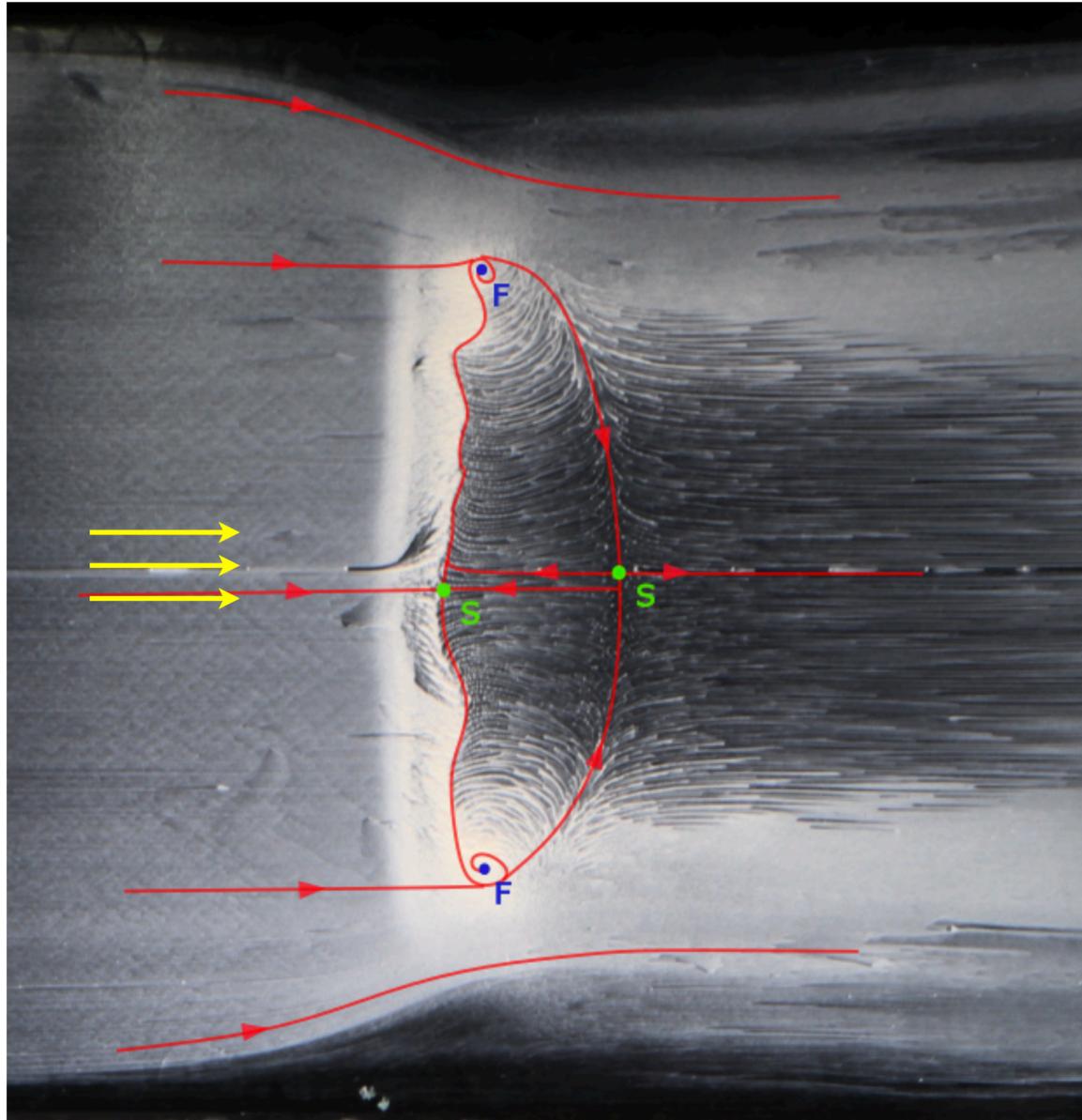


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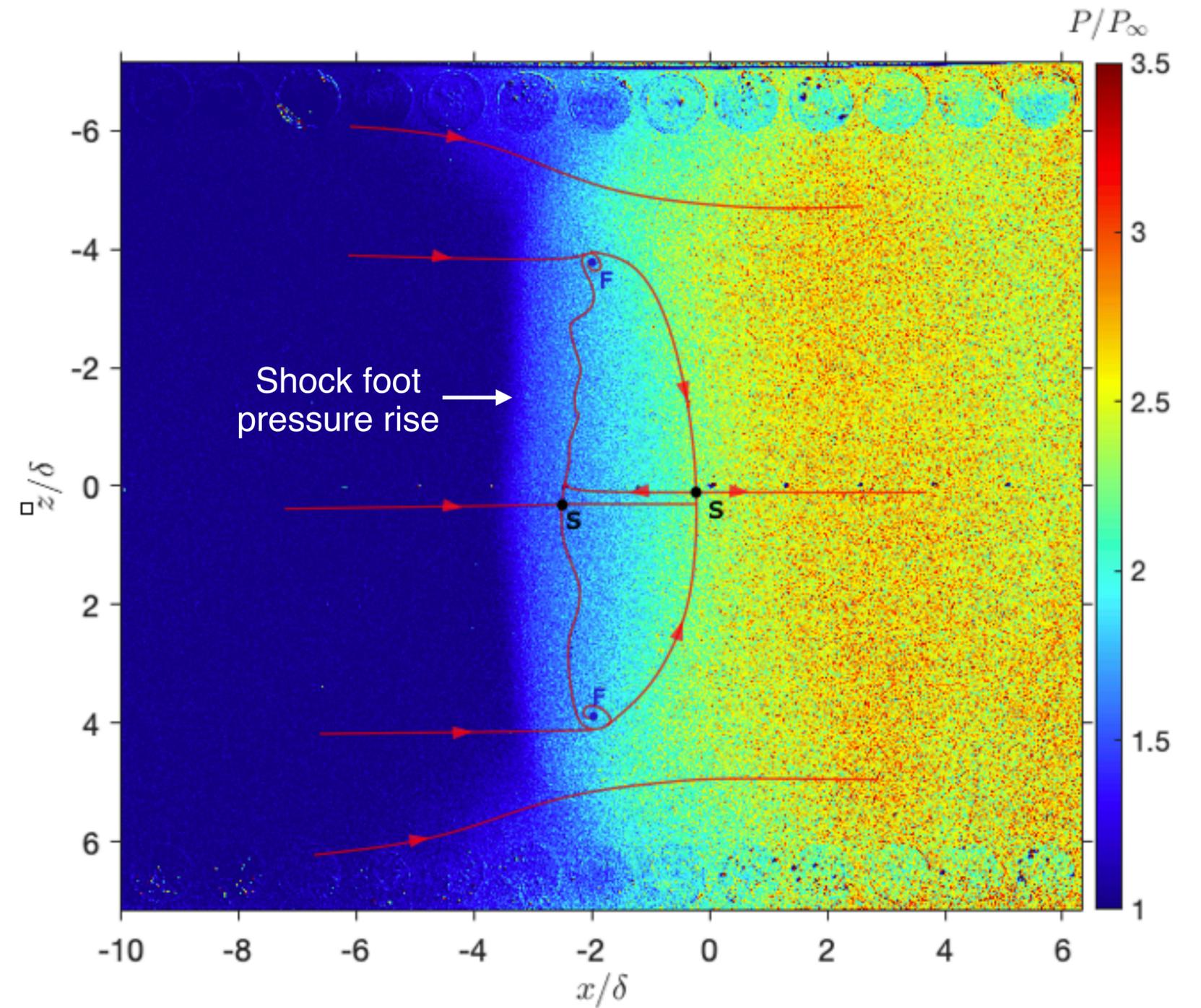
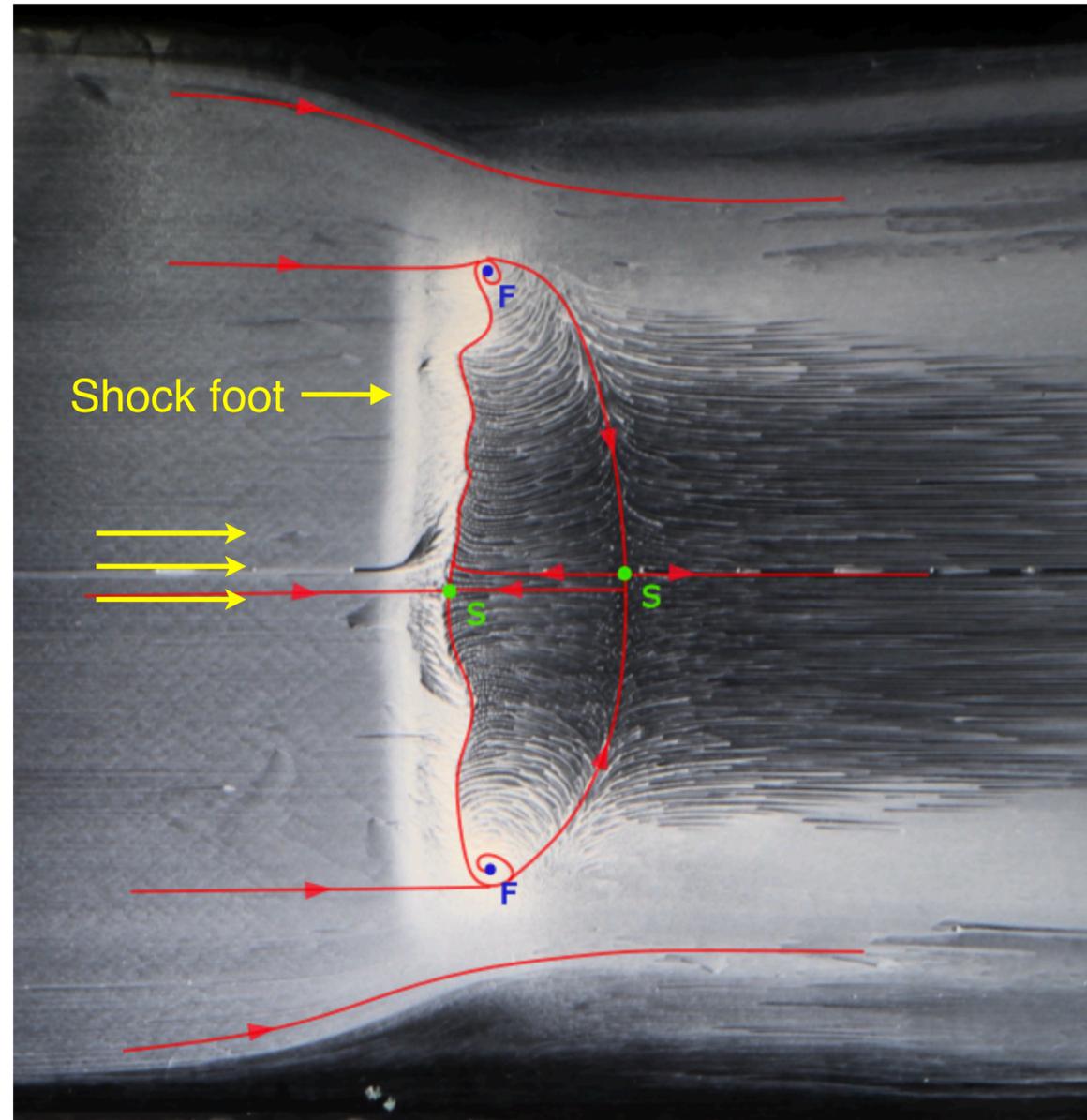
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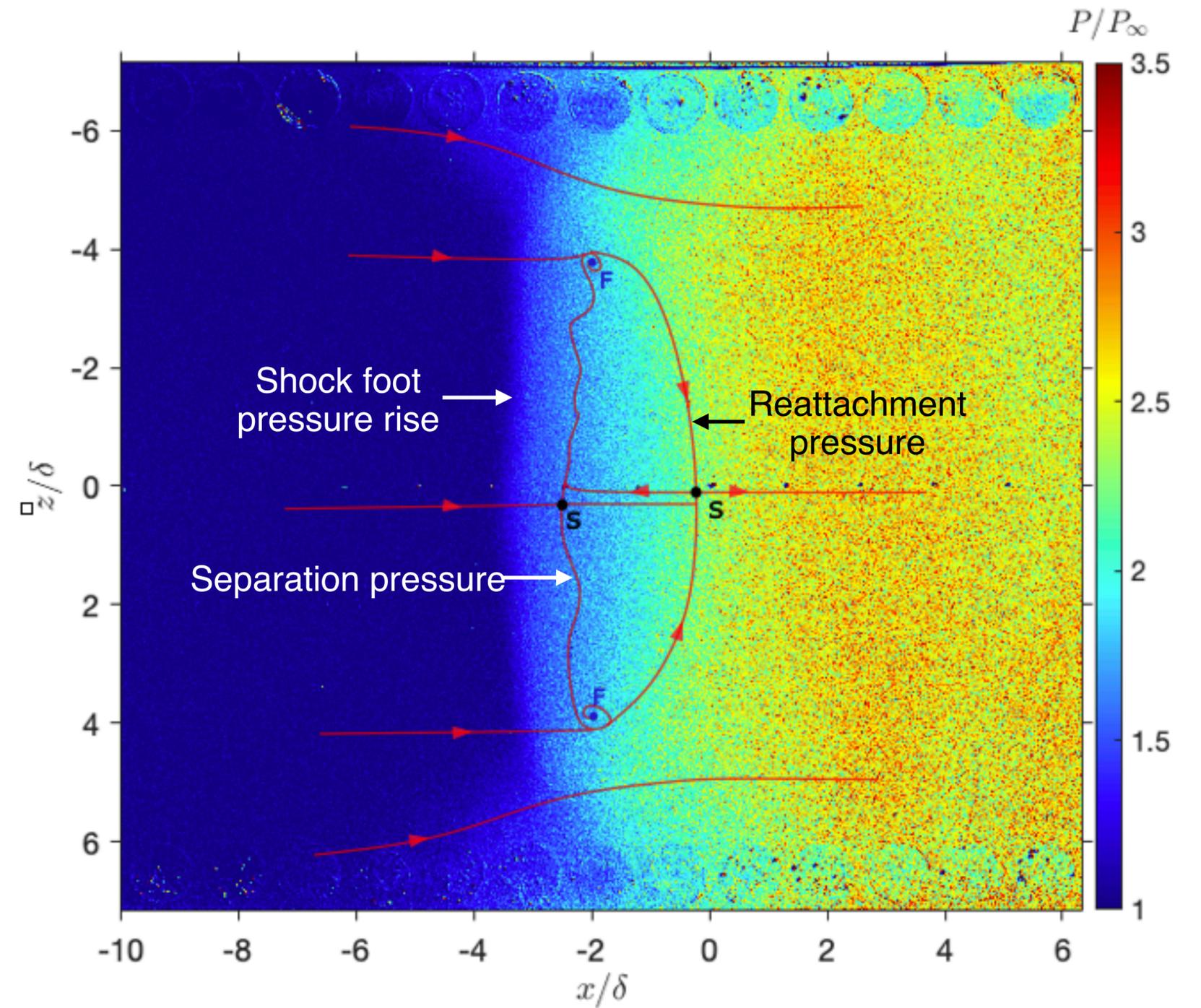
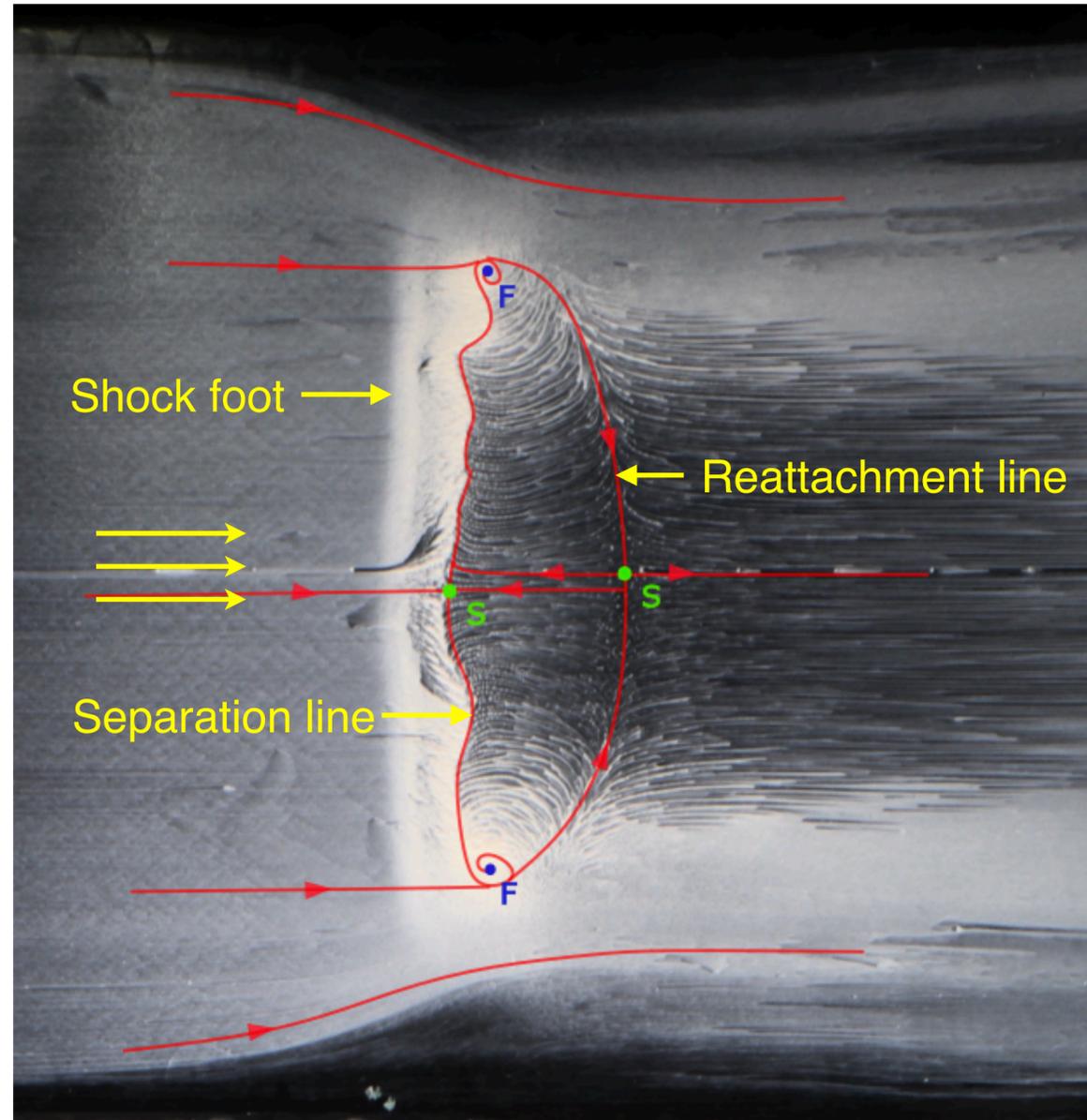
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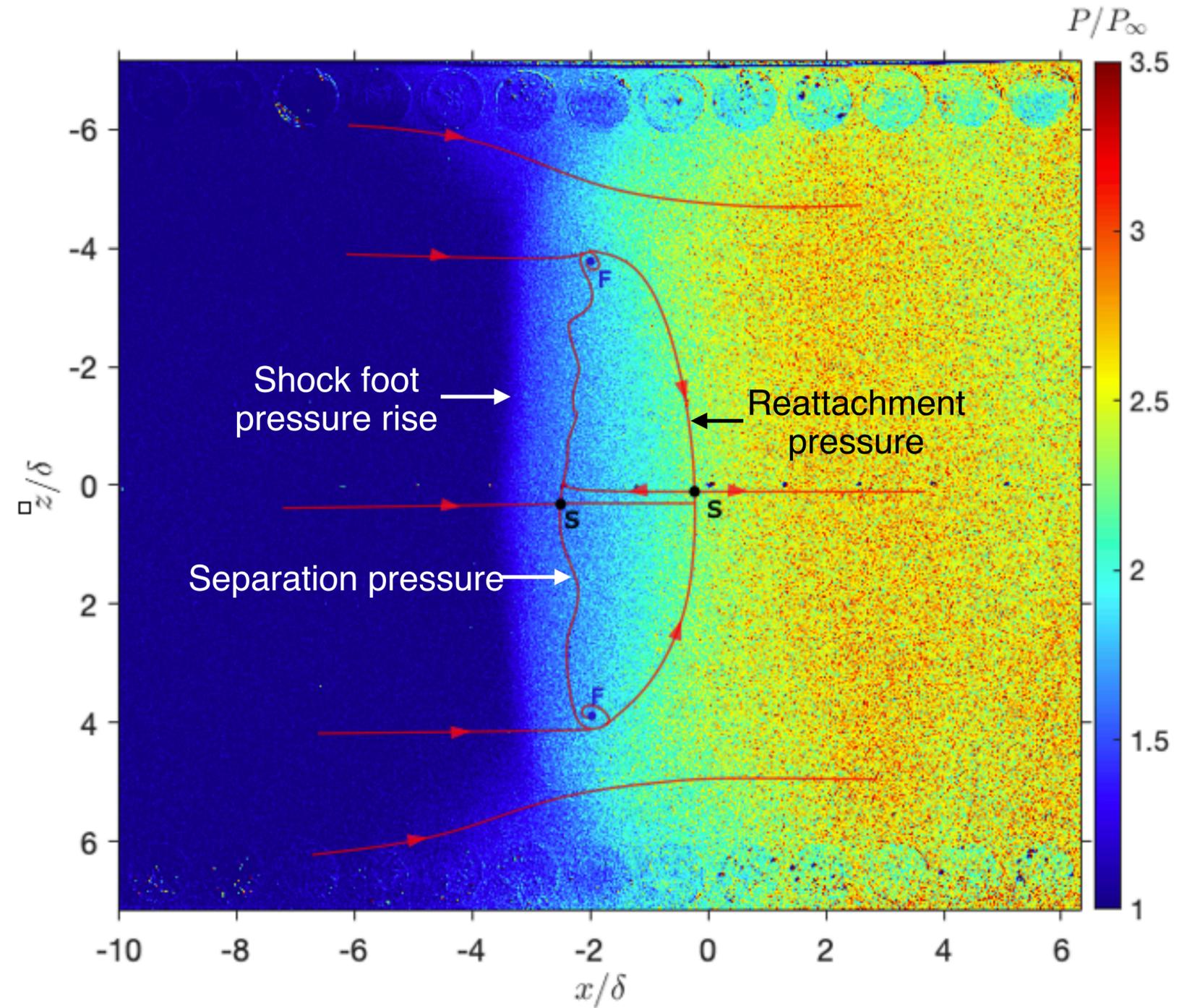
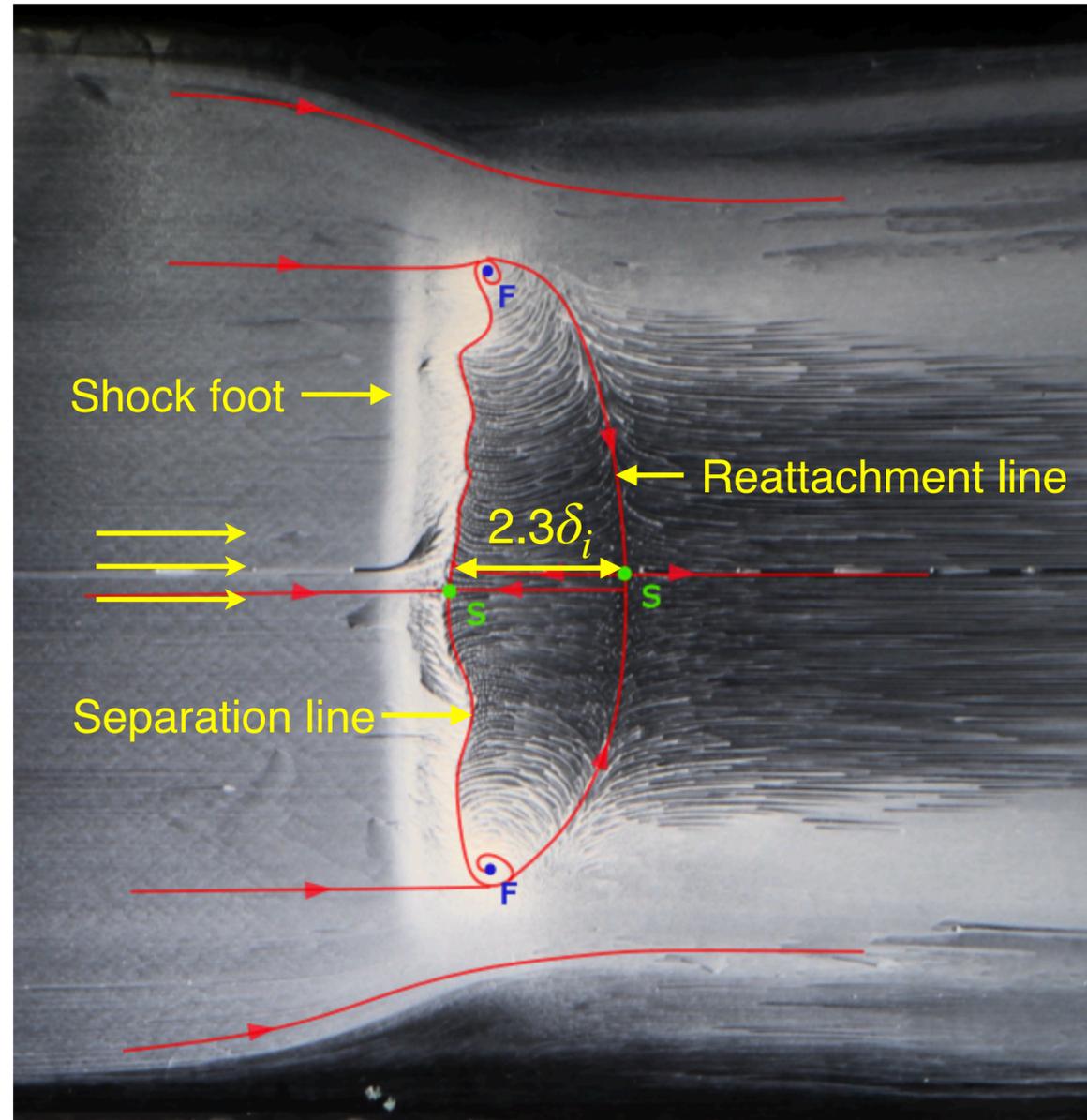
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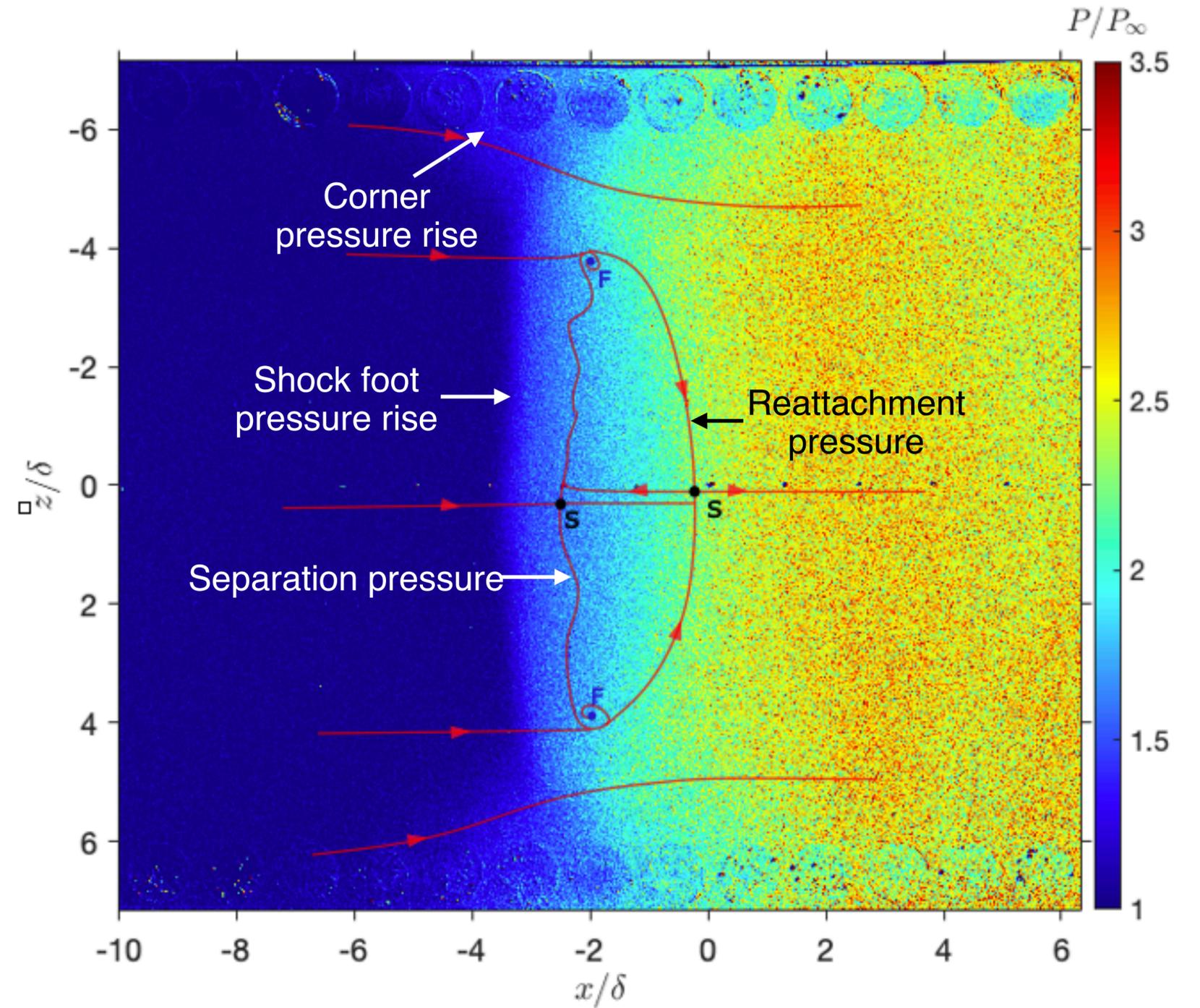
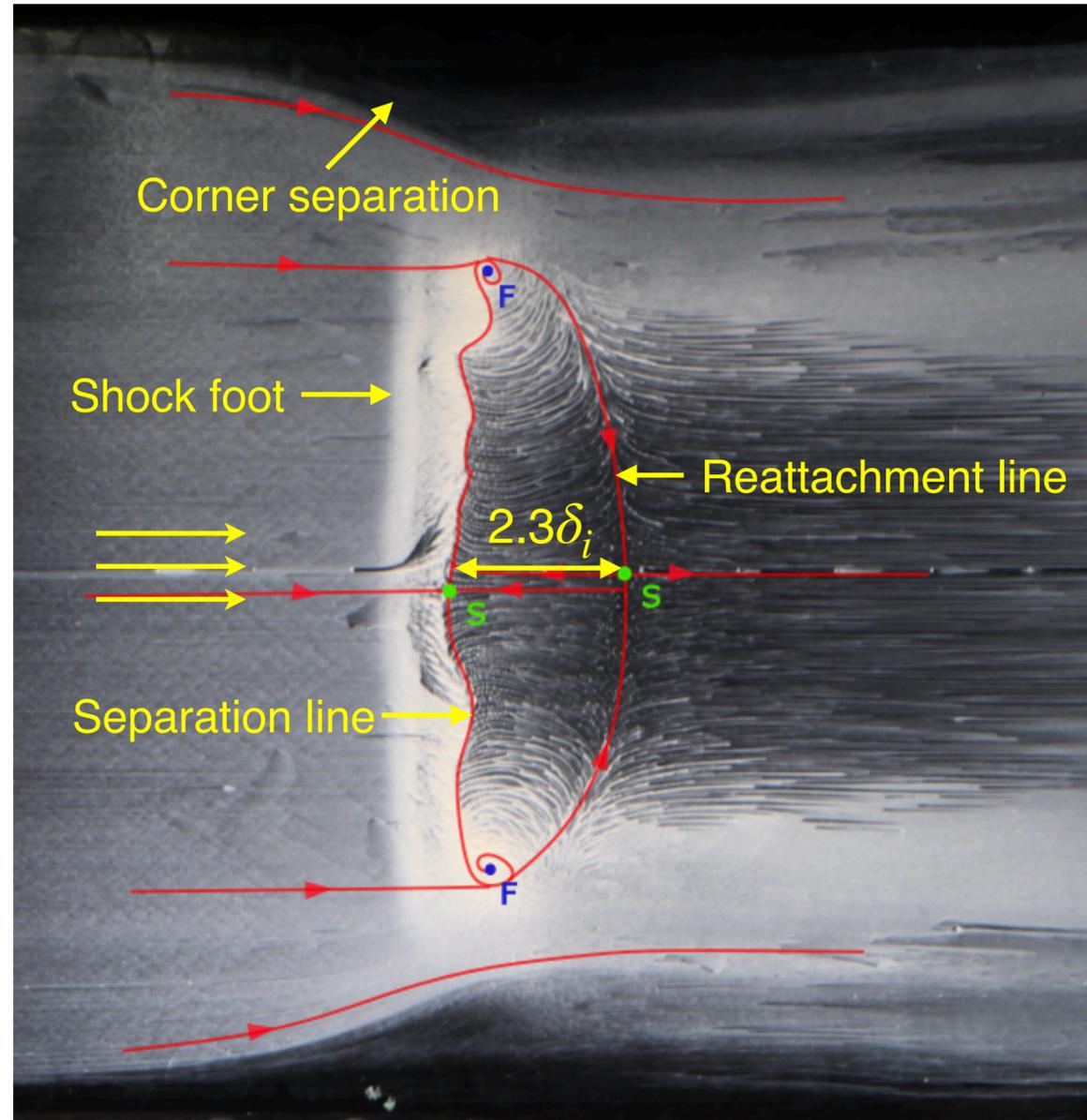
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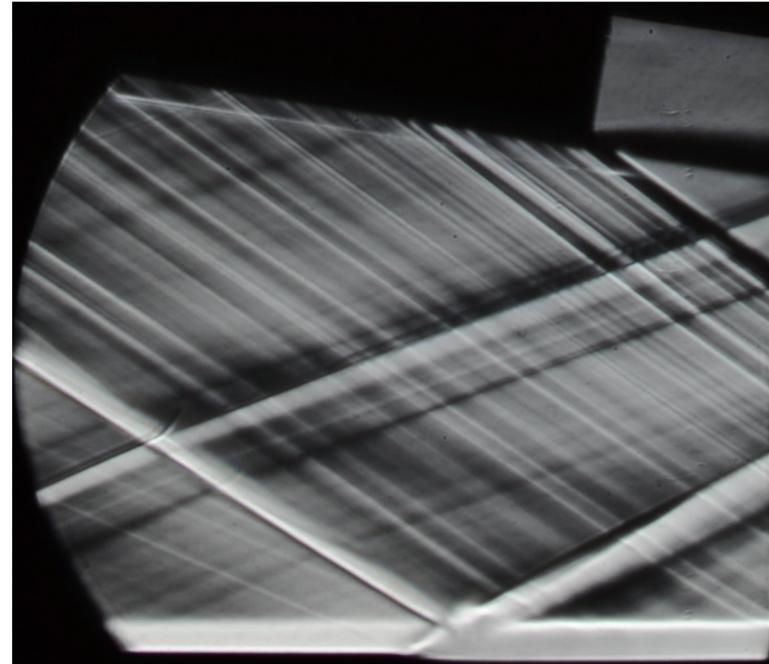


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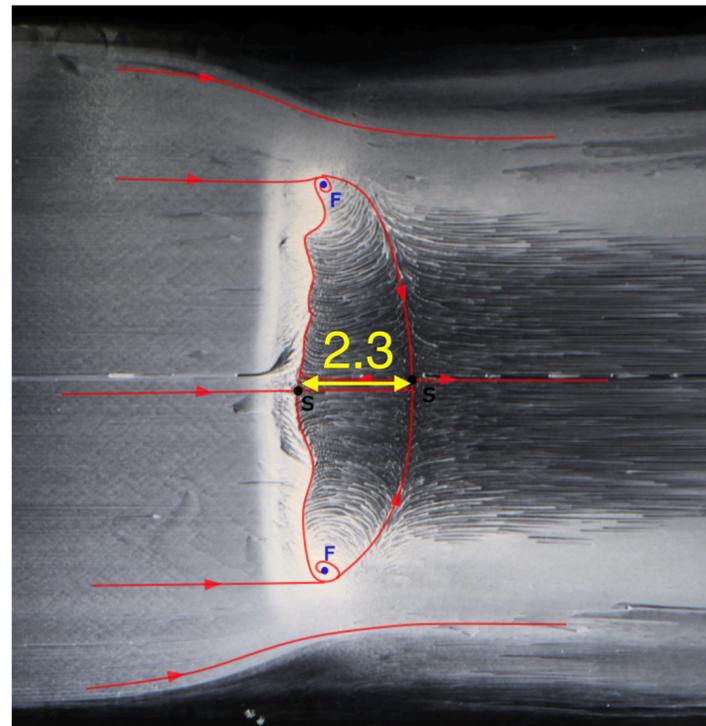


Effect of distance

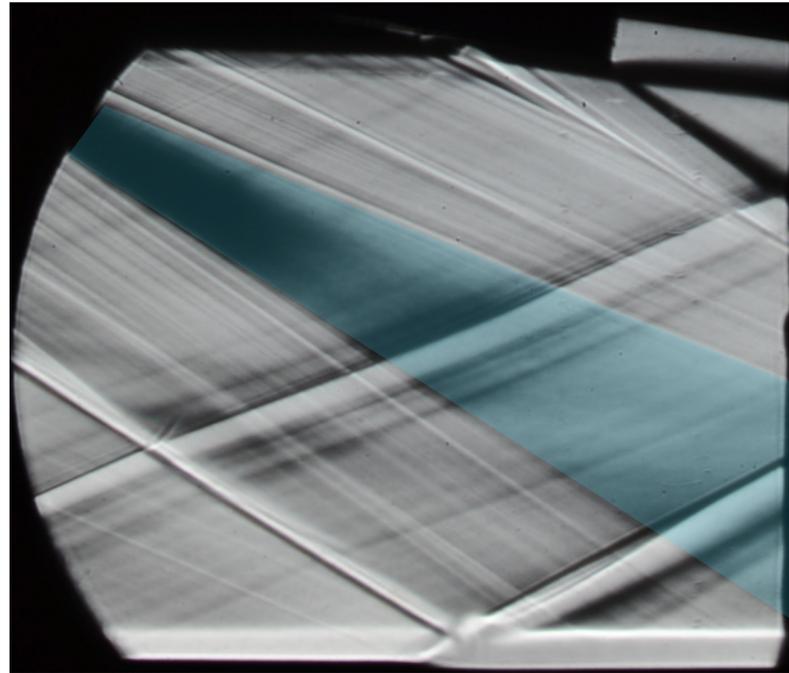
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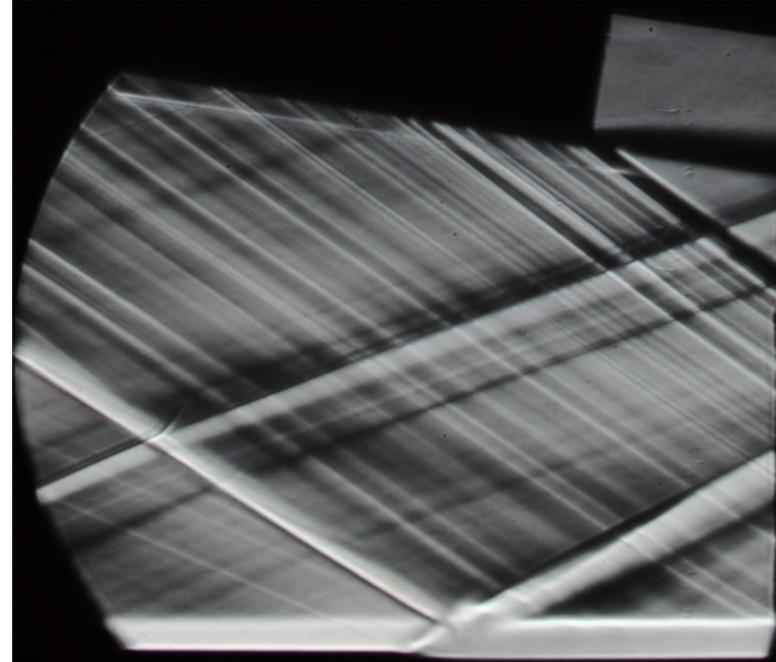
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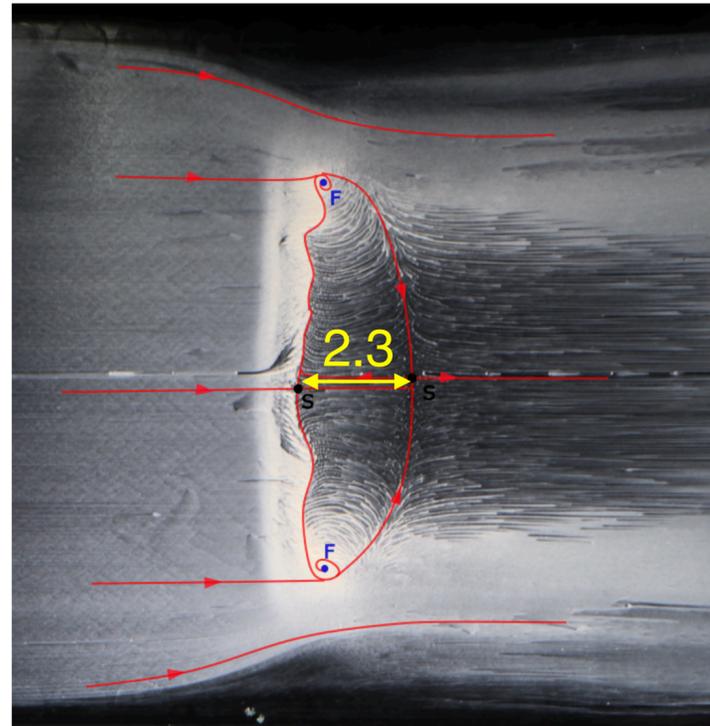
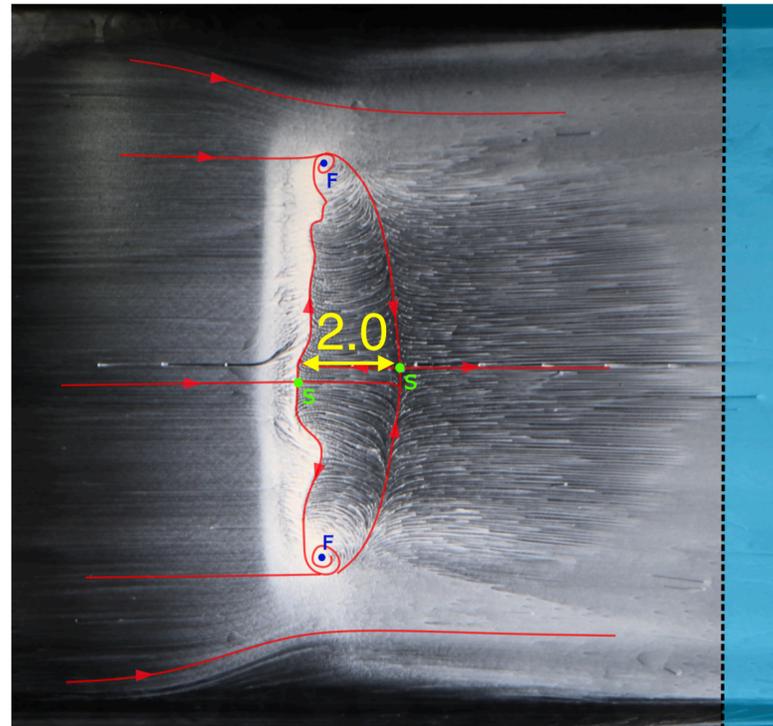
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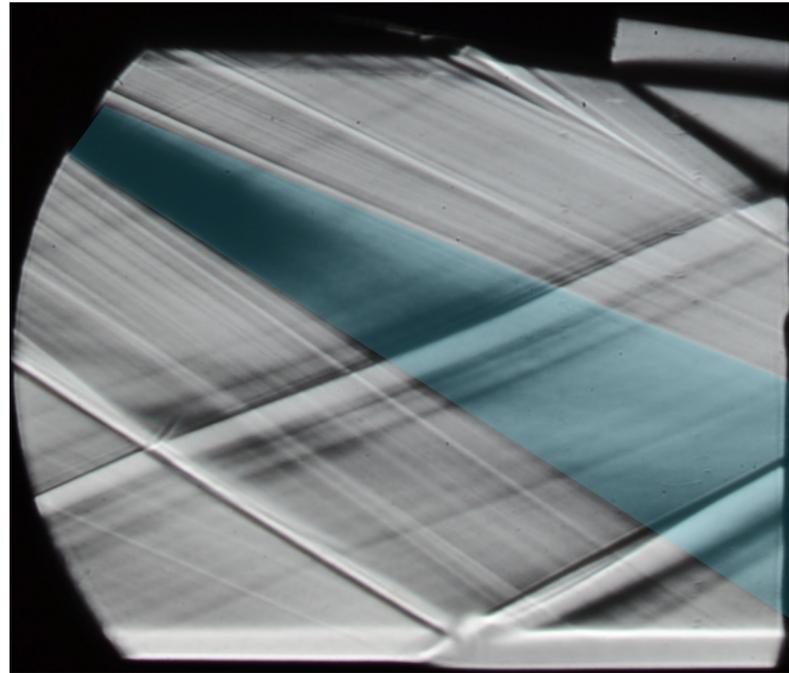
$\bar{d}_f = 6.3$



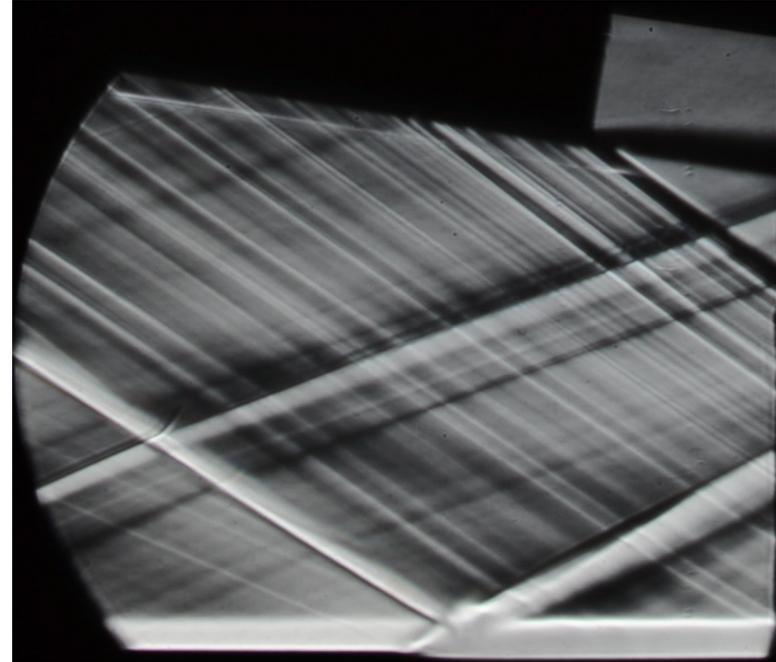
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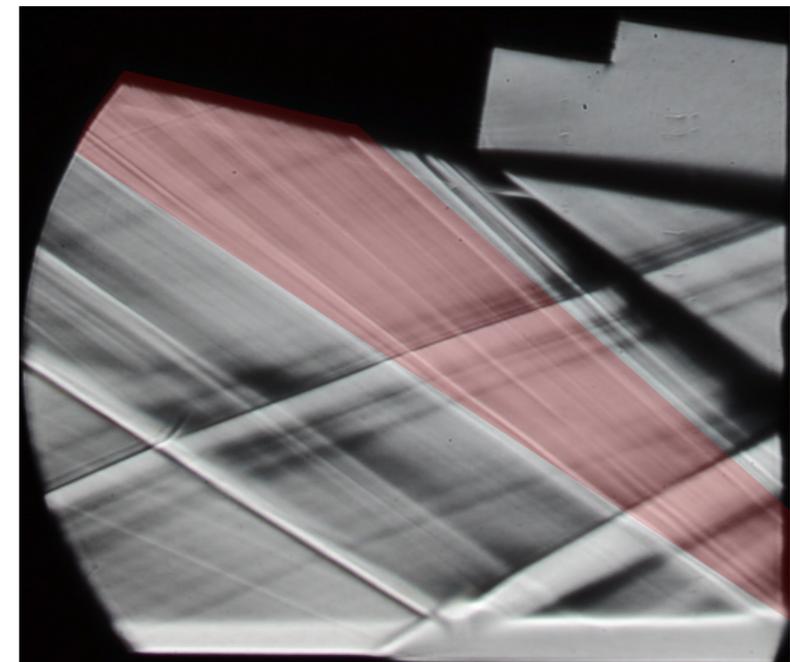
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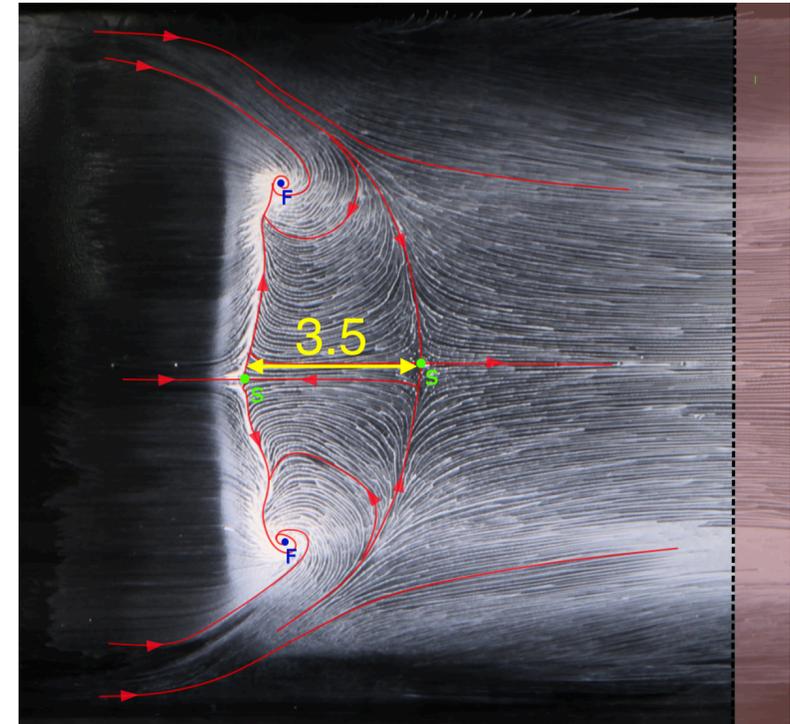
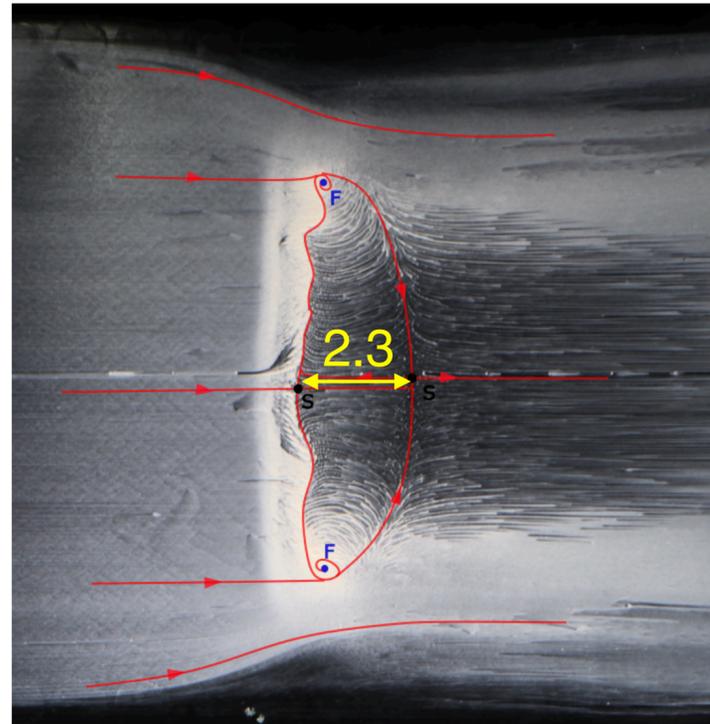
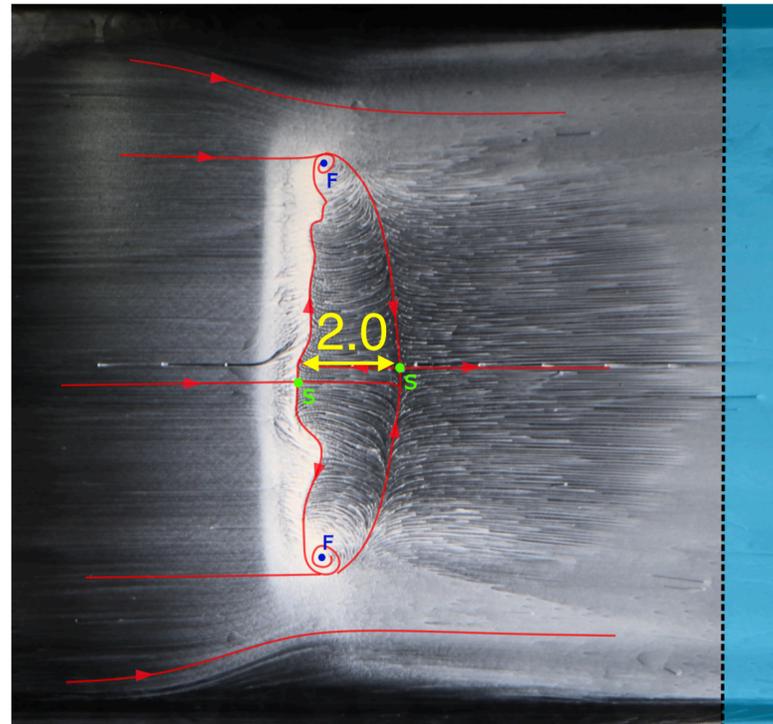
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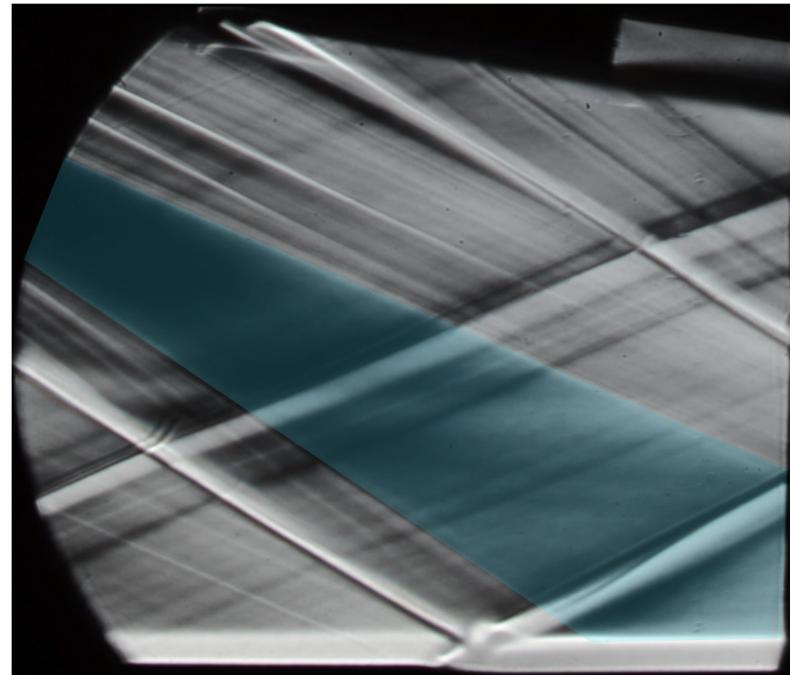
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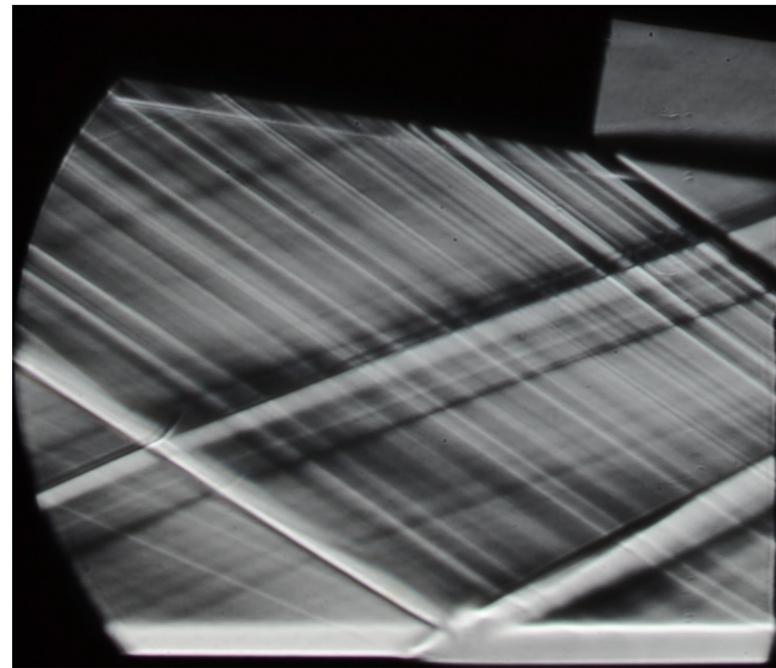
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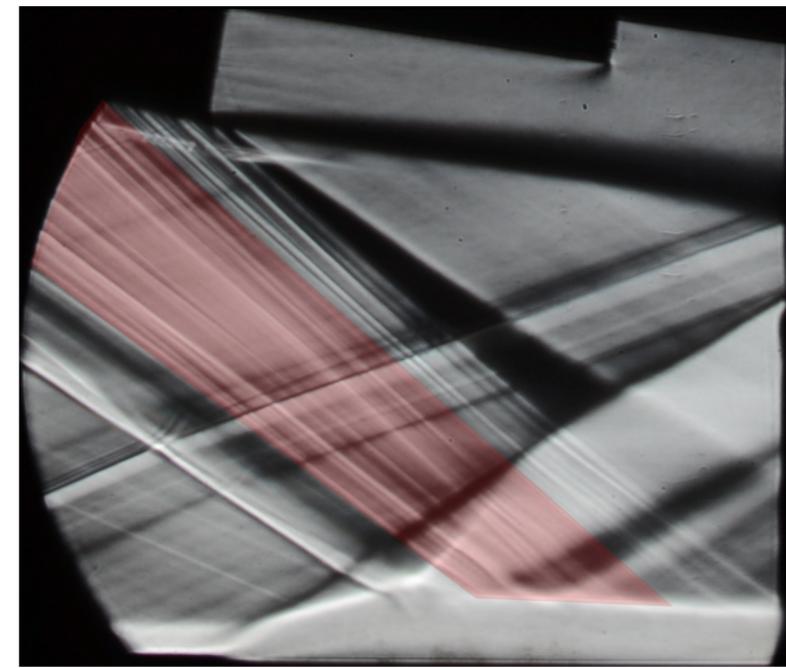
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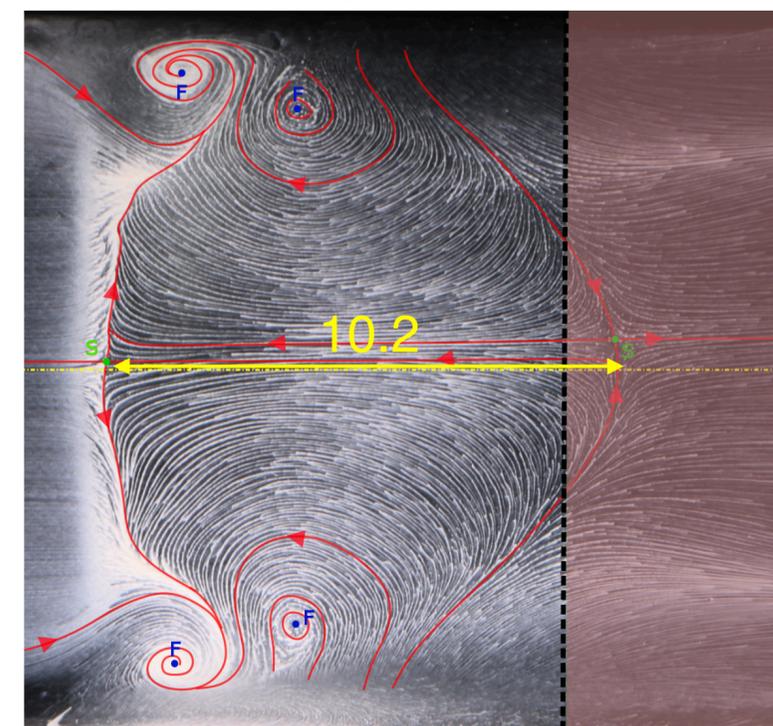
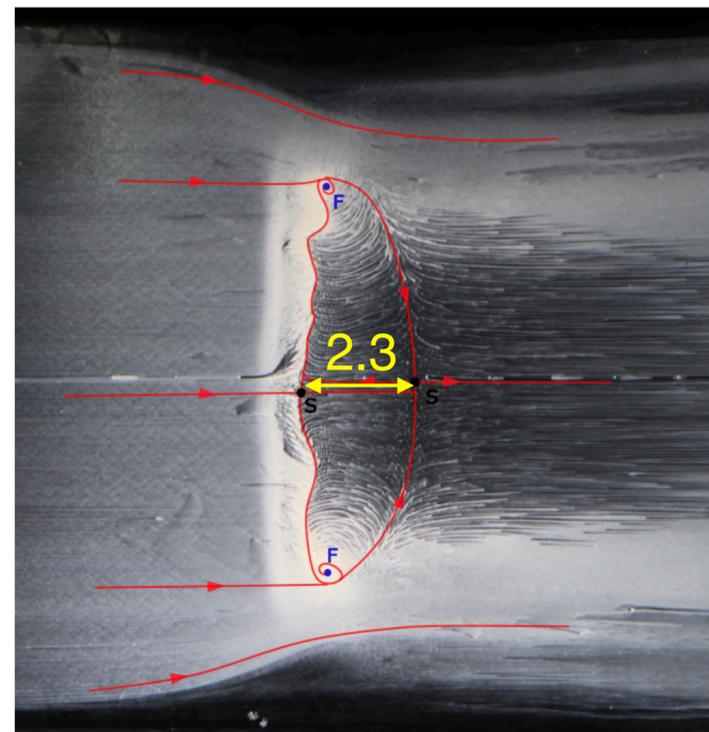
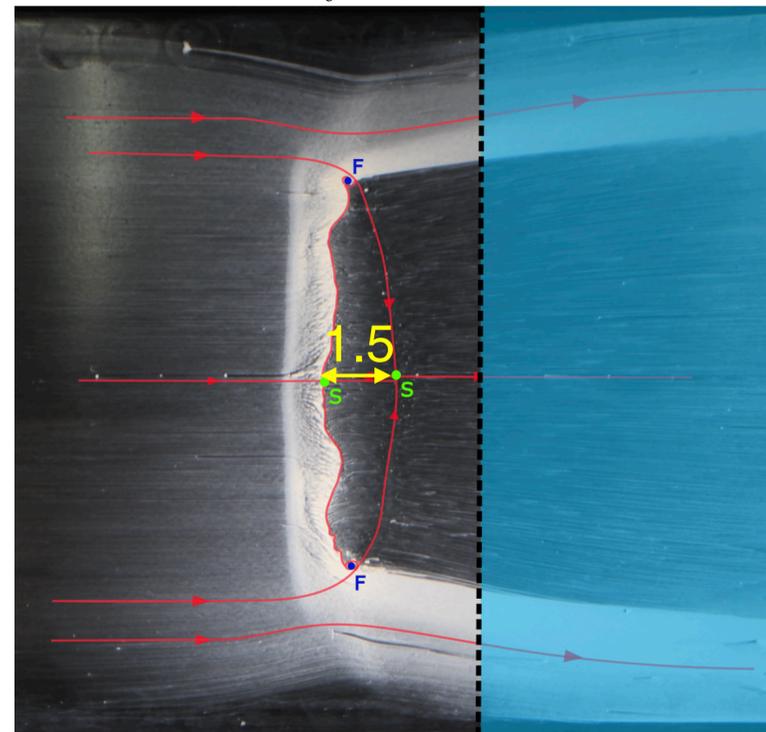
$\bar{d}_f = 1.2$



Baseline

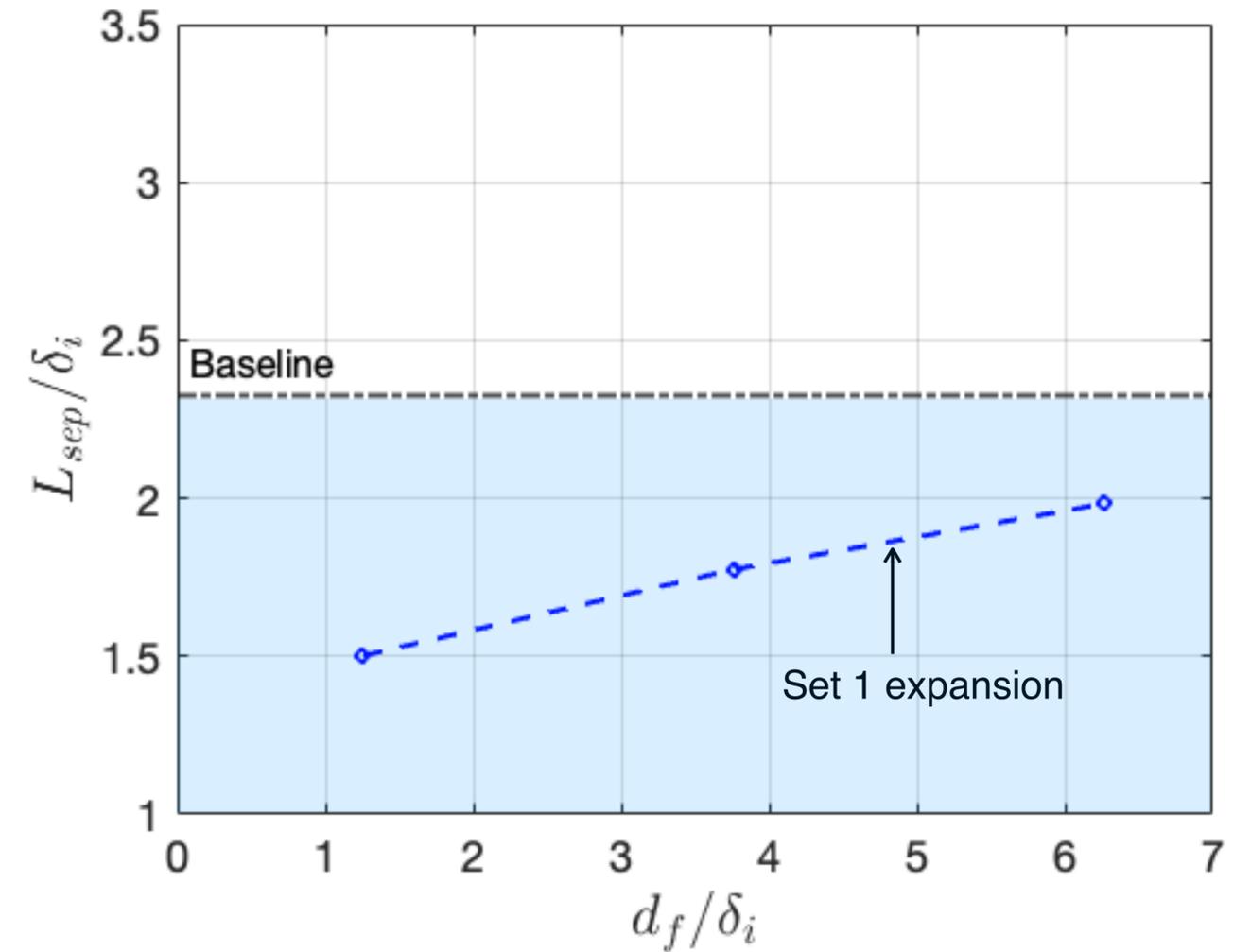
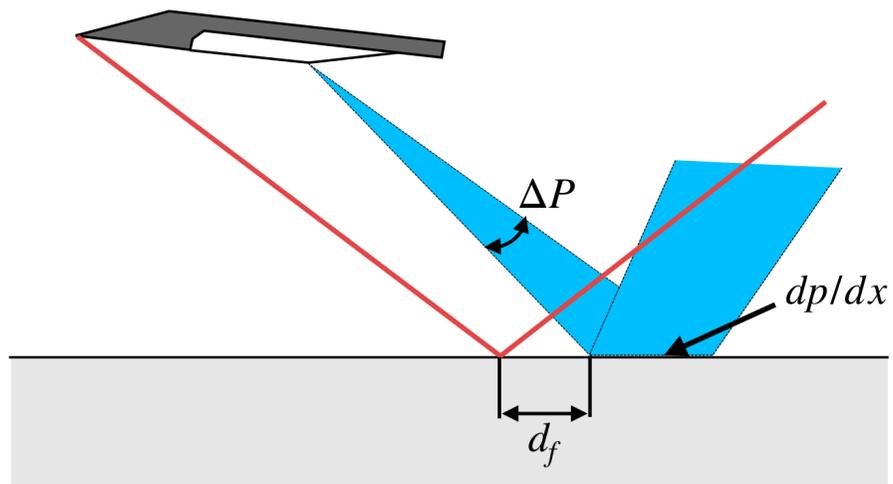


$\bar{d}_f = 1.2$



Key observations on length scales - Expansions

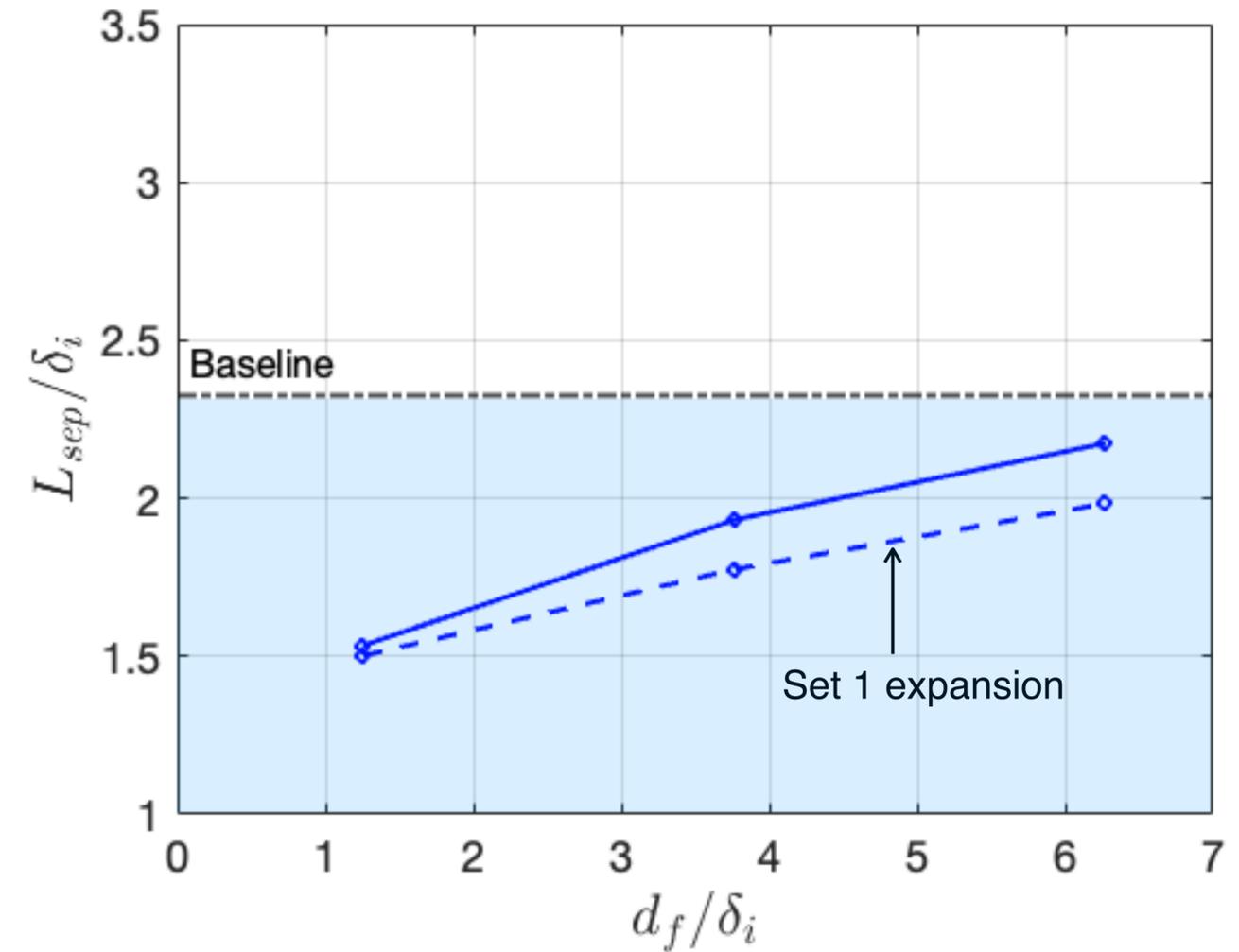
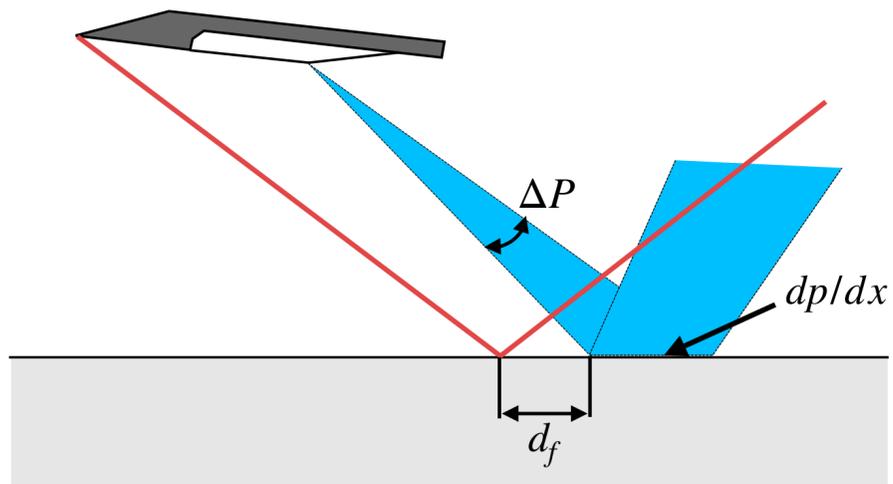
L_{sep} decreases as d_f decreases



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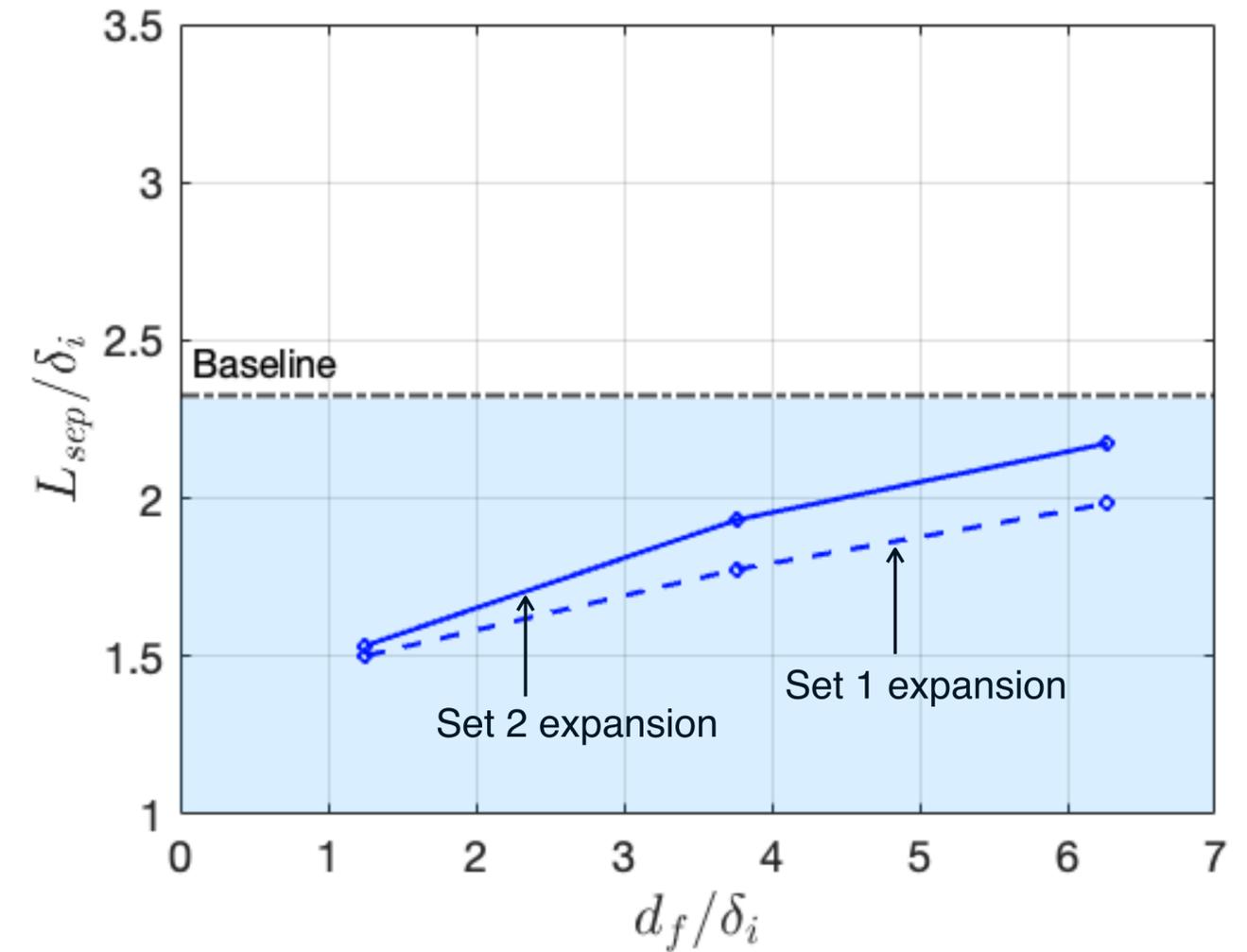
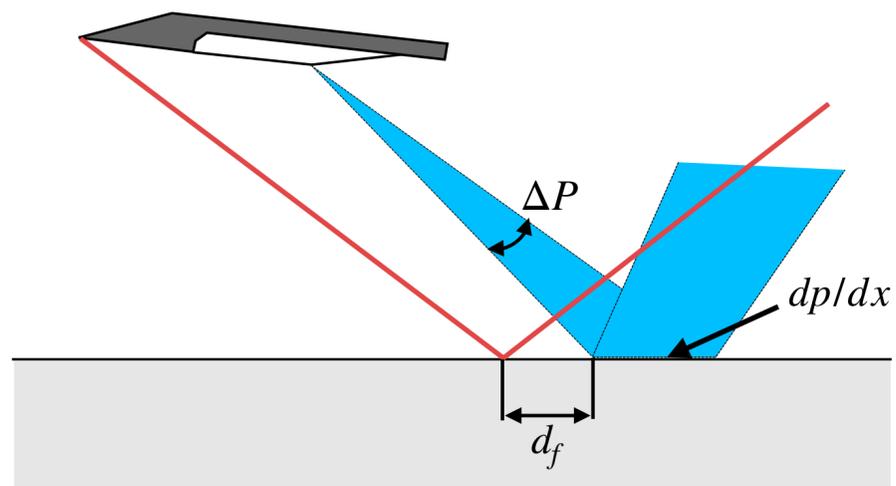
Set 1 and Set 2 similar despite different ΔP and dp/dx



Key observations on length scales - Expansions

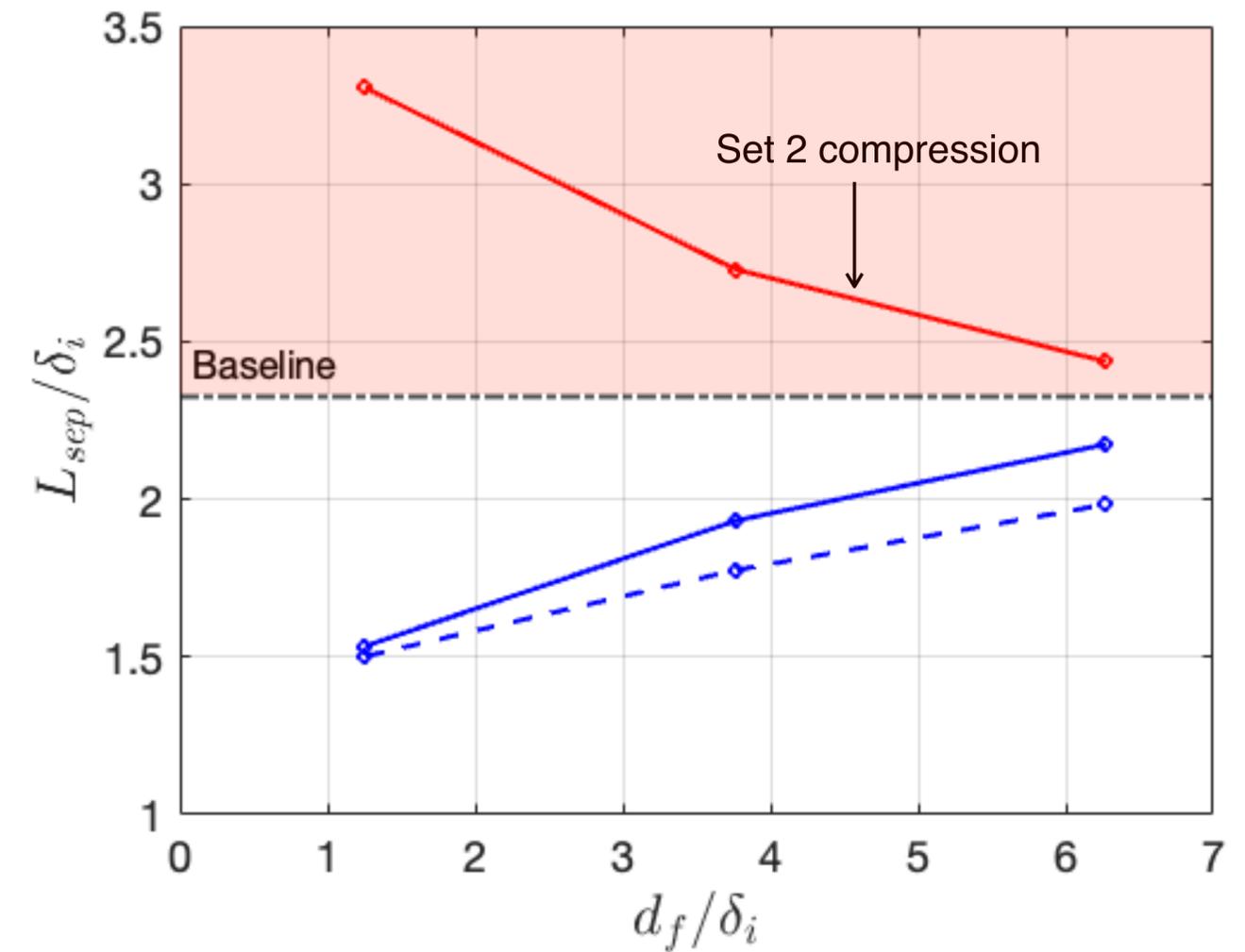
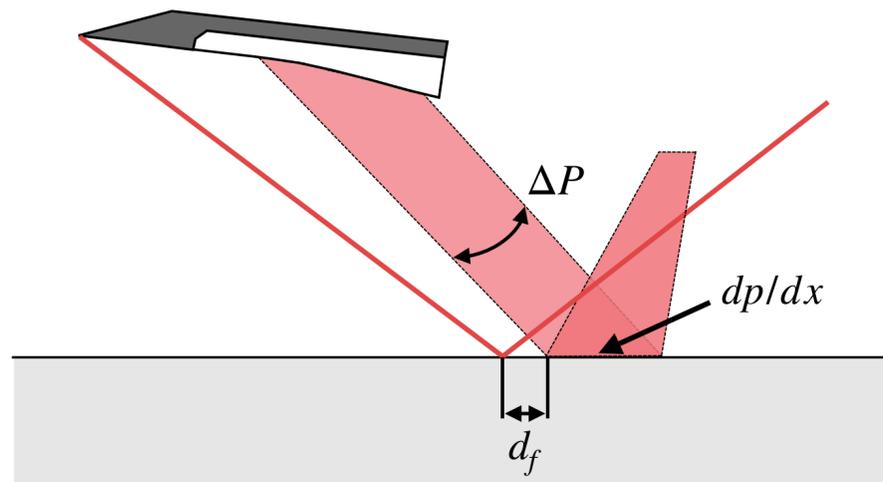
L_{sep} decreases as d_f decreases

Set 1 and Set 2 similar despite different ΔP and dp/dx



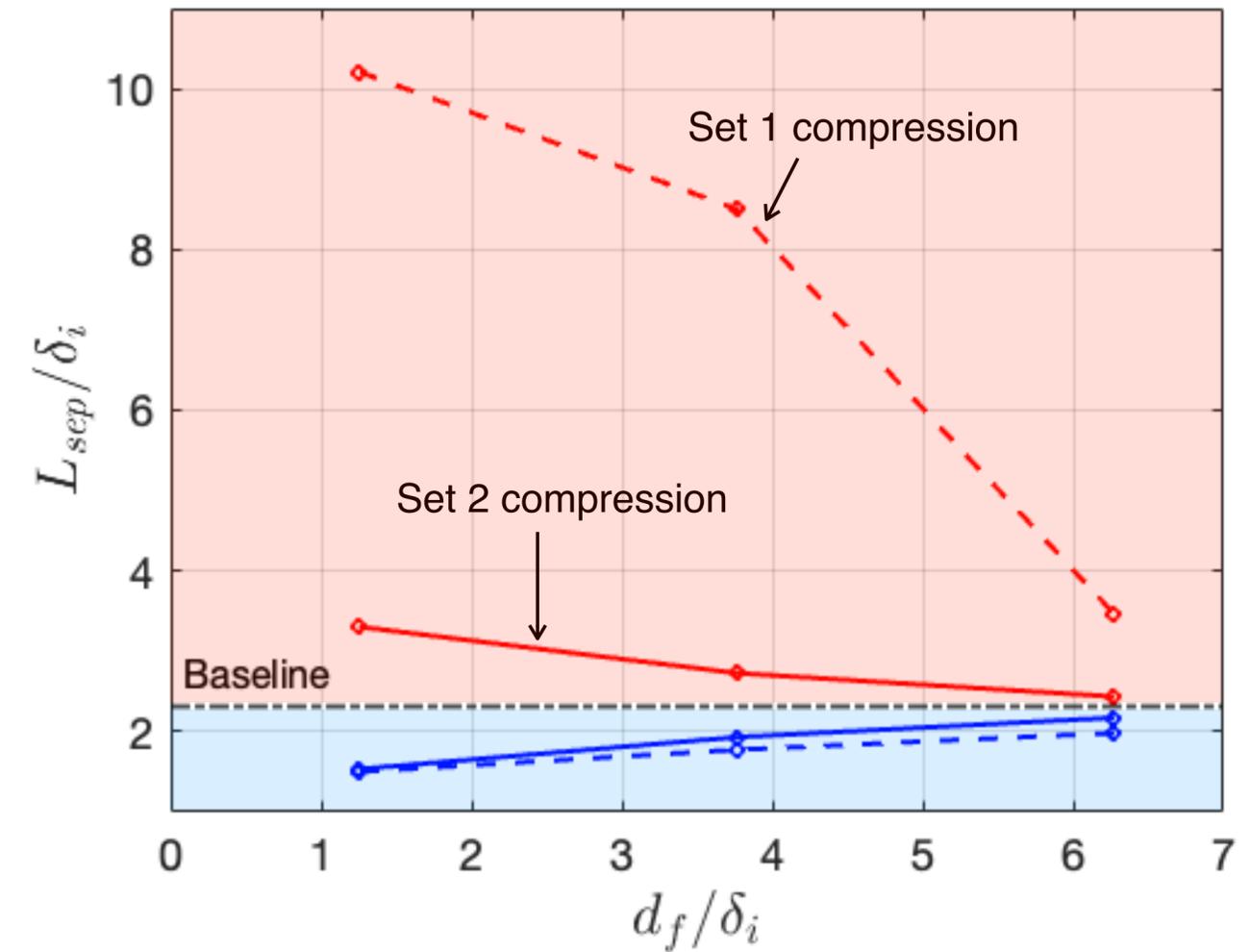
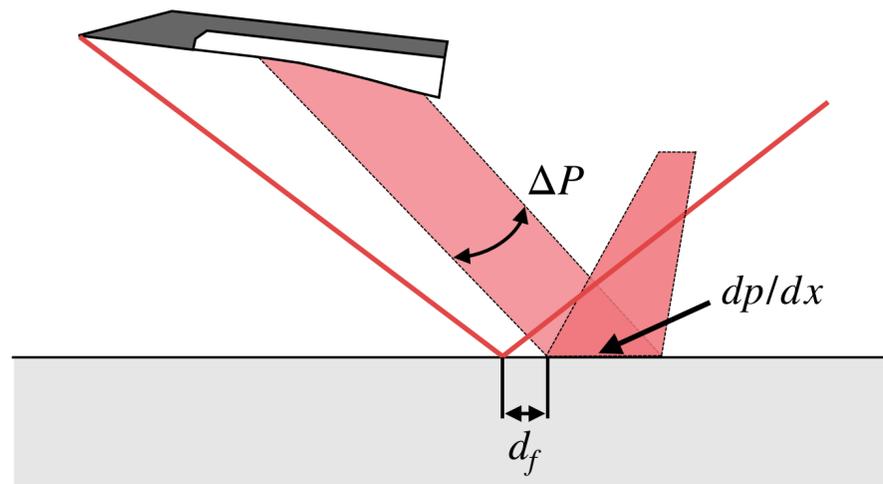
Key observations on length scales - Compressions

L_{sep} increases as d_f decreases



Key observations on length scales - Compressions

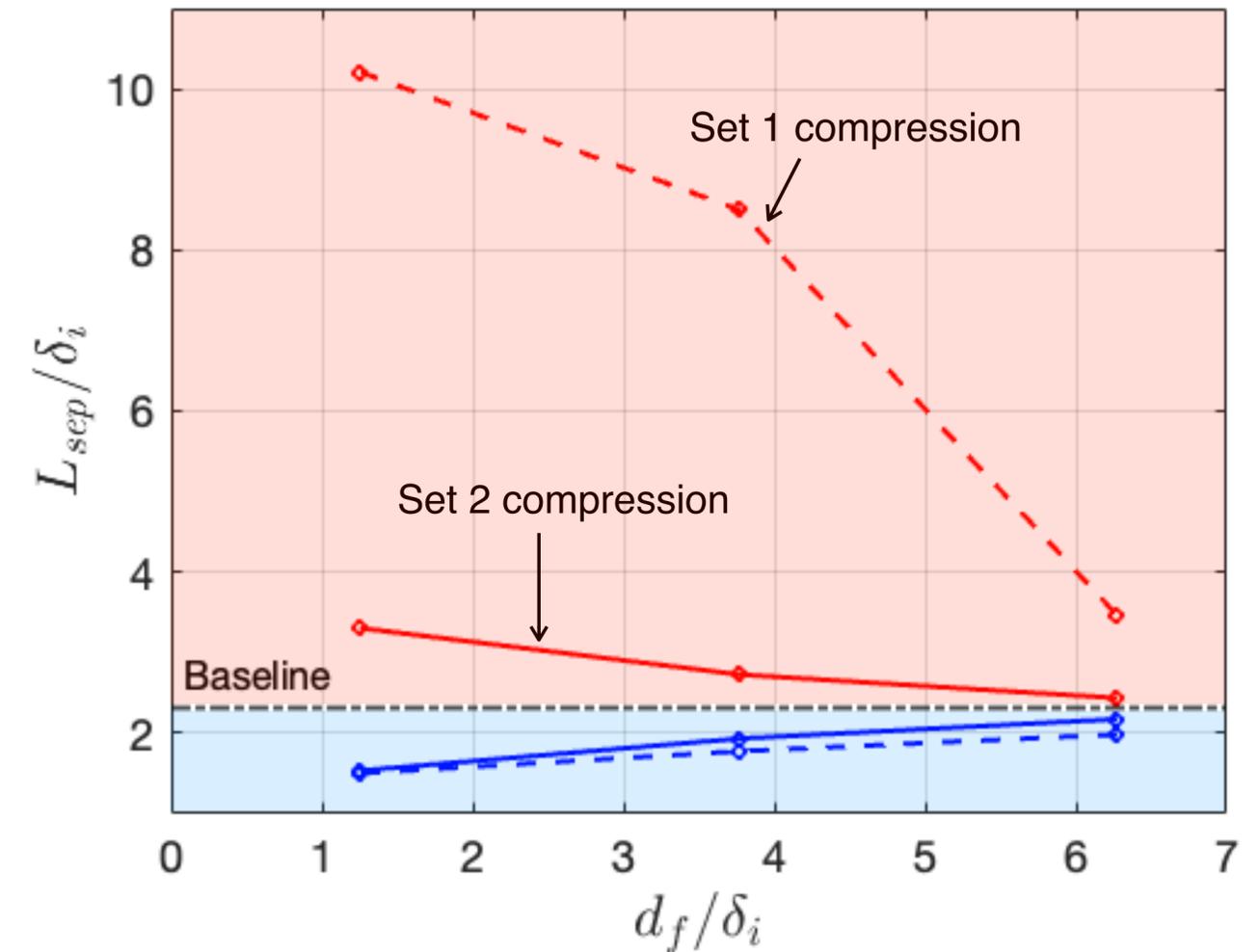
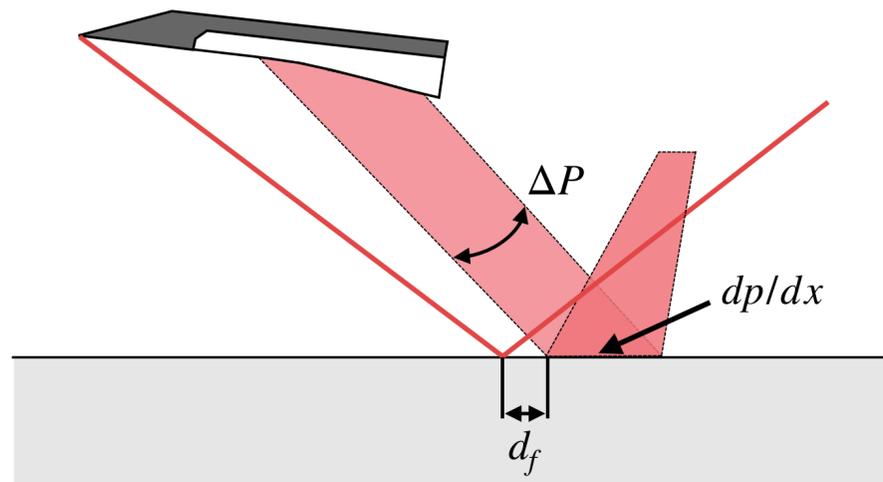
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Key observations on length scales - Compressions

L_{sep} increases as d_f decreases

340% increase in Set 1 vs 40 % in Set 2

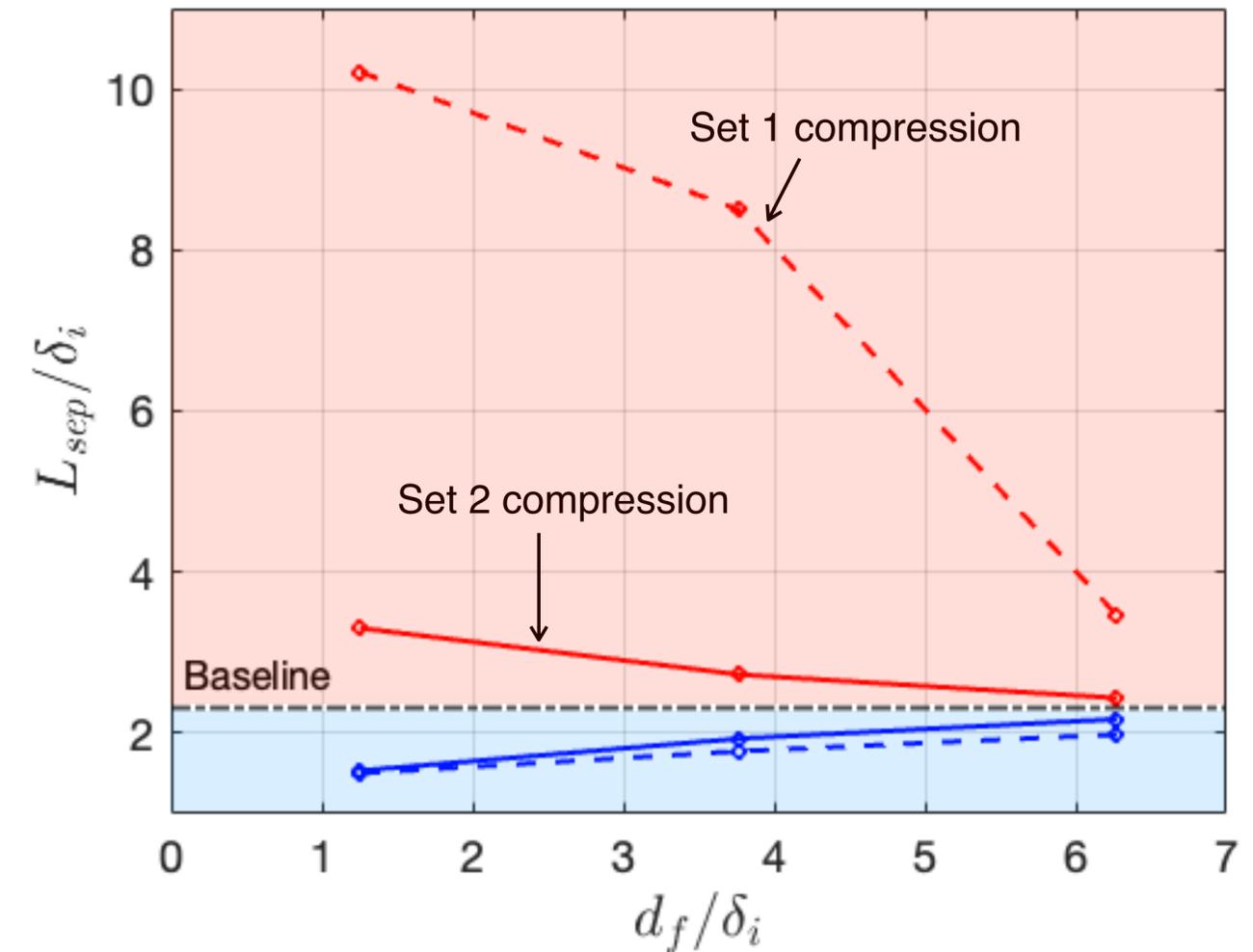
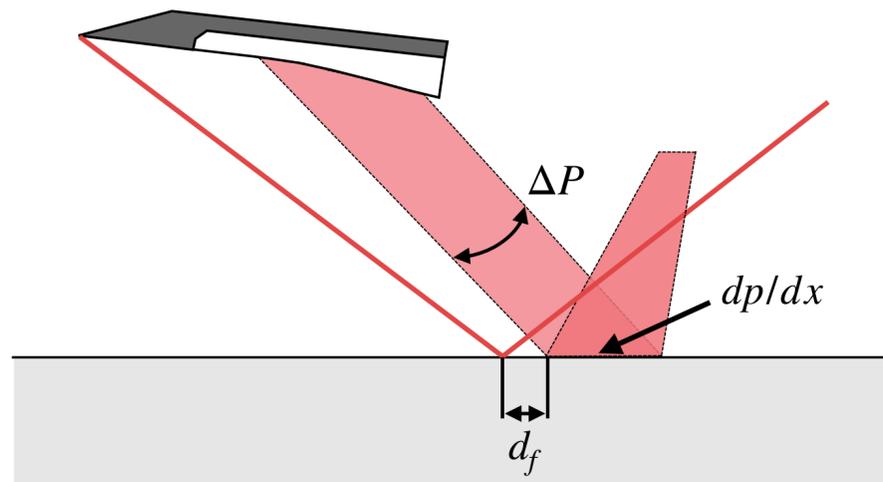


Key observations on length scales - Compressions

L_{sep} increases as d_f decreases

340% increase in Set 1 vs 40 % in Set 2

Discrepancy due to large difference in gradients



Set 2 $\overline{\Delta P} = + 0.64$

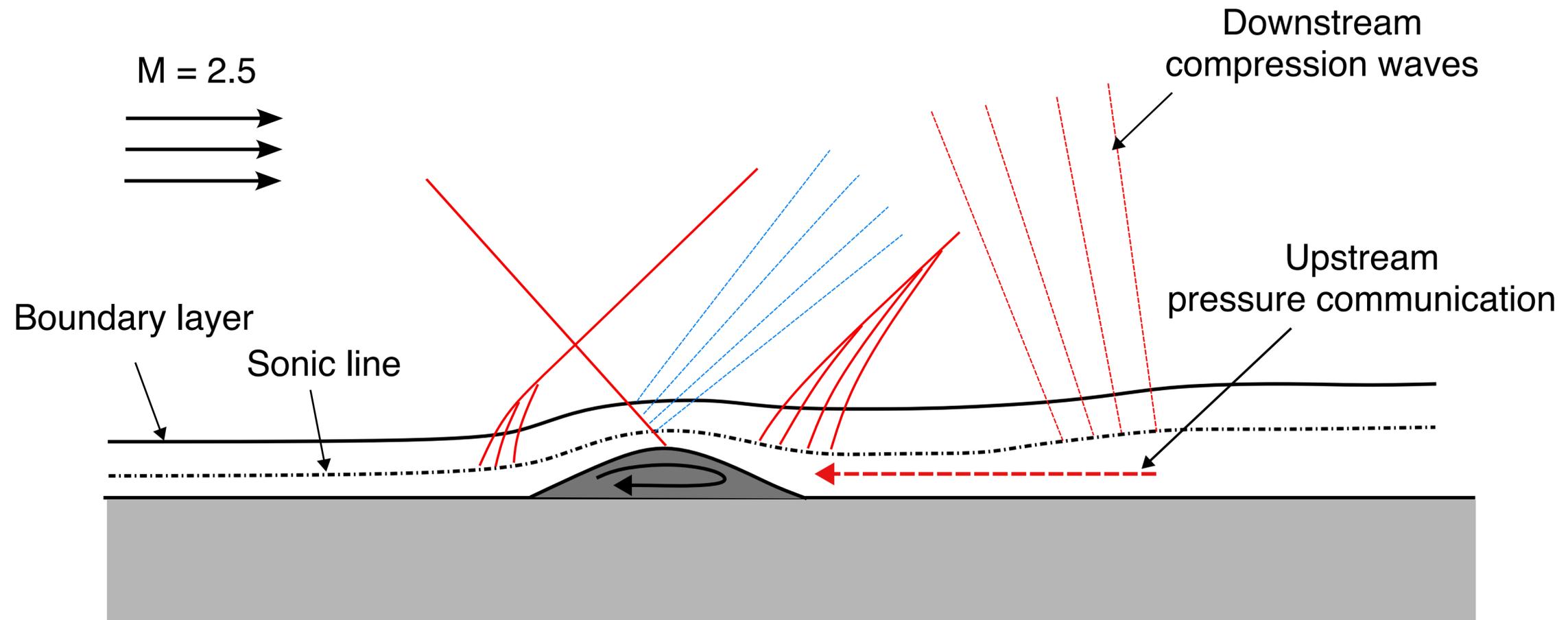
Set 1 $\overline{\Delta P} = + 0.91$

$$\overline{dp/dx} = + 0.07$$

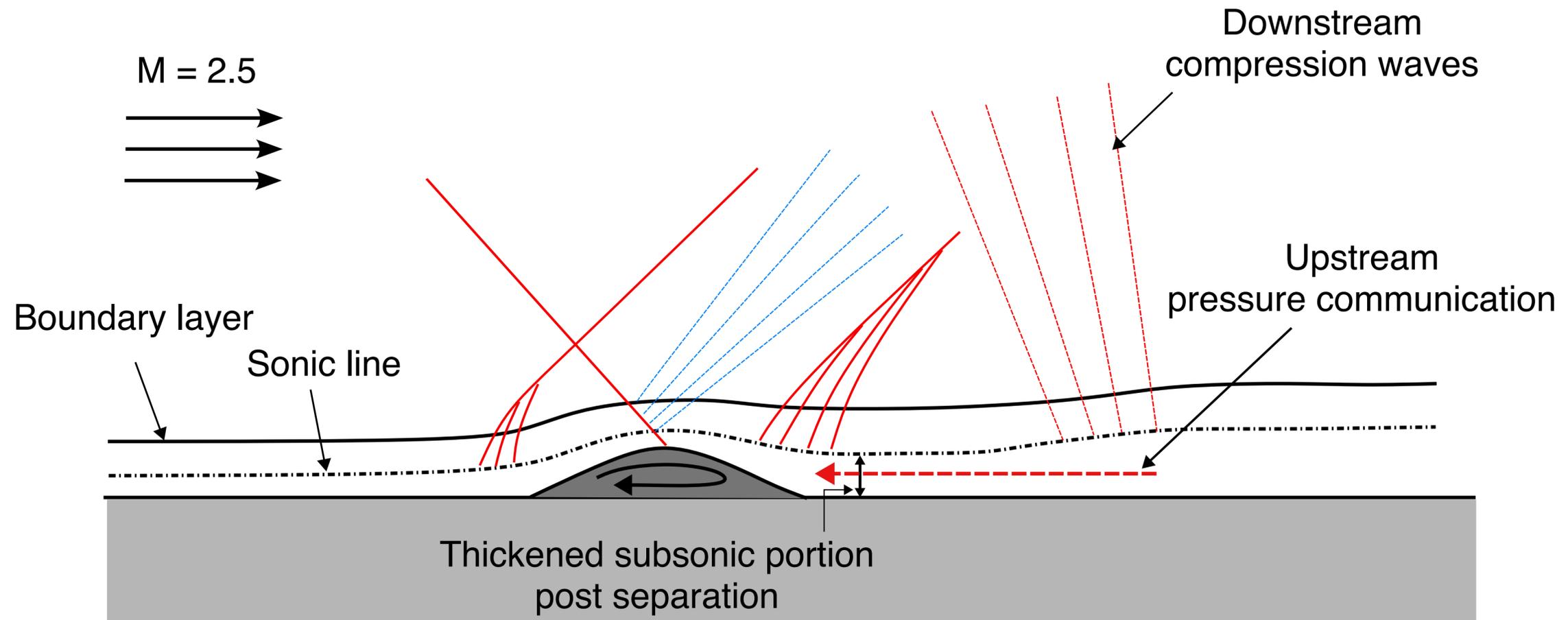
$$\overline{dp/dx} = + 0.32$$

What is the mechanism?

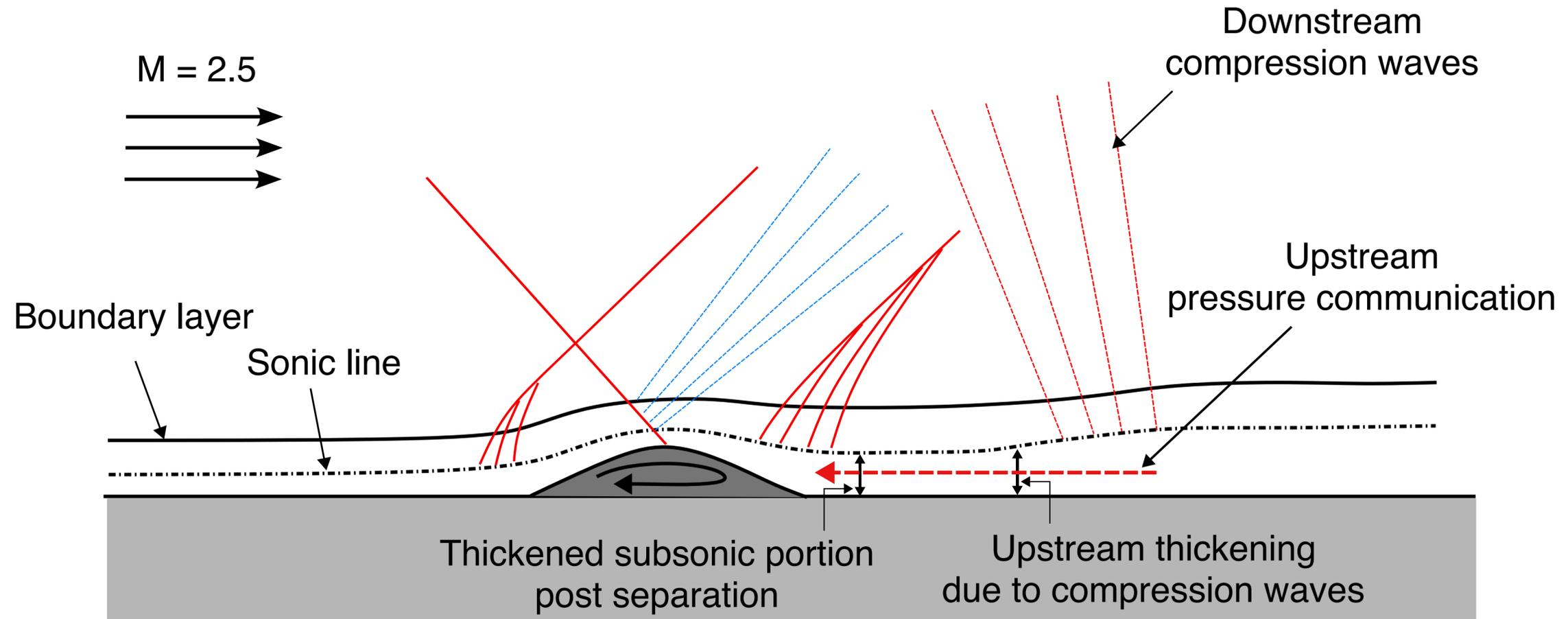
Mechanism 1



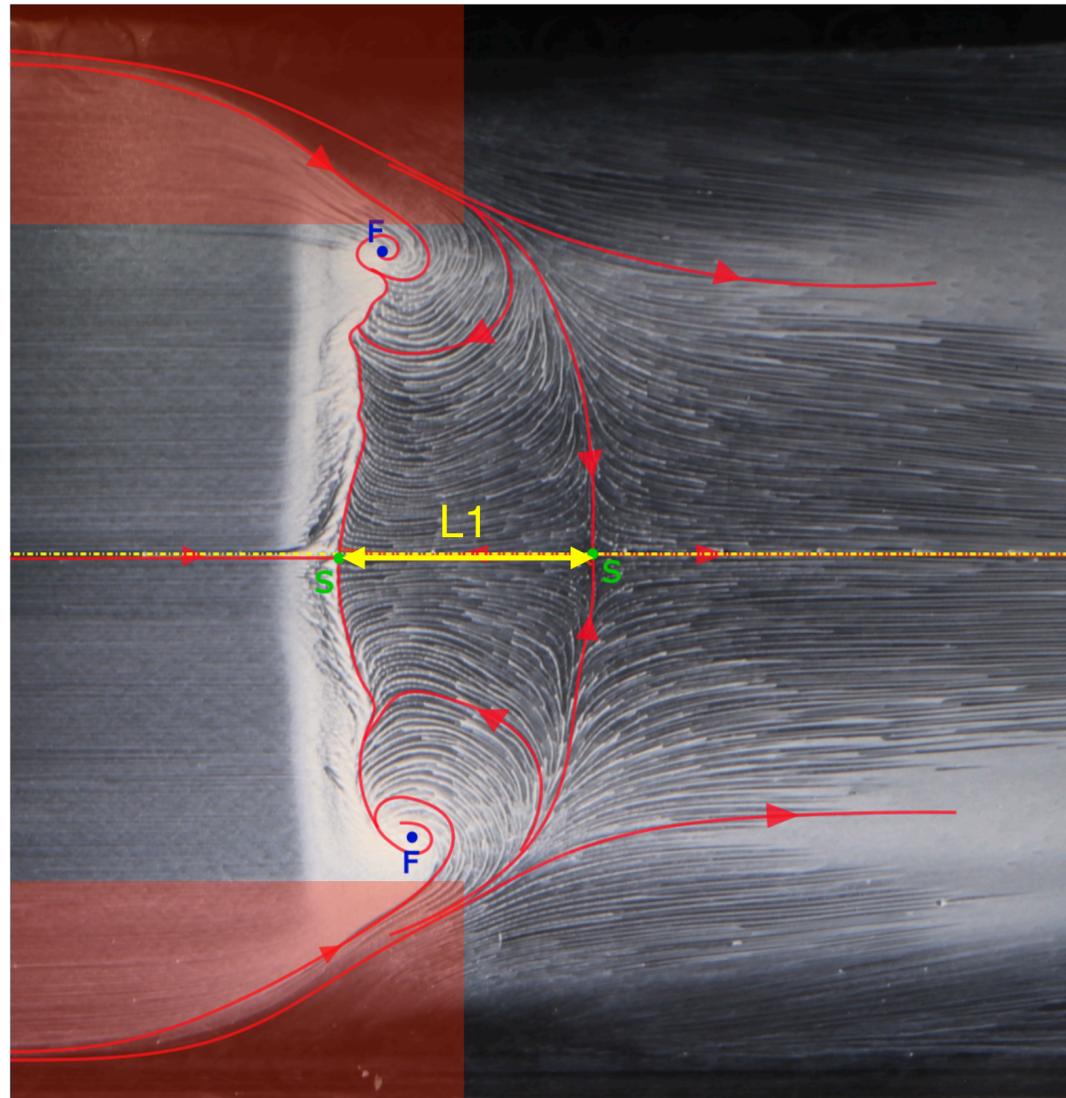
Mechanism 1



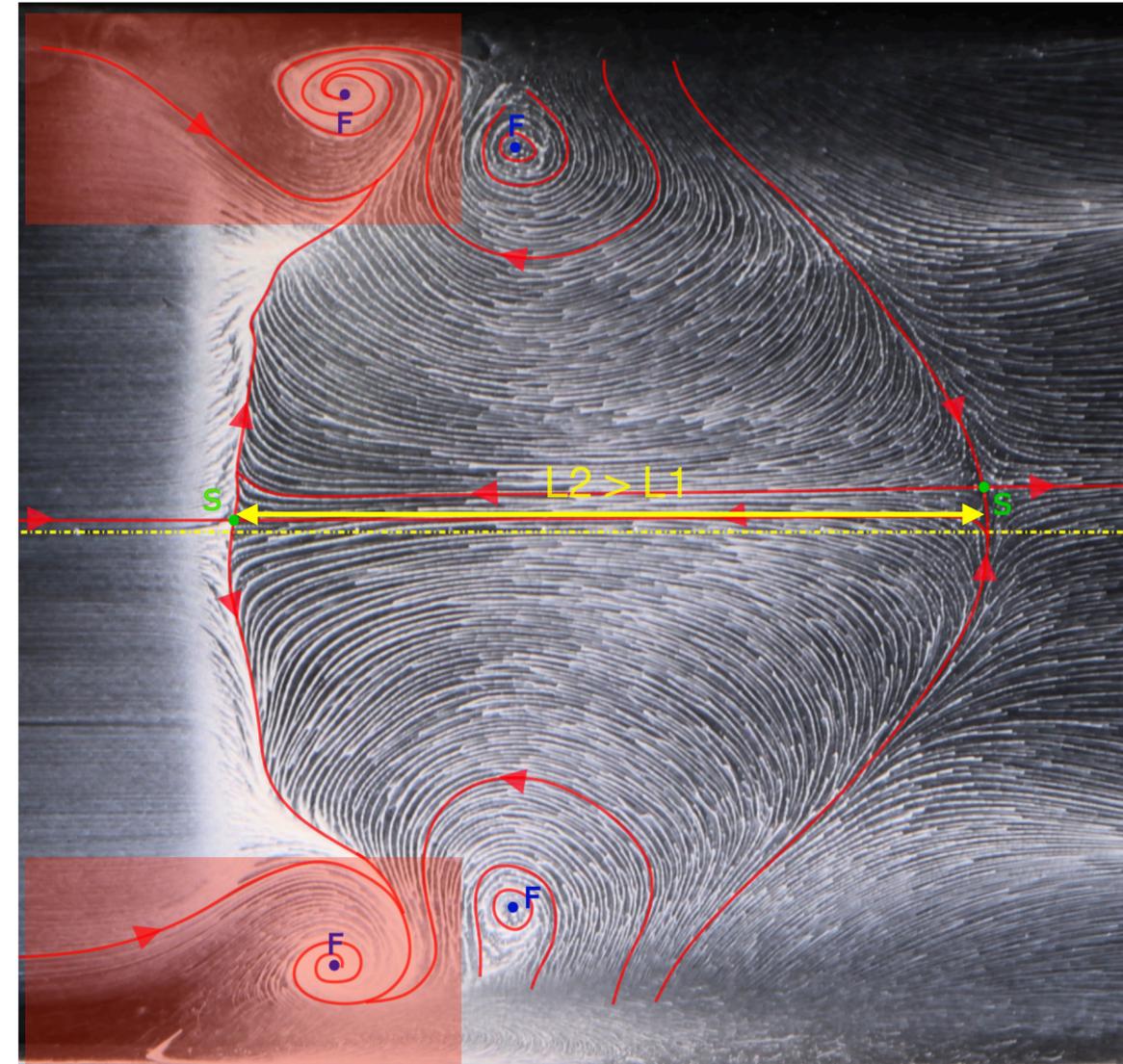
Mechanism 1



Corner flow

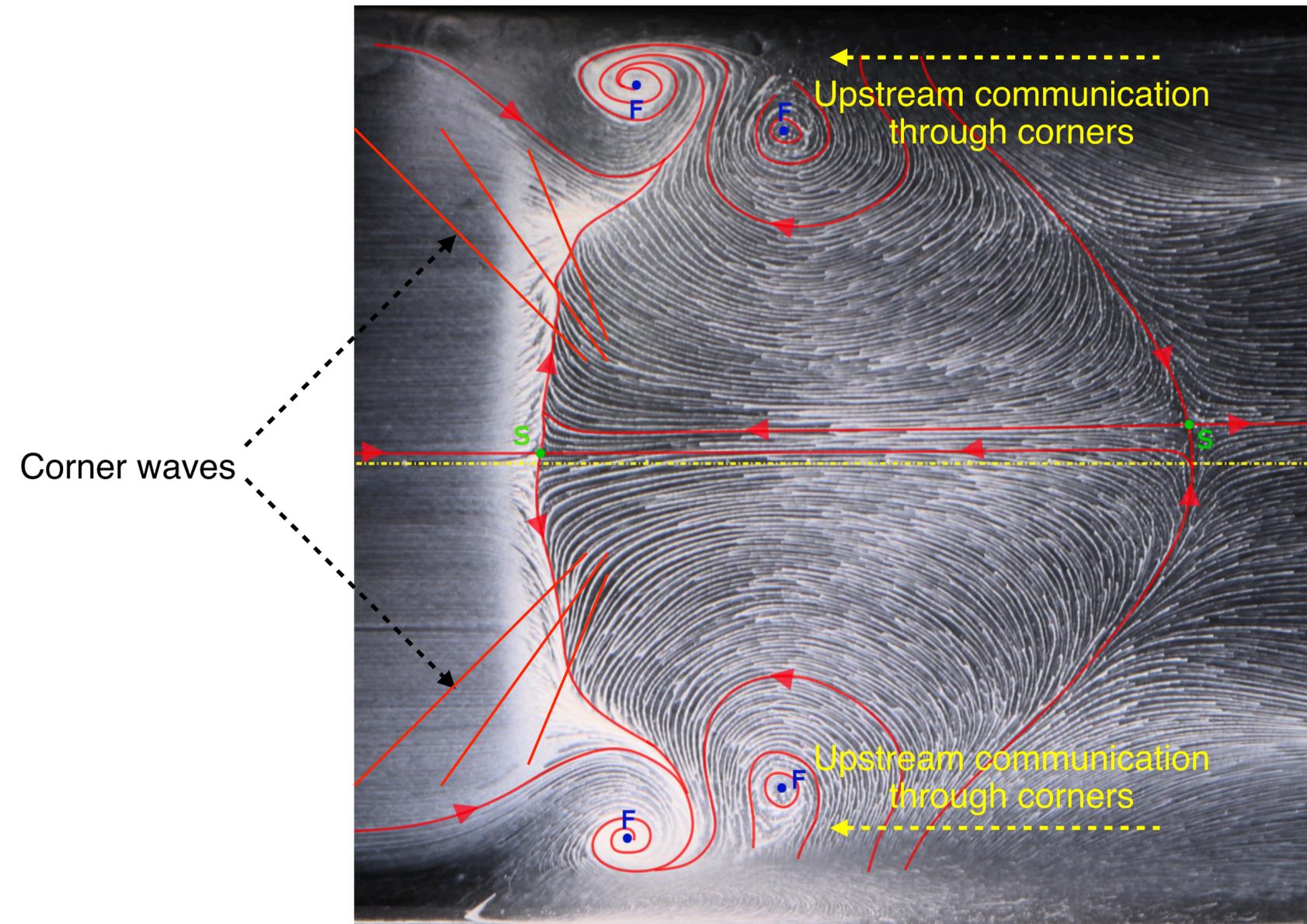


Set 2 compression



Set 1 compression

Mechanism 2



Conclusions

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Downstream waves have a considerable effect on the main SBLI - even up to $6.3\delta_i$

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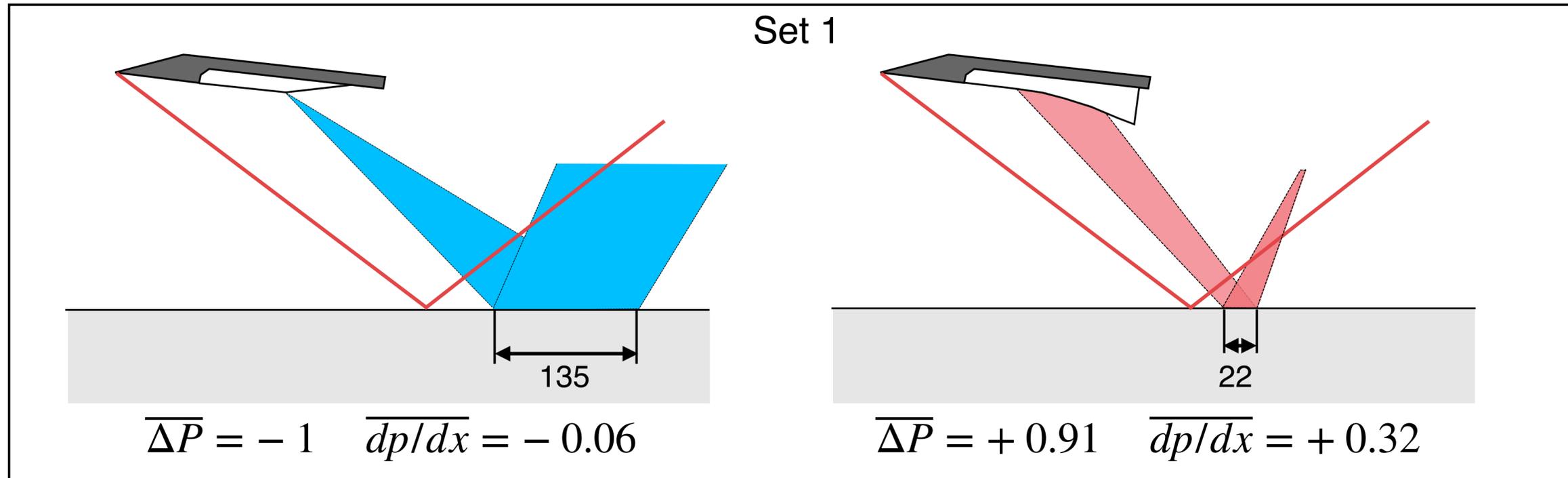
Two possible mechanisms were proposed

References

1. Grossman, I. J., and Bruce, P. J., “Confinement effects on regular-irregular transition in shock-wave-boundary-layer interactions,” *Journal of Fluid Mechanics*, Vol. 853, 2018, pp. 171-204. <https://doi.org/10.1017/jfm.2018.537>.
2. Missing, T., and Babinsky, H., Corner effects on oblique shock wave boundary layer interactions in rectangular channels, *AIAA SciTech 2023 Forum*, 2023-0650. <https://doi.org/10.2514/6.2023-0650>.

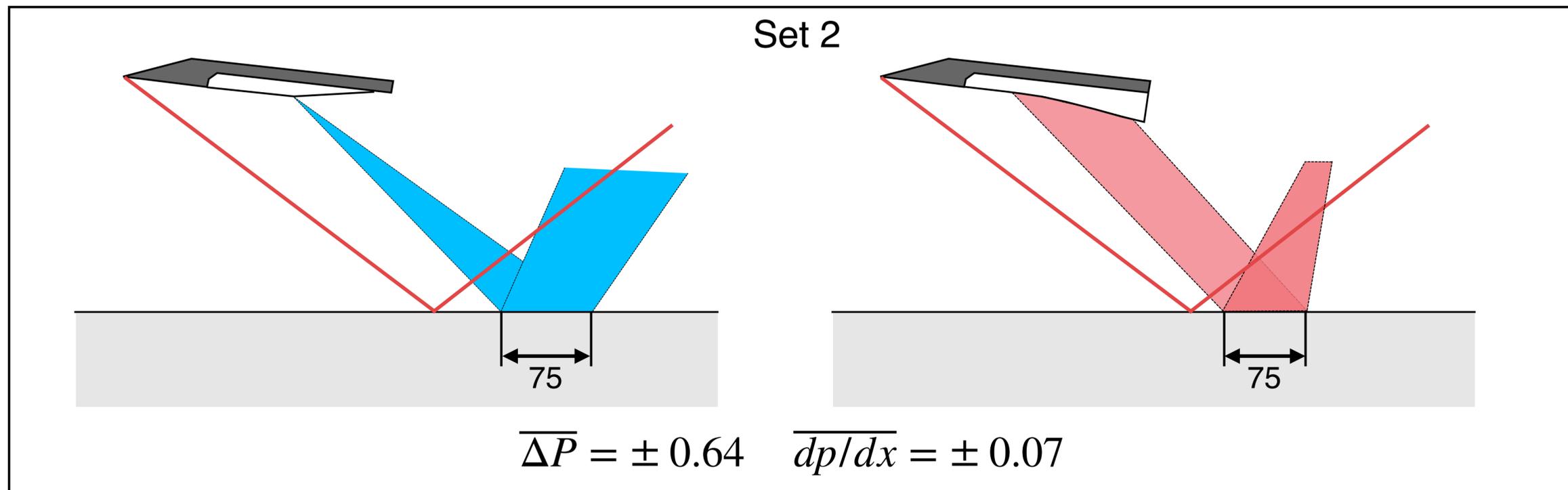
Additional slides: Setup details and data

Details of test setups



Three $\overline{d_f}$'s

- 1.2
- 3.8
- 6.3



Three $\overline{d_f}$'s

- 1.2
- 3.8
- 6.3

Centreline separation lengths

Set 1

Case	Expansion			Baseline	Compression		
d_f/δ_i	1.25	3.76	6.27	-	6.27	3.76	1.25
L_{sep}/δ_i	1.49	1.78	1.98	2.33	3.46	8.51	10.21

Set 2

Case	Expansion			Baseline	Compression		
d_f/δ_i	1.25	3.76	6.27	-	6.27	3.76	1.25
L_{sep}/δ_i	1.53	1.93	2.17	2.33	2.43	2.74	3.30